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[54] UNIVERSAL DYNAMIC NAVIGATION, SURVEILLANCE, EMERGENCY LOCATION, AND COLLISION AVOIDANCE SYSTEM AND METHOD

5,043,903 8/1991 Constant 342/31

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[57] ABSTRACT

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A craft tracking and collision avoidance system is disclosed. The system allows the positions of a plurality of craft, either on land, sea, or air, or space, to be monitored. Each craft determines its own position using an existing position determining system such as LORAN or GPS. Each craft then transmits a radio frequency signal into which position information, preferably identifying information, and other messages, have been encoded. Each craft broadcasts its position, identifying information and other messages on a regular basis without the need for any interrogation signal. The broadcast position and identification information can be received by other craft and, since each craft has determined its own position, can be used to determine the proximity and identity of other craft, and if the craft are on a collision course. Preferably, the position of all the craft within a predetermined range of a craft is represented on a display in order to give the craft operator a visual indication of traffic surrounding his craft.

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[52] U.S. Cl. 364/461; 364/452; 340/961; 342/30; 342/41; 342/455

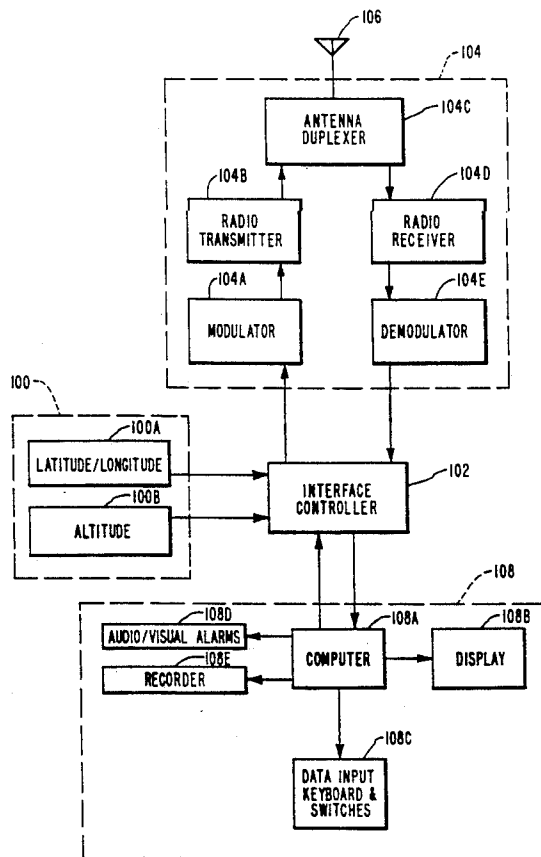
[58] Field of Search 364/452, 459, 460, 461; 340/961; 342/29-32, 41, 450, 451, 455, 463

[56] References Cited

U.S. PATENT DOCUMENTS

3,750,166	7/1973	Dearth	342/30
4,197,538	4/1980	Stocker	343/6.5
4,704,735	11/1987	Swapp et al.	455/68
4,814,711	3/1989	Olsen et al.	324/331
4,835,537	5/1989	Manion	324/30
4,884,208	11/1989	Marinelli et al.	364/460

79 Claims, 14 Drawing Sheets



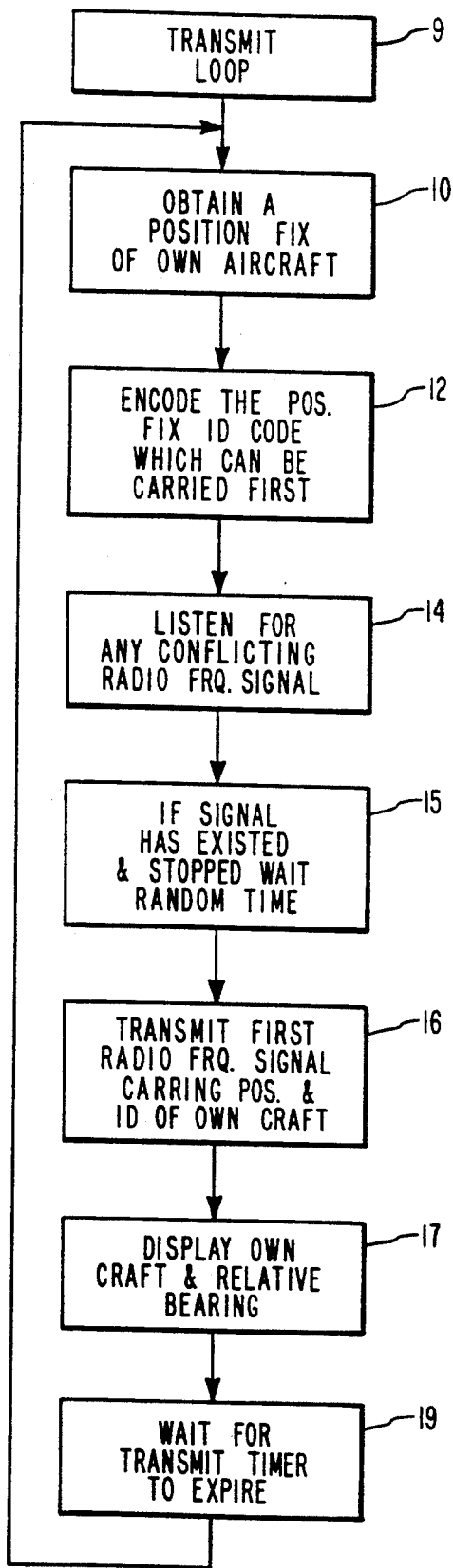


FIG. 1A

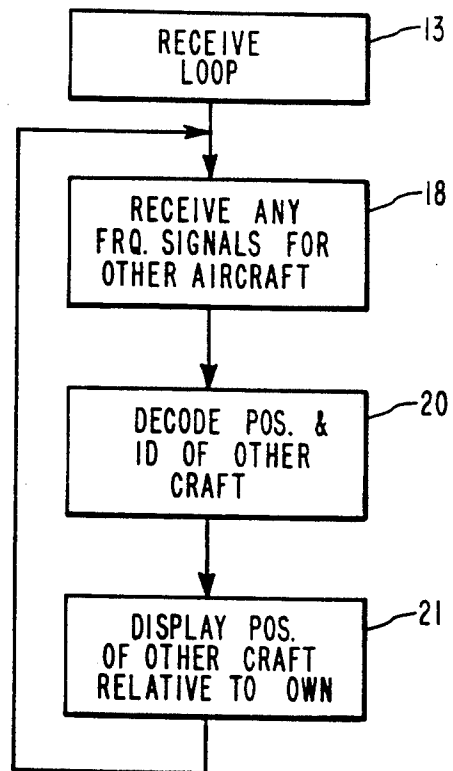
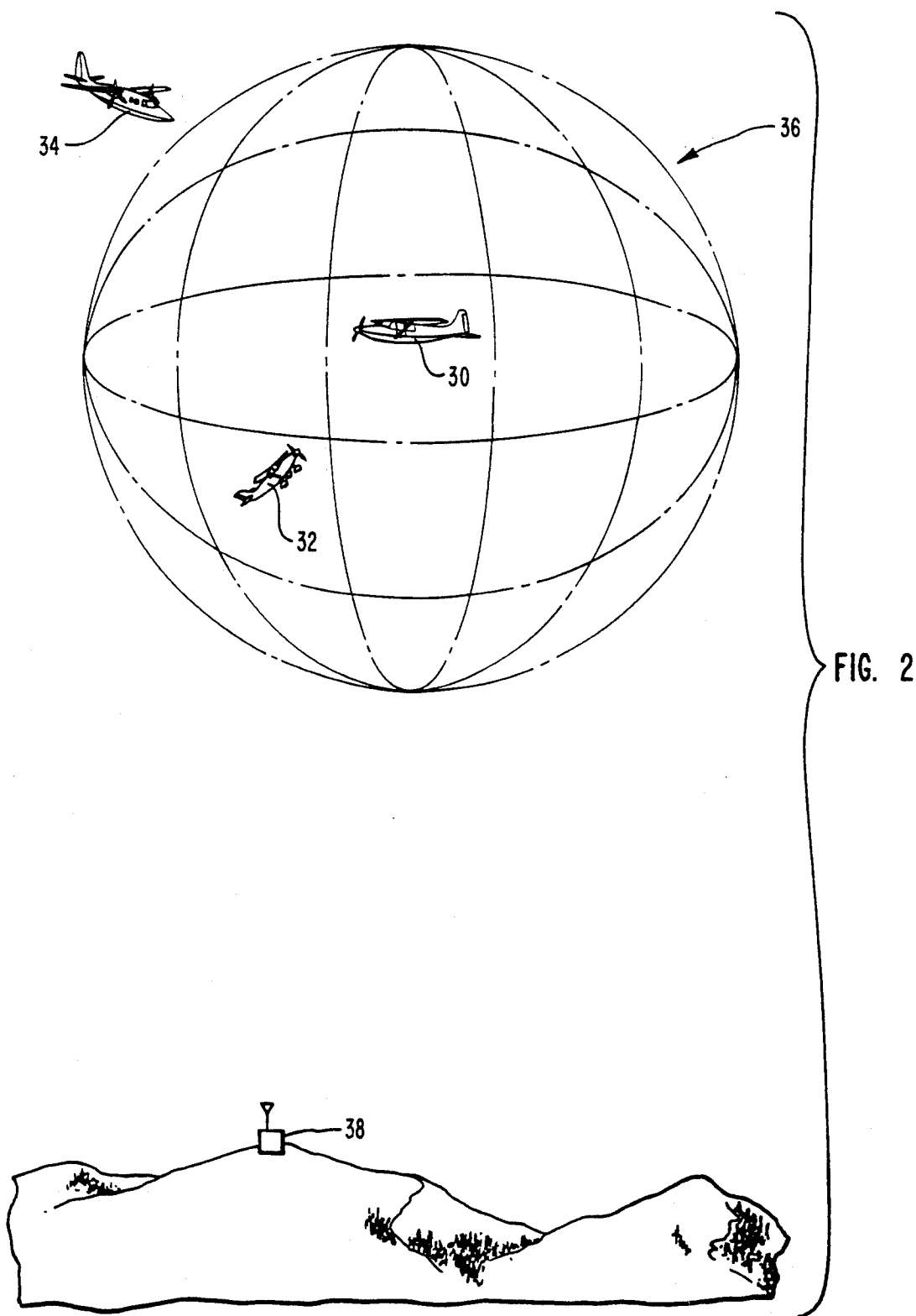


FIG. 1B



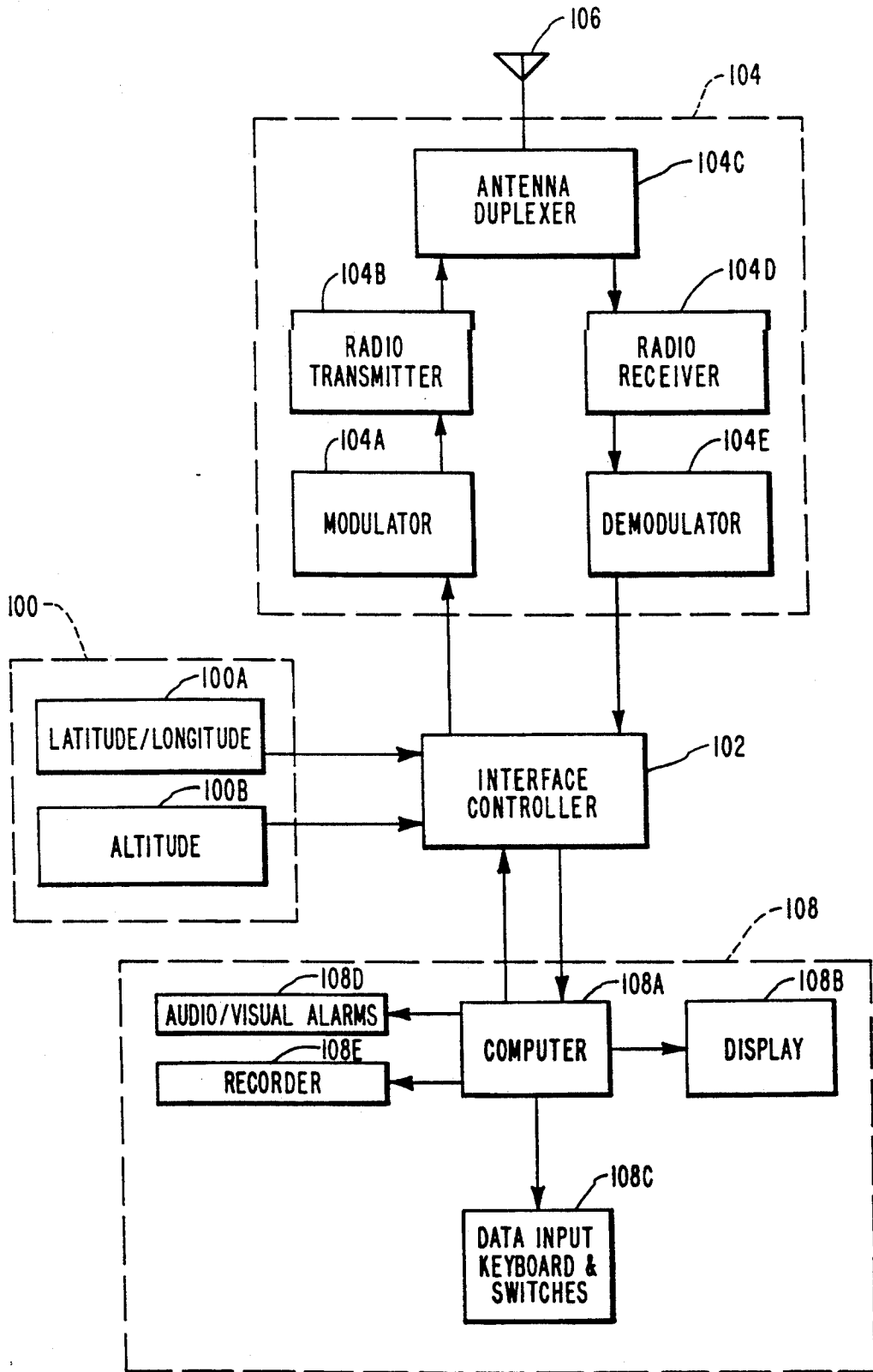


FIG. 3

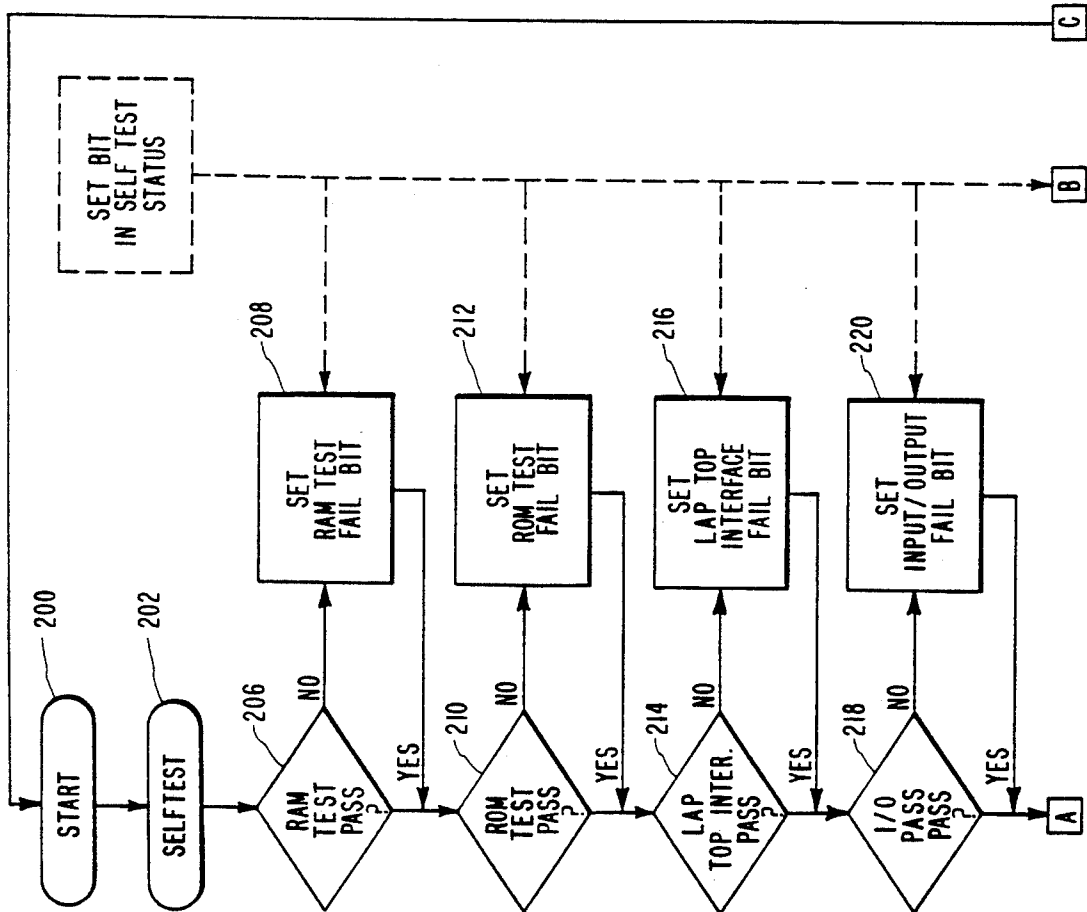


FIG. 4-1

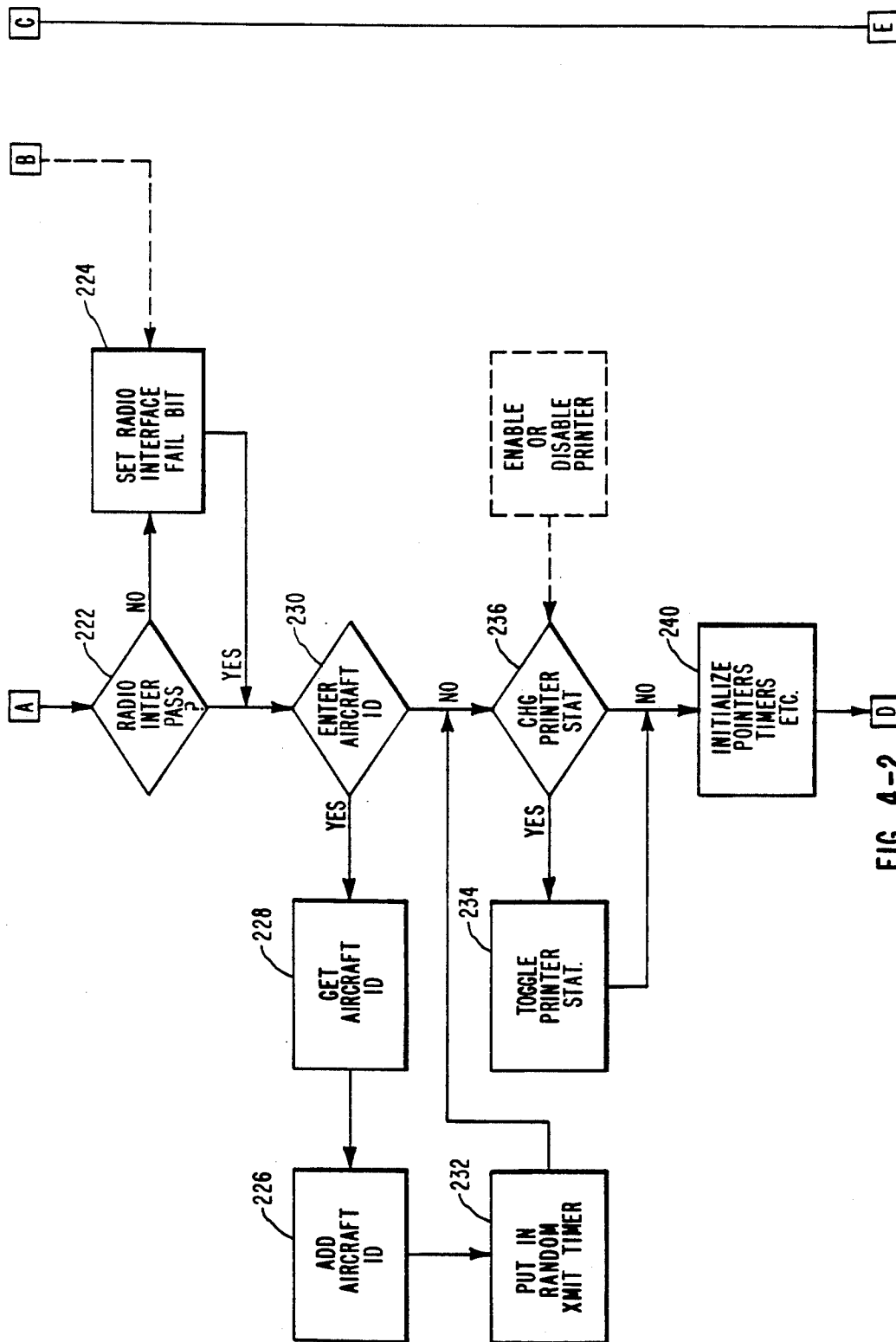


FIG. 4-2

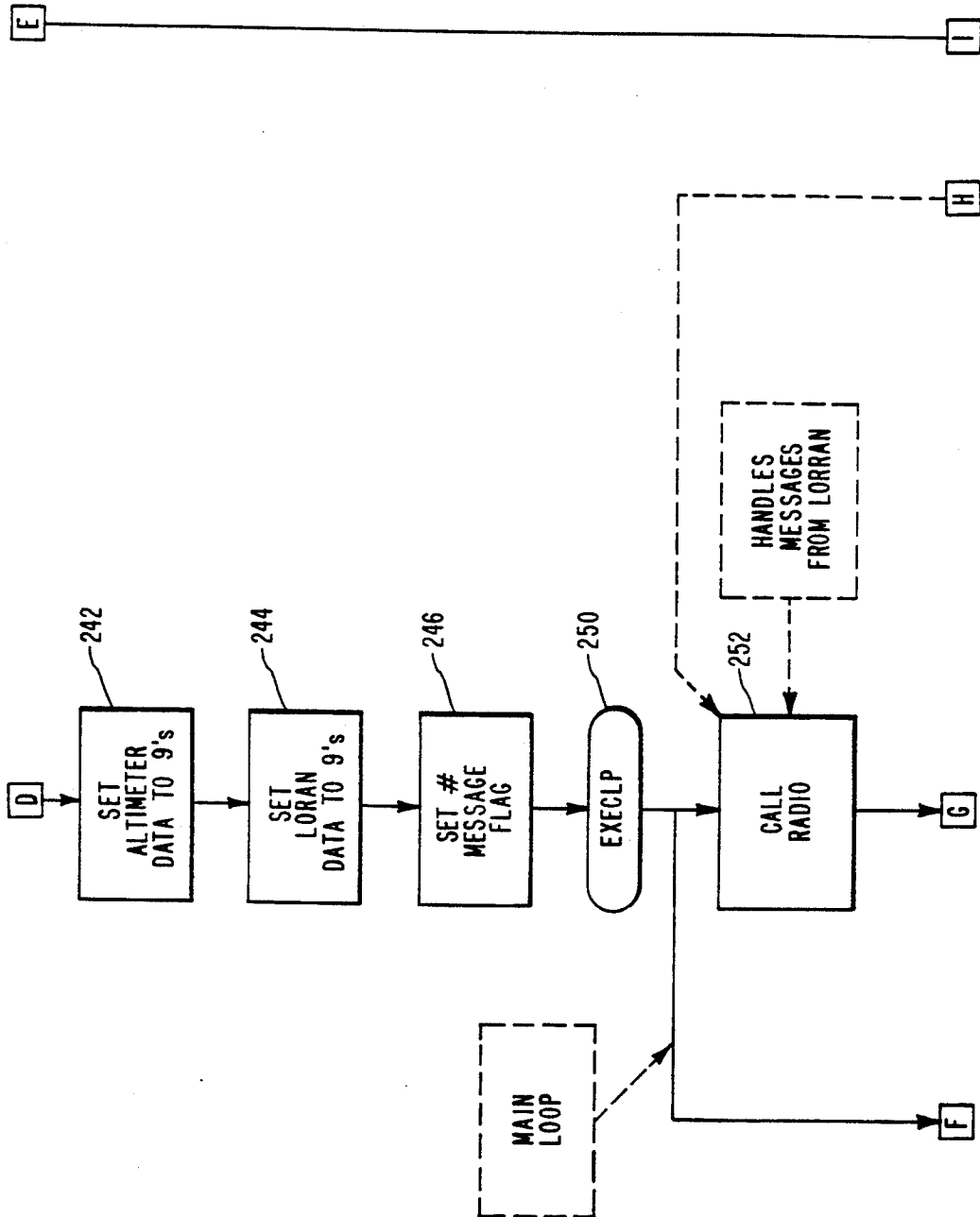


FIG. 4-3

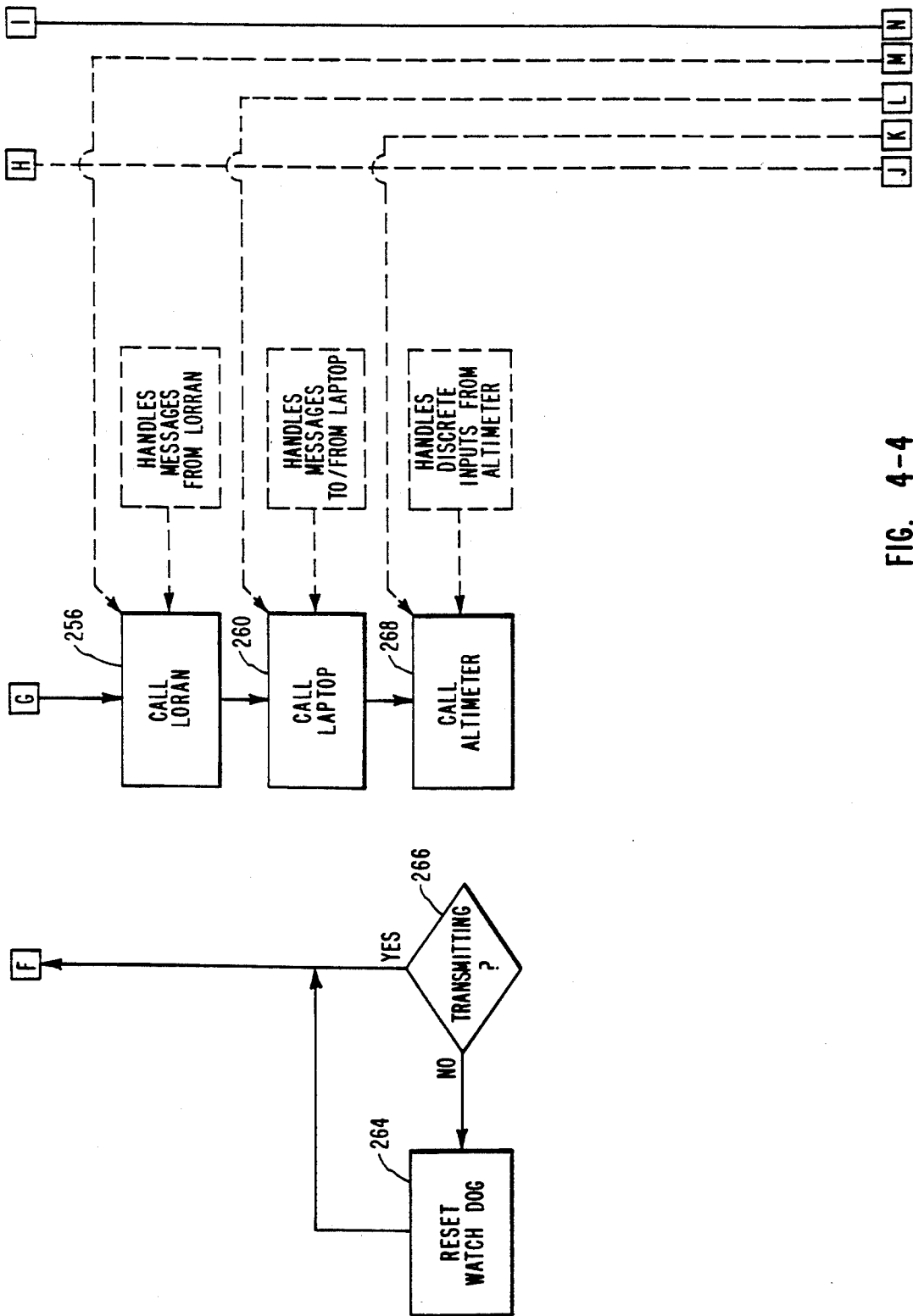


FIG. 4-4

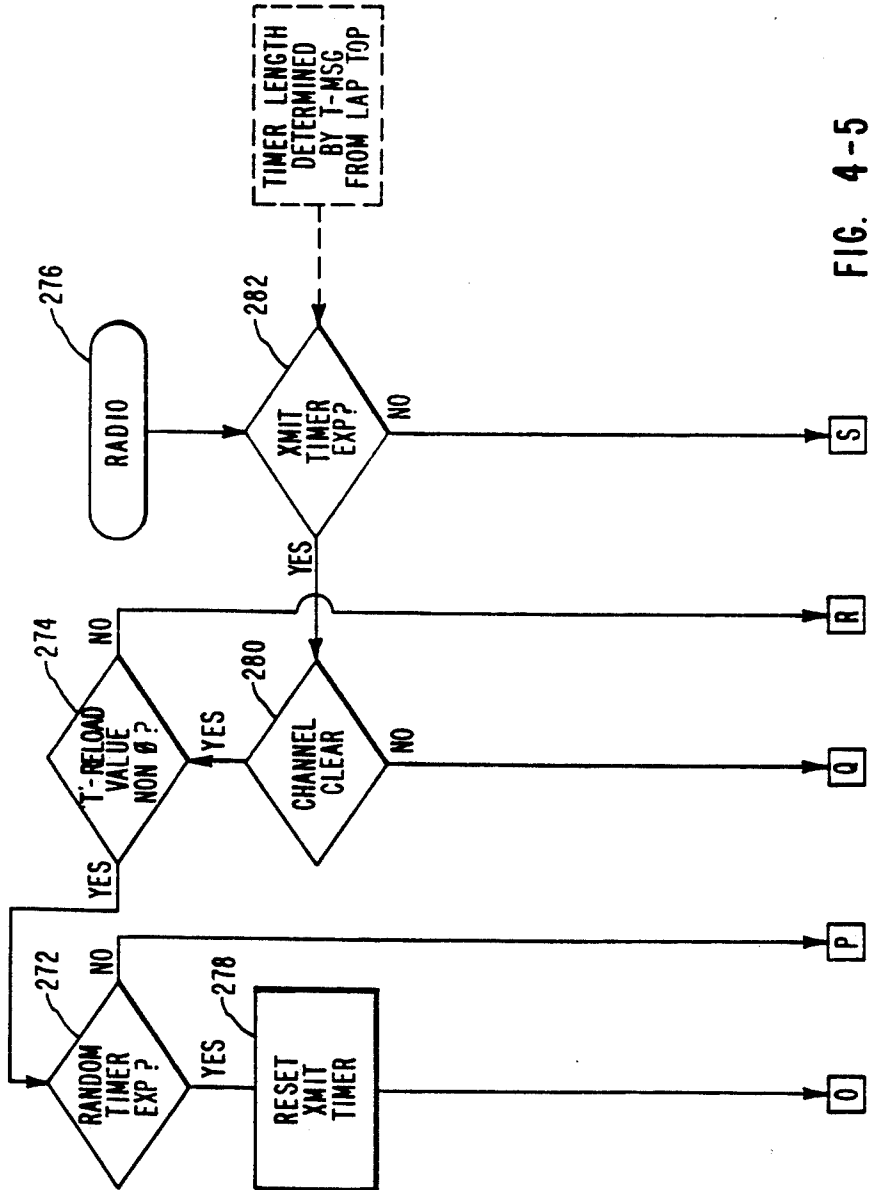
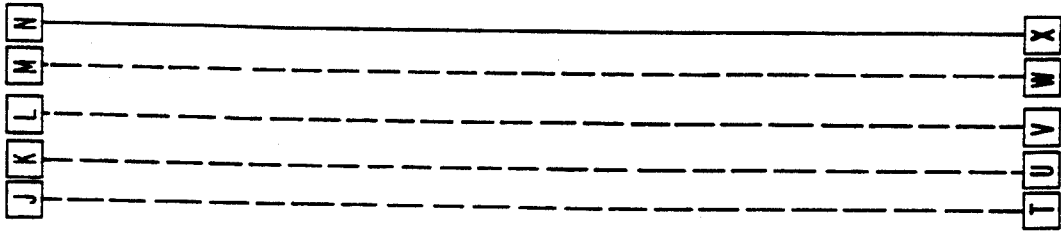


FIG. 4-5

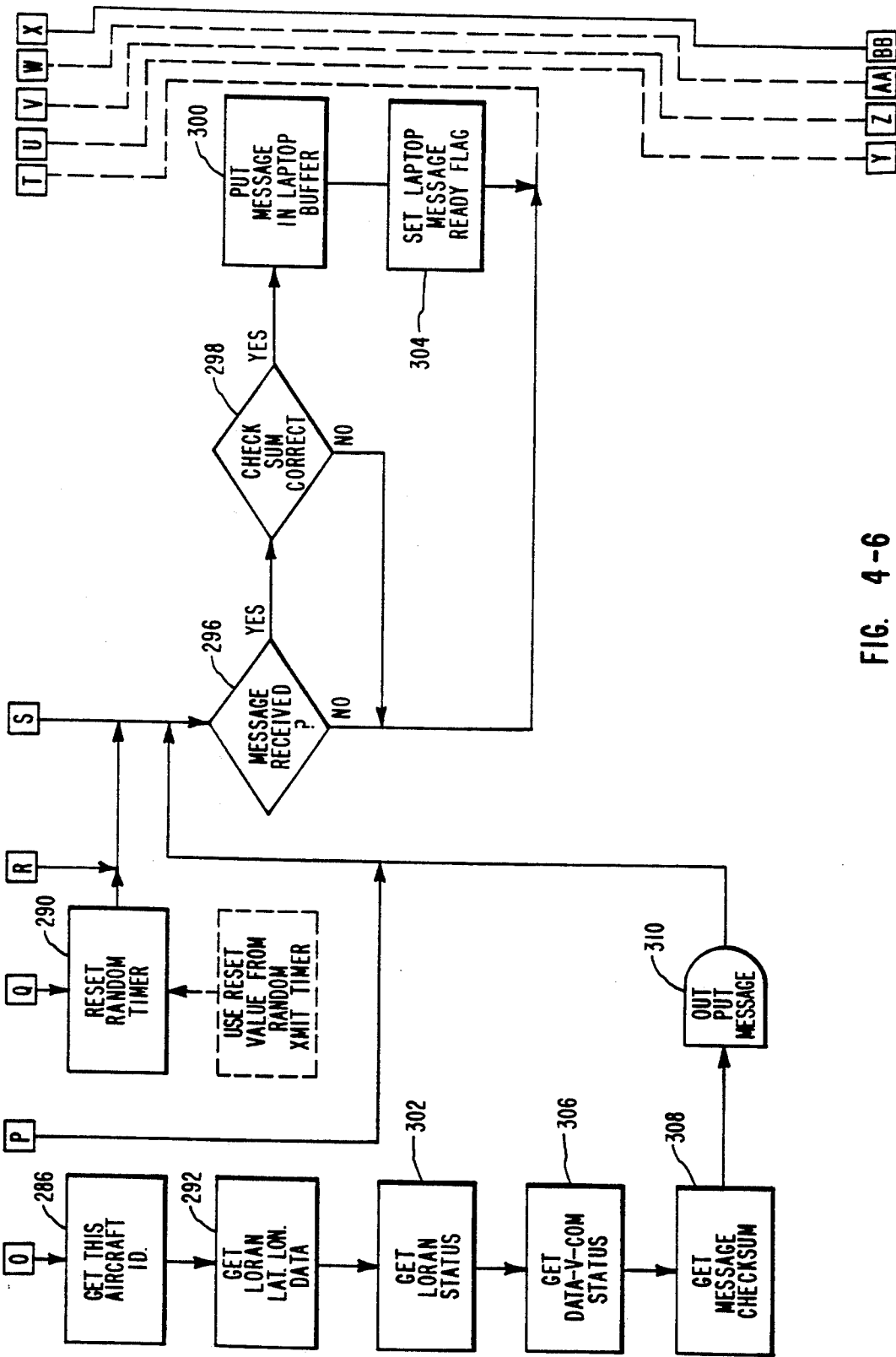


FIG. 4-6

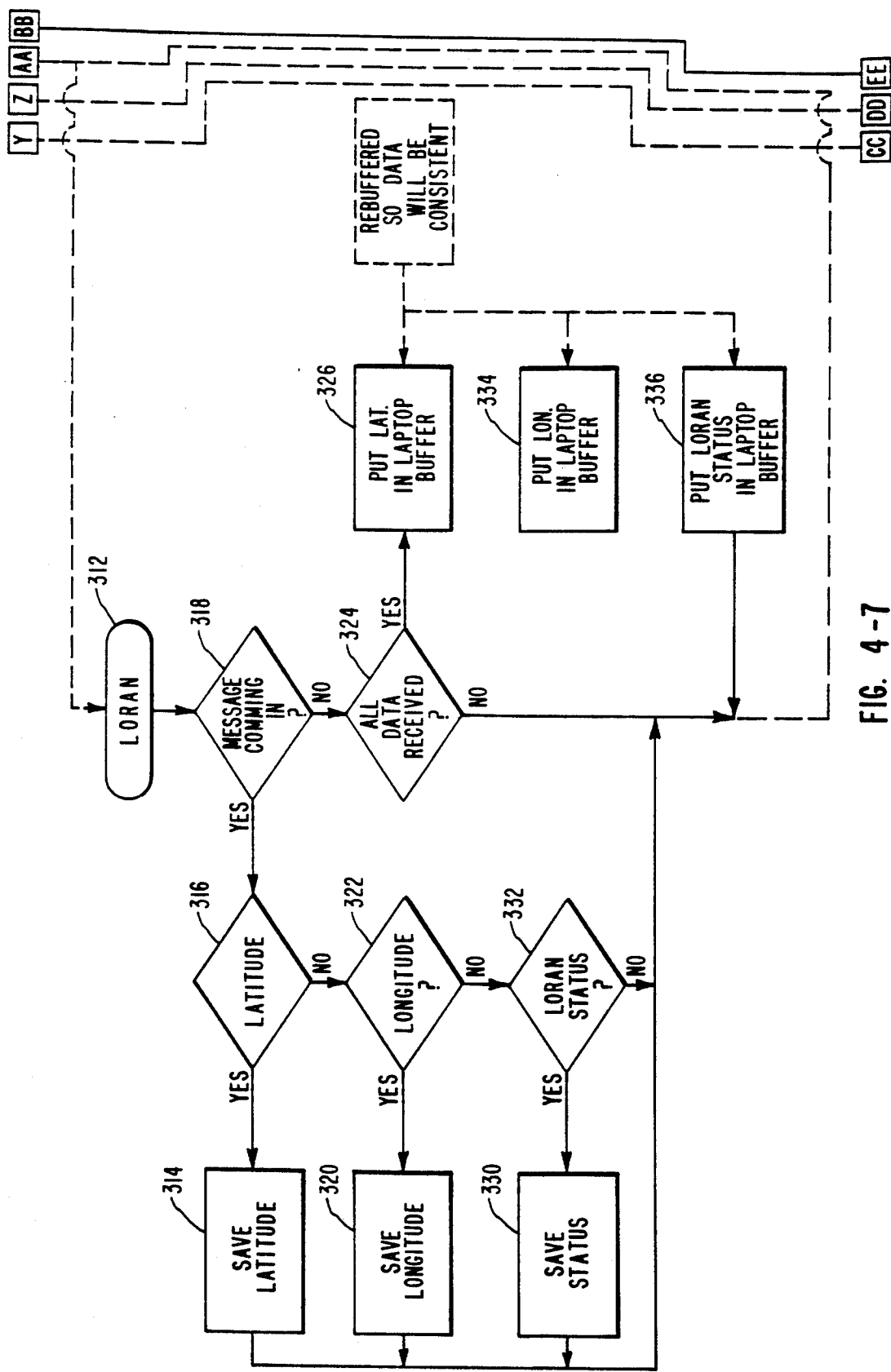


FIG. 4-7

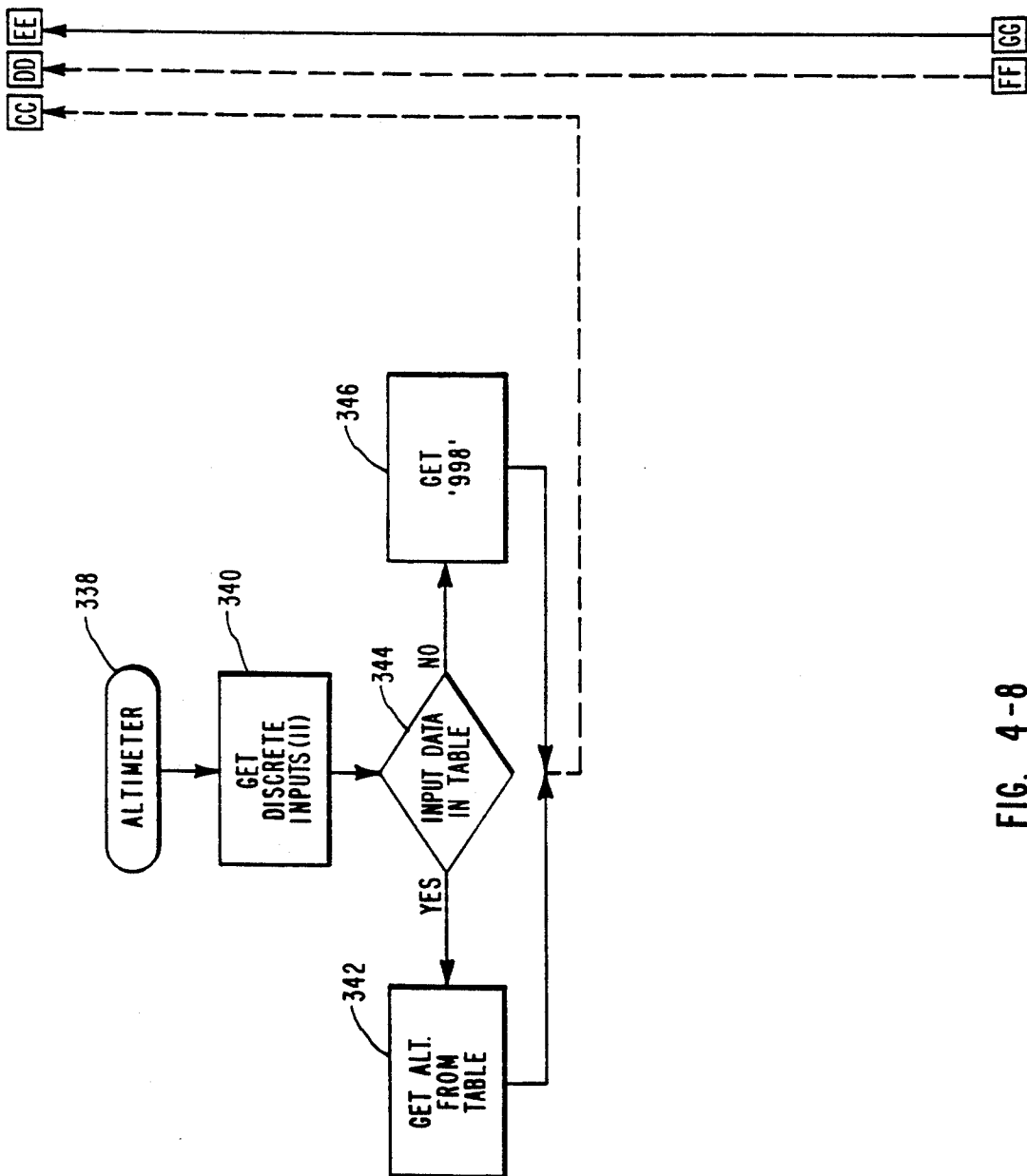


FIG. 4-8

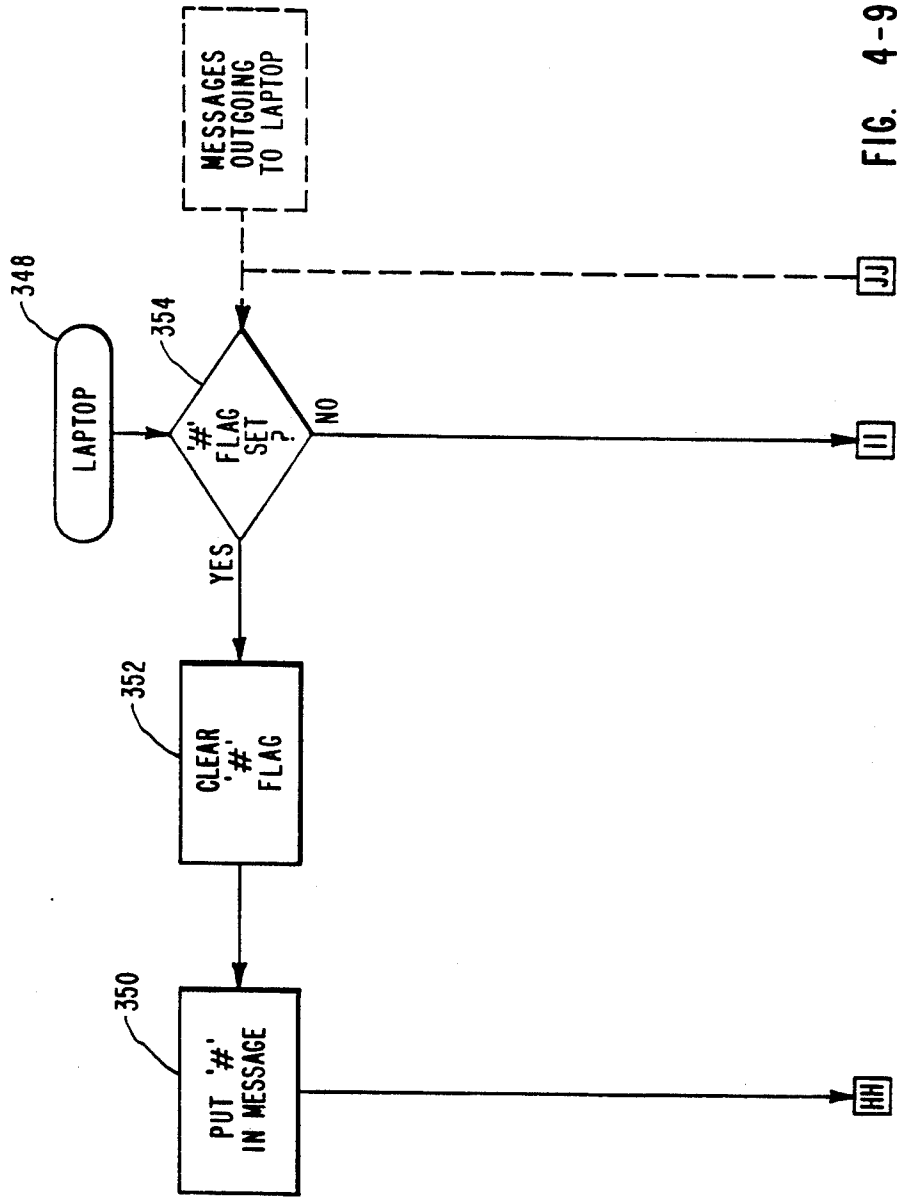
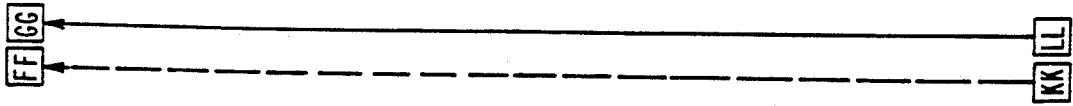


FIG. 4-9

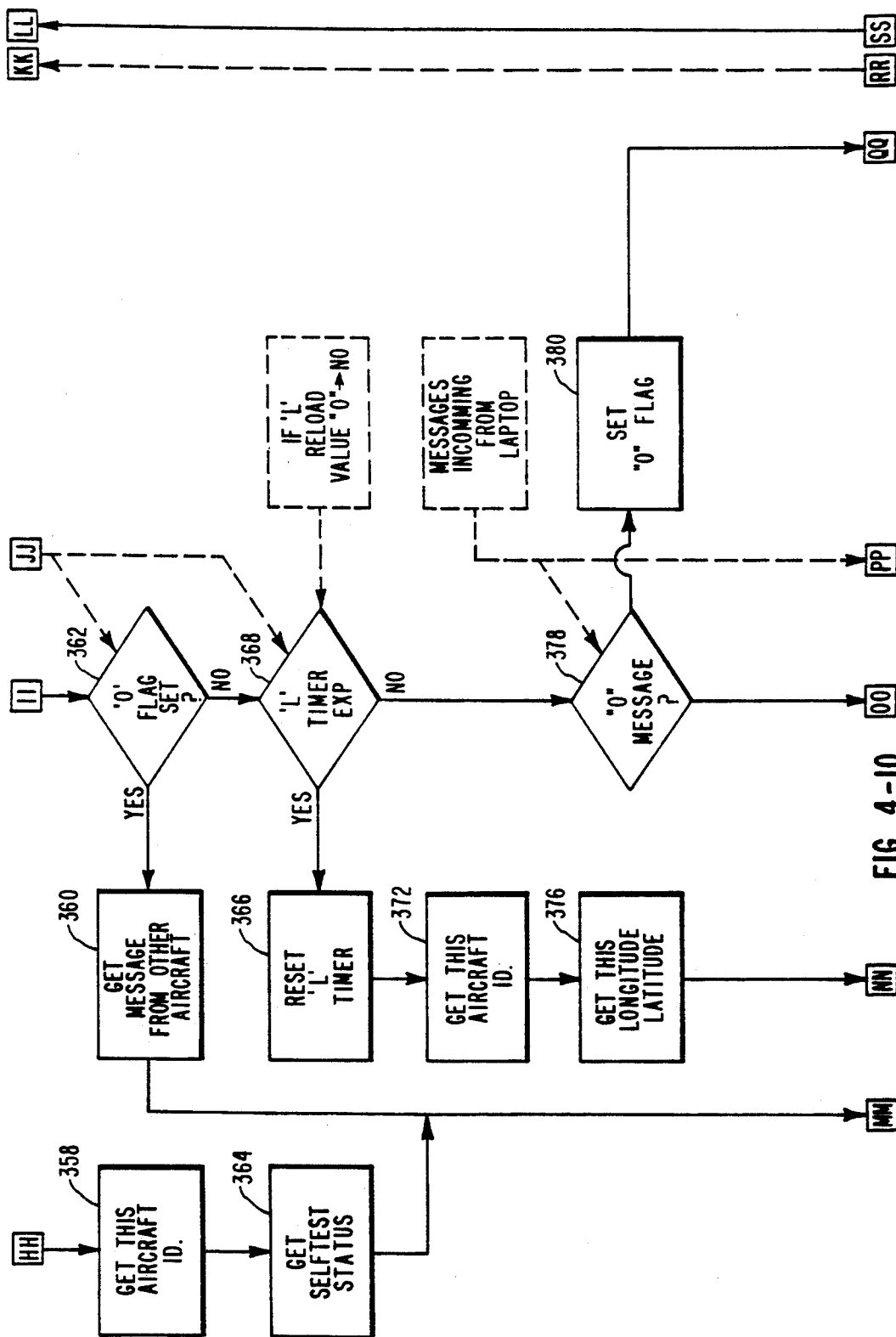


FIG. 4-10

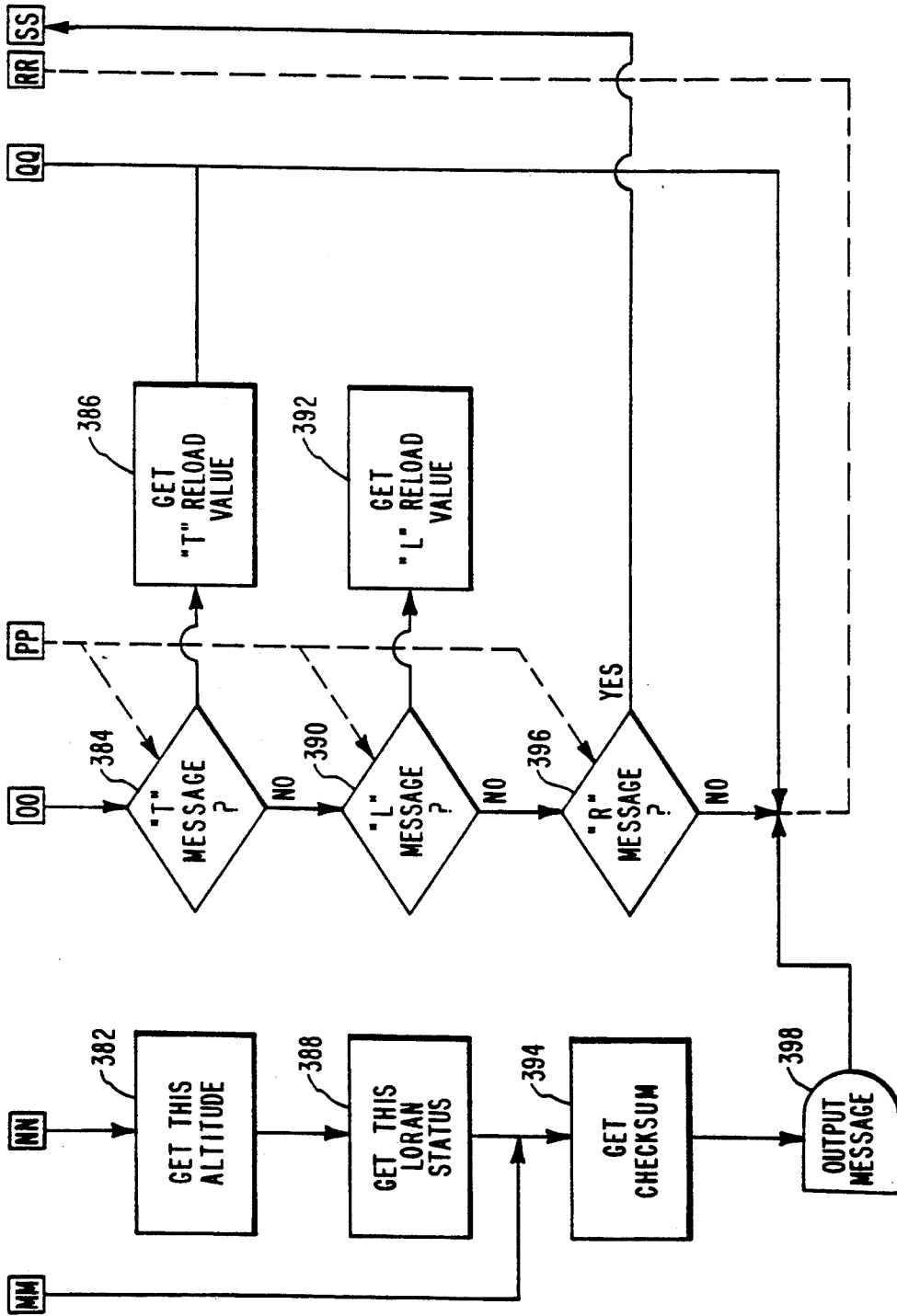


FIG. 4-11

UNIVERSAL DYNAMIC NAVIGATION, SURVEILLANCE, EMERGENCY LOCATION, AND COLLISION AVOIDANCE SYSTEM AND METHOD

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BACKGROUND

1. The Field of the Invention

This invention relates to systems and methods for automatically announcing the position of one or more mobile craft to a receiver positioned at a remote location. More particularly, the present invention relates to systems and methods for tracking the positions of a plurality of mobile craft so as to provide assistance in navigation, surveillance, emergency location and collision avoidance.

2. The Prior Art

Throughout recorded history there has been an ever increasing need for more precise navigational aids. Through the years, the time keeping and position determining devices used for navigation have evolved from sun dials and sextants to sophisticated digital electronic systems capable of providing nearly pinpoint positioning accuracy almost anywhere in the world.

In today's transportation and communications environment, many devices exist that provide specific navigational aid to the operators of air, sea, and land craft. Navigational aids such as the Global Positioning Satellite system (GPS), the Long Range Navigation system (LORAN), and other navigational aids commonly referred to by such various acronyms as: VOR-DME, VOR-TACAN, DECCA, OMEGA, NDB, ILS, MLS, and ADF, are used by craft operators, particularly aircraft pilots, to determine the position of their craft in one, two, or three planes in space. For example, an operator of an aircraft may only desire to know the position of the craft above or below the surface of the earth (altitude or depth) or may desire to know only the position of the craft relative to a fixed terrestrial reference system (latitude and longitude) or the operator may desire to know the craft's position in all three planes in space.

One of the purposes of the various available navigational aids is to allow operators of craft, in particular air and water craft, to avoid collisions between their craft. As is well known, mid-air collisions of aircraft almost always result in disastrous loss of life and property. In an effort to avoid collisions between craft various systems have been implemented in the air and marine transportation industries.

In the United States, government agencies dealing with the air transportation industry have recognized a need to prevent mid-air collisions. In an effort to reduce the occurrence of mid-air collisions, U.S. government agencies have mandated that by the end of 1991 all commercial aircraft with thirty or more seats be equipped with collision avoidance equipment. Several types of collision avoidance equipment have been devised and are classified as either "active" or "passive" systems. These systems are commonly designated as

Traffic Alert and Collision Avoidance Systems, also referred to as TCAS.

TCAS has been designed as a primary collision avoidance system for commercial aircraft and has received the United States Federal Aviation Administration's approval. An "active" TCAS system (TCAS II or III) provides the capability to interrogate other nearby transponder-equipped aircraft to elicit a responsive reply, while "passive" TCAS systems (TCAS I) simply eavesdrop on nearby interrogation replies from other transponders. In order to detect a "bearing," a TCAS system requires an expensive directional antenna.

The TCAS method is dependant upon either a secondary or beacon surveillance radar system, or other transponder (S) type interrogations in order to elicit a responsive reply. It is also transponder dependant, meaning that any aircraft not equipped with a transponder will not be detected, and any transponder not coupled to an altitude encoder will not deliver altitude information.

Recently, serious questions have been raised about how "safe" TCAS is in practice. TCAS operates in a narrow band of Air Traffic Control (ATC) radio frequencies in the microwave region of the spectrum. Thus, because of the number of interrogation requests and replies elicited, and the amount of information needed to be processed both TCAS and the ATC system may be "overwhelmed," and their operation sufficiently degraded to the point of "saturation," where the number of aircraft under the control of ATC exceeds the capacity of the system. Already, computer overloads and radar shutdowns have occurred at some busy locations. There is also the danger of a malfunction occurring, such as an item of interrogation equipment becoming stuck in the transmit mode, thereby "locking up" the entire system.

The high cost of the on board TCAS equipment makes its installation prohibitively expensive to most aircraft other than commercial aircraft. Moreover, TCAS III has not yet been fully developed. Even further, TCAS II and III are designed with the commercial aviation market in mind, but of the 215,926 active aircraft registered in the United States in 1988, the commercial aviation fleet only amounted to about three percent (3%) of the total. Thus, it is apparent that improved collision avoidance systems need to become more accessible to a much larger portion of the aviation industry, as well as to land and sea-based craft.

In view of these drawbacks and difficulties it would be an advance in the art to provide a complete traffic control system not requiring radar, and which provides a system and method for announcing the position of a craft to a remote receiver such that the position of the craft can be continuously monitored and collisions with other craft avoided. It would also be an advance in the art to provide a system and method for collision avoidance which may be used in addition to, and without interference with, preexisting collision avoidance systems and which can be economically implemented in the general aviation and marine industries so that, for example, with a single frequency select switch seagoing vessels could monitor air or land-based traffic and vice versa.

It would be a further advance in the art to provide a system and method which allows the operator of a first craft to monitor the position of a plurality of other craft within a predetermined range of the first craft and to alert the operator of a craft of a potential collision be-

tween craft as early as possible. It would be yet another advance in the art to provide a system and method for announcing the position of craft to other craft which are within a predetermined range which is reliable, does not distract an operator of a craft from other duties, and provides short and long range navigational assistance to the operator of a craft.

It would be a still further advance in the art to provide a system and method of tracking the position of one or more craft within a predetermined range of a location anywhere on the earth such as at a remote airstrip, or beneath the surface of the ocean, or in space. It would be an even further advance in the art to provide a collision avoidance system which does not become saturated in areas of heavy traffic and which is capable of assisting with the landing or mooring of craft operating under poor visibility conditions. Still further, it would be an advance in the art to provide a system and method for tracking the position of a number of craft, for example a fleet of land craft which are carrying out tasks such as delivery of goods or people or other tasks.

BRIEF SUMMARY AND OBJECTS OF THE INVENTION

In view of the above described state of the art, the present invention seeks to realize the following objects and advantages.

It is a primary object of the present invention to announce the position of a craft to a remote receiver such that the position of the craft can be monitored and collisions with other craft avoided.

It is also an object of the present invention to provide a system and method for collision avoidance which may be used in addition to, and without interference with, preexisting collision avoidance systems.

It is a further object of the present invention to provide an emergency location system for distressed or disabled craft.

It is a still further object of the present invention to provide a collision avoidance system which is economically implemented in the general aviation or marine industries.

It is another object of the present invention to provide a system and method which allows the operator of a first craft to monitor the position of a plurality of other craft within a predetermined range of the first craft.

It is yet another object of the present invention to provide a system and method which will alert the operator of a craft of a potential collision as early as possible.

It is another object of the present invention to provide a system and method for announcing the position of any craft to other craft which are within a predetermined range which is reliable and does not distract an operator of a craft from other duties unless operator attention is necessary.

It is a further object of the present invention to provide a system and method for providing short and long range navigational assistance to the operator of a craft.

It is a still further object of the present invention to provide a system and method which allows the operator of a first craft to continually monitor the positions of a plurality of other craft within a predetermined range.

It is another object to the present invention to provide a system and method of tracking the position of one or more craft within a predetermined range of a location anywhere on the earth such as at a remote

airstrip as long as one is operating within the constraints of primary navigation inputs.

It is another object of the present invention to provide a system and method for tracking the position of each of a plurality of craft which may be operated within an independent fleet of craft.

It is a further object of the present invention to provide a collision avoidance system which operates well in areas of heavy traffic.

It is still another object of the present invention to provide a system and method for assisting with the landing or mooring of craft under poor visibility conditions.

It is yet another object of the present invention to provide a system and method for collision avoidance which presents appropriate information to the operator of the craft in an easily comprehensible manner and which provides traffic alerts and resolution advisories.

It is another object of the present invention to allow a permanent record to be made of the movement of any craft within a predetermined range of a monitoring receiver.

Another object of the present invention is to provide a direct pilot-based system of traffic control and communication, and to provide a traffic control system and method which does not require radar and that will not interfere with other existing traffic control systems which are based on radar.

A further object of the present invention is to provide a comprehensive monitoring, tracking and communication system and method to promote inter-craft operational safety.

These and other objects and advantages of the invention will become more fully apparent from the description and claims which follow, or may be learned by the practice of the invention.

The present invention allows the position of a first craft to be monitored at a location remote from the craft. The position of the craft can be monitored at a stationary receiver or from on board another craft. Since a craft equipped with the present invention can monitor the position of other properly equipped craft, collisions therebetween can be avoided.

In the case where the present invention is fully implemented, i.e., all the craft in a fleet are equipped with embodiments of the present invention, all the craft are able to monitor the position of all other craft within a predetermined range. Moreover, each craft is able to accurately transmit its own position so the present invention is a great aid to navigation.

The present invention includes means, on board a first craft, for receiving or determining the position fix of the first craft. Any number of position-determining devices may be used. For example, the position of the first craft may be obtained by using an altimeter or depth finder, the widely accepted LORAN navigation system, or GPS position determining techniques.

The position of the first craft is encoded, preferably as a digital signal, so that the position information of the first craft can be carried by a radio frequency signal. Preferably, both the craft position fix, craft identification code, and other relevant information such as radio frequency number, directory advisories, and equipment error flags are encoded so they can be carried by the radio frequency signal.

A large number of craft often operate within a limited range and may all transmit position information on the same radio frequency. In order to avoid interference

with conflicting radio frequency signals, it is preferred that conflicting signals are listened for prior to transmitting the radio frequency signal.

Importantly, the present invention does not require interrogation by a land-based, or other transmitter before position information will be transmitted by a craft. The radio frequency signal carrying the position information, and preferably other messages and identification information, is regularly transmitted on a continuous basis.

The radio frequency signal which is transmitted (carrying at least the position information for the craft) is received by another craft or a stationary monitoring receiver at a remote location. The received radio frequency signal is decoded to extract the craft position, identification and other information therefrom. Using the position information, the position of the other craft is displayed.

If the radio frequency signal is received on board another craft, the position of the first craft is of great help to the operators of that craft to navigate, communicate and especially to avoid possible collisions. In order to aid the operators of that craft, the relative positions of each craft are also determined and preferably displayed.

By displaying the positions of the two craft as the craft move, a heading for each of the craft can be derived and displayed and the possibility of a collision assessed. In accordance with the present invention, if a collision is possible between two craft on their present headings the operators of each craft are alerted so appropriate action can be taken. It will be appreciated that the greatest benefit from the present invention accrues when all craft have an embodiment of the present invention on board. When all the craft are so equipped, all of the craft can monitor the position of all other craft which are within a predetermined range.

Importantly, the present invention has applications other than providing collision avoidance advice. The present invention may be used as a navigational aid, for example, by automatically displaying the craft's position on a terrain map or graphically showing a landing approach. Moreover, with appropriate navigation information, the present invention may be implemented economically and some embodiments of the invention may be used anywhere in the world. Also, the embodiments of the present invention can be used to locate and direct precision emergency location and rescue efforts in the case of disabled craft.

Significantly, the present invention does not rely on interrogation by other craft to transmit its position fix. Rather, the radio frequency signal carrying the position information is transmitted regardless of the presence of any other craft or monitoring receiver. The strength of the radio frequency signal is limited so that it is received only within a predetermined range which is of interest. For example, the predetermined range of the radio frequency signal may be five, ten, thirty, or more miles. Moreover, rather than requiring a sophisticated radio direction finding antenna, the radio frequency signal of the present invention is preferably transmitted omnidirectionally so that all craft within the predetermined range will be able to extract the position information therefrom.

BRIEF DESCRIPTION OF THE DRAWINGS

In order that the manner in which the herein-recited and other advantages and objects of the invention are

obtained can be appreciated, a more particular description of the invention briefly described above will be rendered by reference to a specific embodiment thereof which is illustrated in the appended drawings. Understanding that these drawings depict only a typical embodiment of the invention and are not therefore to be considered limiting of its scope, the invention will be described and explained with additional specificity and detail through the use of the accompanying drawings in which:

FIGS. 1A-1B are high level flow charts describing the presently preferred steps carried out by the system described herein.

FIG. 2 depicts three aircraft as an example of the operating environment of the described embodiment of the present invention.

FIG. 3 is a block diagram showing the major functional components of the described embodiment.

FIGS. 4-11 through 4-11 provide a flow chart describing the steps carried out by the preferred embodiment of the apparatus of the present invention.

DETAILED DESCRIPTION OF THE PREFERRED EMBODIMENT

Reference will now be made to the drawings wherein like structures will be provided with like reference designations.

1. General Discussion

As discussed above, the widely promoted and available collision avoidance systems, particularly those available to the aviation industry, inherently have several drawbacks. A major drawback is the fact that the cost of installing the necessary on board equipment may exceed the cost of an entire small aircraft. Thus, economic consideration prevents such systems as TCAS, from being adopted by the great majority of the general aviation industry.

Moreover, TCAS equipment requires complex on board computers in conjunction with radio direction finding equipment. Since TCAS equipped aircraft are capable of interrogating all other aircraft in close proximity, the result is that a large number of interrogations will occur in high density air space. Eventually, the radio transmission media will become saturated thereby rendering both TCAS and secondary surveillance radar inoperative.

Even when working optimally, the TCAS and other collision avoidance systems provide only a transponder code assignment for identification, gives only relative heading information and if properly equipped, altitude information. In addition, the entire system is interrogation dependent.

In contrast to the previously available collision avoidance systems, the present invention makes use of existing ground based or satellite based equipment which has already been widely accepted, rather than requiring new dedicated equipment. Moreover, the present invention provides navigational aid and emergency location functions not contemplated in previously available systems. Importantly, the collision avoidance, navigation aid, and emergency location functions are all provided economically and thus the necessary equipment can be installed on board even small privately owned aircraft. Still further, the present invention can be applied in space, air, land, and sea craft and provide collision avoidance, navigation, and emer-

gency location functions in nearly every environment, including IFR conditions.

The present invention is able to achieve its great advantages over the previously available systems by utilizing one of the existing position determining systems, or their equivalent, that is now available or currently being implemented. For example, using the LORAN system a craft, or in the case of portable LORAN receivers an individual, can immediately obtain a position fix measured in latitude and longitude, in many areas of the world. Also, the GPS scheme, when fully implemented, will allow a position fix measured in latitude, longitude, and altitude, to be determined anywhere in the world with great precision. These existing position determining systems, and others when available, are used by the present invention to provide advantages not heretofore available in the art.

The described presently preferred embodiment of the invention utilizes the LORAN position determining system. While one form or another of the LORAN system has been known for many years, especially in the marine environment, it has recently gained world-wide acceptance among other types of craft operators, and particularly among aircraft operators. The LORAN system provides more useful navigational information than many other navigational instruments now in popular use, especially in the aviation environment. It has proven to be a highly reliable and safe system and has even been conditionally approved by the FAA for non-precision approaches to approved landing strips.

The precision of the LORAN system generally ranges from 60 to 600 feet depending on the location of the craft in relation to the geometry of several widely spaced low frequency radio transmitters which provide the necessary position fixing signals. Because of the precision and versatility of the LORAN system, LORAN receivers are available which include computers, databases and displays providing mapping and navigational aids to the operator of a craft as the craft travels. For example, LORAN equipment intended for aviation use is often provided with databases that are updated monthly to provide all current airport frequencies, runway orientation and length, airport elevations, VOR, restricted and prohibited flight areas, and other useful information.

Furthermore, some aviation LORAN receivers are able to immediately locate several of the nearest airports in times of emergency. LORAN equipment installed on aircraft is able to compute ground speed, wind direction and speed, tracking error, offer tracking and vertical navigation advisories, and report minimum safe altitudes and degrees of magnetic variation.

Excellent LORAN coverage is now available throughout most of North America. LORAN coverage is partially available in most other areas of the world including the North Atlantic, the United Kingdom, Europe, the Mediterranean Sea, North Africa, the Persian Gulf, Saudi Arabia, and Japan. The relatively low cost and high benefits of installing LORAN transmitters make the LORAN system attractive to governments of both industrialized and developing nations. Furthermore, upon its full implementation, the GPS scheme will provide even greater benefits. Since an existing position determining system is not available everywhere, the term "operational range" will be used to denote when a craft is in an area in which an existing position determining system can be effectively used.

Complete information concerning the LORAN system can be obtained from the publications Melton, L., *The Complete Loran-C Handbook* (Marine Publishing Co. 1986) and Sweeny, D. J., "Learning About Loran," *Radio-Electronics Magazine* 50-58, 69 (May 1987) which are incorporated herein by reference. Likewise, the GPS scheme may also be used in accordance with the present invention. Further information concerning GPS can be obtained from the publications entitled "Introduction to Global Positioning," *Civil Engineering* 16-20 (January/February 1987), Ashjaee, J., "Global Positioning System: Refined Processing for Better Accuracy," *Sea Technology* 20, 22-25 (March 1986) and Enge, P. K. et al, "Differential Operation of the Global Positioning System," *26 IEEE Communications Magazine* 48-60 (July 1988) which are incorporated herein by reference.

It will be appreciated that the present invention has application in many different circumstances. Thus, as used herein the term "craft" is intended to include any stationary object, any mobile animal or person, or any mobile apparatus or vehicle. For example, land, sea and air craft are specifically intended to fall within the meaning of the term "craft."

Also, as used herein, the term "monitoring receiver" means any device or apparatus invention which can monitor the position of a craft which is transmitting position information in accordance with the present invention.

The present invention comprises an apparatus and method for announcing the position of a craft to a monitoring receiver, either stationary or mobile, remotely located from the craft. In one application, the present invention can be used to track the position of one or more craft at a stationary receiver. In another application, the present invention can be used by one craft to track one or more other craft. With the present invention in place, as a first craft tracks all the other craft positioned within a predetermined range of the first craft, and the operator of the first craft is apprised of the position of all the other craft, collisions between the craft can be avoided.

The presently preferred embodiment of the invention which is described herein is intended to be used primarily in the aviation industry. Those familiar with the aviation industry will appreciate that the problems of collision avoidance are particularly troublesome in today's crowded airspace. Adding to the problem of collision avoidance in the aviation industry are: the limited operator's field of vision in an aircraft cockpit; the speeds of the craft on a collision course provide very little time for a pilot to react and take corrective action; and an aircraft on a collision course may come from any direction in space.

Also a universally useable or mandated collision avoidance system must be economically feasible for both small, modestly priced aircraft carrying just one or two persons and large multi-million dollar aircraft carrying hundreds of passengers and/or freight. Thus, at the current time, a most urgent need for the present invention lies within the aviation industry.

Nevertheless, the present invention can be used as a position announcing and/or collision avoidance system in other demanding situations such as with water craft. Also, the present invention can be applied where a dispatcher needs to track the movement of a plurality of vehicles in a fleet, for example, motor trucks as their movements are coordinated for making deliveries and-

/or pickups, buses, trains, construction and other land based equipment.

2. Detailed Description in Reference to FIGS. 1-4

In order to describe the presently understood best mode for making and using the present invention, the presently preferred embodiment of the present invention, as intended to be used in an aircraft environment will be described. As will be appreciated, the application of the present invention to air traffic monitoring and collision avoidance is one of, if not the most, demanding of the contemplated applications of the present invention because of, among other things, the speeds of the craft involved and the desirability to monitor three dimensional space as opposed to two dimensional space in the case of water and land craft.

Referring now to FIGS. 1A-1B, the preferred general method used by the system of the present invention will be explained.

As shown at 10 in FIG. 1A, the initial step of the transmit loop 9 is to obtain a position fix for one's own craft, also sometimes referred to herein as a first craft. The position fix is obtained using apparatus on board the craft. It is preferred that the position fix be obtained using an existing automated navigational system which will provide the position of the craft relative to a fixed reference, for example, latitude and longitude. This arrangement is in contrast to other collision avoidance systems which merely determine the distance between two craft and not their position relative to a fixed reference system. The LORAN system and the GPS system are among the preferred existing external automated navigational systems.

While navigational systems which are capable of providing a position fix in two dimensions (e.g., LORAN) or three dimensions (e.g., LORAN in combination with an altitude encoder or GPS) are preferred, in some applications of the present invention it may only be necessary to obtain a position fix in one dimension or plane of space. For example, in one application of the present invention, an operator of an aircraft may only need to know if another craft is operating at the same altitude. While such information is only a portion of that which the described embodiment of the present invention can provide, it is of use to alert the operator of an aircraft. If another aircraft is operating within a close range at the same altitude, an operator of the aircraft should be warned that a collision with another aircraft is possible if precautions are not taken.

Thus, the position fix can be obtained from an altimeter only (for a one dimensional position fix), from a LORAN receiver (for a two dimensional position fix), from both a LORAN receiver and an altimeter or a GPS device (for a three dimensional position fix). Other devices can also be used to obtain a position fix.

As indicated at step 12 in FIG. 1, the position fix which was obtained in step 10 is encoded so that it can be carried by a radio frequency signal. In addition to the position fix, an identification code (such as an aircraft registration number) and/or other messages are also encoded. It is preferred that the position fix and the identification code and other messages be digitally encoded so that they can be transmitted at a relatively high baud rate as will be described in detail below.

In the United States, the radio frequency allocation of 1.6 Gigahertz has been reserved for aviation use. Thus, it is presently preferred, but not required, to utilize the 1.6 Gigahertz frequency allocation. Since all aircraft

utilizing the present invention preferably utilize the same frequency, step 14 provides that any conflicting radio frequency signals are listened for.

One of the significant advantages of the present invention is that, as opposed to previously available collision avoidance systems, a craft transmits its own position fix and identification code information without being interrogated.

Since many craft may be using the described embodiment of the present invention, all using just one or a few radio frequency allocations, avoidance of conflicting radio frequency signals is necessary. In use, some conflicts may occur in the described embodiment in areas where craft are highly concentrated. In the described embodiment, the position fix and identification code information is regularly and repeatedly transmitted. Thus, if an attempt at transmitting the position fix and identification code is postponed or "walked on" by another transmission, the information will be repeated in a short period of time.

One preferred arrangement for preventing interference between conflicting radio frequency signals is to first listen for conflicting signals, as at step 14, and if a conflicting signal has existed, then wait a random period of time after the detection of a conflicting radio frequency signal, as at step 15. The length of the random period of time can be, for example, derived from the registration number of the aircraft, a random generation, or some other means. Alternatively, after the detection of a conflicting radio frequency signal, a predetermined period of time may be waited before transmission.

Additional details concerning the settlement of a conflict between radio frequency signals can be devised by those skilled in the art and further information concerning the preferred example is provided later in this disclosure. It will be understood that various schemes, including the use of scanned multiple frequencies (speed spectrum technology), higher transmission rates, and other non-interrogation schemes, can be devised and used with the present invention.

At step 16 in FIG. 1, the radio frequency signal carrying the position fix and identification code information is transmitted and the pilot's own craft and relative bearing are displayed at step 17. The system then waits at step 19 for the transmit timer to expire, before re-entering the loop for another transmission.

Whether it is desired to avoid collisions between multiple craft or to track a single craft, it is nearly always desirable to have the radio frequency signal received only within a limited range. Thus, the radio frequency signal is of limited strength. For example, when functioning to avoid collisions between aircraft, the area of interest is generally a radius of between from about 1 to about 30 miles around one's own aircraft or around the airport. Nevertheless, in some cases, such as in the case of land or sea craft, the range of interest may be greater or lesser than that which is desirable for aircraft.

The radio frequency signal carrying the position fix, identification code information, and other messages, is of limited power so that only the radio frequency signal transmitted from craft within a predetermined range will be received. This reduces the number of radio frequency signals which conflict with one another, allowing a large number of aircraft to be simultaneously operating within a predetermined range. Other embodiments of the present invention can be fabricated which

allow a greater number of craft to be simultaneously operating within the predetermined range.

Preferably, the radio frequency signal from each of the aircraft is transmitted, and received, omnidirectionally. Since position information is encoded within the radio frequency signal, and not derived from the signal strength or direction, complicated antenna arrangements are not necessary with the present invention. As will be appreciated, in the case of aircraft the term "omnidirectionally" is intended to include three planes of space while in the case of land or water craft, the term "omnidirectionally" generally need only include horizontal two dimensional planes surrounding the craft.

As indicated at step 18 of receive loop 13 in FIG. 1B, the radio frequency signals for other craft are transmitted and are received at a craft or another location which is equipped with a monitoring receiver in accordance with the present invention. The receiving location may be a ground location which is tracking one or more craft. The radio frequency signals are decoded, as indicated at step 20, and the position fix and identification code information is displayed at step 21.

Once the position information has been extracted from the radio frequency signals, the position of all the other craft within the predetermined range can be displayed as represented at step 21 in FIG. 1B. As each craft regularly transmits its own position, it is possible to calculate all relative essential elements of heading, speed and distance of each craft which can then be displaced. In the case of craft on a collision course, the projected point of impact and time of impact can be calculated and displayed while the operators of the craft are alerted to the danger and given appropriate advisories. While it is not always necessary, it is preferred that an identification code for each craft be transmitted. The transmission of a craft identification code and its current operating communication radio frequency simplifies the organization of incoming data and enhances safety by providing a craft identification for two-way radio communication.

Included in step 21 of displaying the position of all craft transmitting a radio frequency signal within the predetermined range is preferably the step of displaying the position of one's own craft. Also, it is preferable to issue a collision alert if a collision is possible between two craft on their present headings. Also, since, one's own position fix has already been determined, it is possible to display the position of one's own craft against a map superimposed on the display. Data to create such maps can be provided by databases as are available in the art.

Preferably, each of the craft may be provided with a recording device which will provide a non-volatile record of all position and identification information which is received, and if desired the position information of one's own craft. Thus, if an aircraft were to become disabled and go down, the radio frequency transmissions would continue to transmit the position and identification of the craft to be observed by other craft. If the apparatus did not survive a crash, any properly equipped craft within range recording all received position and identification information and the last transmitted radio frequency signal from the downed aircraft would provide precise information of the location information of the crash site. With either alternative, the present invention can be used as an emergency location system that will supply more precise informa-

tion than that commonly available on the ELT (Emergency Location Transmitter) system in current use.

Referring next to FIG. 2, a primary aircraft 30 is represented at the center of a sphere 36. The sphere (which is not to scale) represents the predetermined range in which the radio frequency signal broadcast from the primary aircraft 30 can be received by other aircraft also carrying the system of the present invention. Thus, the operator of a secondary aircraft 32, being within the predetermined range, will receive a collision alert if the two craft are on a collision course. Alternatively, another secondary craft 34 will not be made aware of the position of the other two aircraft until it comes within the predetermined range represented by sphere 36. A ground based monitoring receiver 38 is also represented in FIG. 2. The ground based monitoring receiver can function as an air traffic control station as the craft come within range.

Referring next to FIG. 3, a block diagram of the presently preferred embodiment of the present invention is provided. The high level functional blocks include: a position determining block 100; an interface controller block 102; a radio frequency block 104; an antenna 106; and a control block 108. In most of the functional blocks, additional components are represented. The components represented in FIG. 3 are generally included on board a craft or can be located at a stationary location.

Represented within the position determining block 100 are a latitude/longitude position determining device 100A and an altitude determining device 100B. It will be appreciated that many aircraft already include appropriate latitude/longitude position determining devices and an altitude determining device and thus the components represented within the position determining block 100 may not be included in all embodiments of the present invention. It is one of the advantages of the present invention that the existing equipment already provided on many aircraft can be used in conjunction with embodiments of the present invention. In the case of a water or land craft, the inclusion of only a latitude/longitude device is generally all that is necessary.

The presently preferred latitude/longitude position determining device is a LORAN-C receiver available from ARNAV, a subsidiary of Flight Dynamics, Inc., of Portland, Oreg. and referred to as Model R-50. The described LORAN receiver provides many desirable features as well as the ability to communicate with other devices via the industry standard RS-232 communication protocol.

The presently preferred altitude determining device is an altimeter available from ACK Technologies, Inc. of San Jose, Calif., Model A-30. Information concerning interfacing the indicated altimeter to other components is available from the manufacturer.

The devices included in the position determining block 100 are a preferred example of a means for determining, on board a craft, the craft's position using an existing external navigation system. The means for determining the craft's position can include, depending upon the particular application of the invention, devices for determining the craft's position in one, two, or three dimensions (or planes) of space.

The present invention includes means for encoding the position of the craft, which can also preferably encode an identification code for the craft, so such information can be carried by a radio frequency signal. The presently preferred example of a means for encod-

ing the position and identification of the craft is the interface controller 102 represented in FIG. 3.

The identification code may be an aircraft registration number or any other number unique to the craft. While it is not essential to encode and transmit the craft identification code with the radio frequency signal, it is desirable to do so.

Still referring to FIG. 3, the interface controller 102 communicates with the devices of the position determining block 100 by way of communication ports included in the devices. The presently preferred example of the interface controller 102 is a system available from Enduratek Corporation of Salt Lake City, Utah and referred to as the Data-V-Com system. Complete information concerning the Data-V-Com system can be obtained from the publication "Mobile Data Terminal System Operations Manual" available from Enduratek Corporation and which is incorporated herein by reference.

The interface controller 102 system performs data encoding and preparation functions needed to transmit the digital information via a radio frequency signal and functions as the presently preferred example of a means for encoding the position of the craft into information which can be carried by a first radio frequency signal and a means for decoding the other craft's position from a radio frequency signal. Other devices performing similar or equivalent functions are intended to be included within the scope of the means for encoding included within the present invention. The designated interface controller 102 also functions as the presently preferred example of a means for decoding the position and identification information received from other craft. The radio frequency block 104 includes a modulator 104A, a radio transmitter 104B, an antenna duplexer 104C, a radio receiver 104D, and a demodulator 104E. The components of the radio frequency block are preferably included in a radio frequency transceiver capable of operating in the microwave band containing 1.6 Gigahertz. One presently preferred transceiver which operates in the VHF band is available from Icom, Model No. IC-A20. It will be appreciated that devices other than the designated transceiver, and frequencies other than 1.6 Gigahertz, for example any appropriate radio frequency may be used within the scope of the present invention.

Also represented in FIG. 3 is an antenna 106, which may comprise one or more individual antennas or antenna elements, and which is preferably one that will provide an omnidirectional radiation pattern. It is desirable that the radiation pattern and the reception pattern both be omnidirectional. The antenna duplexer 104C serves to isolate the radio transmitter 104B from the radio receiver 104D while allowing both to use the same antenna.

Still referring to FIG. 3, a modulator 104A functions to modulate the carrier radio frequency wave with the craft position and identification information encoded by the interface controller 102. A demodulator 104E similarly functions to extract the position and identification information from any radio frequency signal which is received and to pass the same onto the interface controller 102. The modulator 104A and demodulator 104E may be embodied in a device commonly referred to as a modem, which is available in the art.

The components represented within the radio block 104 and the antenna 106 are the presently preferred example of a means for transmitting the first radio fre-

quency signal and a means for receiving a radio frequency signal. The interface controller 102 communicates with, and operates, the components of the radio frequency block 104 by way of connectors and cables customarily included on radio frequency transceivers as will be understood by those skilled in the pertinent art.

The interface controller 102 represented in FIG. 3 also communicates with a computer 108A which is included in the components of the control block 108. The computer 108 is preferably one which is equivalent to an IBM model PC AT. One of the many commercially available portable models (generally referred to as "Lap Tops") of the specified equivalent computer can also be used. The computer 108A and the interface controller 102 communicate with each other by way of the customary connectors and protocols (such as the RS-232 protocol) which are well known to those skilled in the art.

The programming code which is presently preferred for use with the interface controller 102 is attached hereto as Appendix A. The programming code which is presently preferred for use with the computer 108A is attached hereto as Appendix B.

Included in the control block 108 is a display 108B. The display 108B may be integral with the computer 108A or may be separate therefrom. The display is preferably used to provide a graphical representation of the position of the craft which are within the predetermined range of the monitoring craft or receiver. Thus, the operator of the craft can observe on the display the position of both his own craft (desirably in the center of the display) and the position of all other craft surrounding his craft within the predetermined range.

One preferred embodiment of the display would incorporate the use of a head-up display to display not only collision avoidance information, but to also aid in proximity reporting, navigation under IFR conditions, coordinated descent into controlled and uncontrolled airports and landing strips, and other relevant information, and to project these pertinent images in a forward view mode into the wind screen and line of the actual and preferred flight path.

It will be appreciated that some embodiments of the present invention may provide additional benefits if the display is one which not only provides high resolution in two dimensions, but one which is also suitable for displaying three dimensions, i.e., a display wherein depth can be represented by either using a three dimensional graphics mode or by decreasing intensity as a position recedes from the viewer and greater intensity as the position moves toward the viewer. Altitude information as well as indicators showing whether or not aircraft are climbing or descending can also be depicted on the display.

In the case of the present invention as applied to the aviation industry, it is desirable that the position of aircraft other than those in the same horizontal plane be displayed. Namely, the display 108B should provide position information of a plurality aircraft vertically displaced from the monitoring craft (i.e., differing altitude) as well as in the same horizontal plane (latitude and longitude). Depending upon the particular embodiment of the invention, the display 108B, or all of the components represented in radio block 108, function as a means for displaying a craft's position as defined by the present invention. Other devices and arrangements performing similar or identical functions are intended to be considered equivalent to the described structures.

Also represented in the control block 108 are input devices 108C. Input devices 108C can include a keyboard (which may be integral with the computer 108A) or other general purpose or dedicated input devices. An audio/visual alarm 108D is also represented in control block 108. The audio/visual alarm can be integral with the computer 108A or can be one of several dedicated alarm devices intended to apprise the operator of a craft of a potential collision or other situation requiring attention.

A recorder 108E is also provided in the control block 108 to make a non-volatile recording of selected data received by the computer 108A. Preferably, the recorder 108E is used to periodically log the position of all craft within the predetermined range. Thus, if a monitored craft becomes disabled and stops transmitting its radio frequency signal carrying position information, the last recorded position fix logged in the recorder 108E of any craft or observation station within the predetermined range can aid in the location of a disabled craft.

Furthermore, if a craft continues to transmit its radio frequency signal with its own position fix after becoming disabled, it will function as an emergency location device and will assist search and rescue crews in the rapid location of the disabled craft. In some embodiments of the invention, the radio frequency signal can carry, in addition to position and identification information, a distress signal as well as other messages to indicate a need for emergency assistance and even describe what type of emergency assistance is required at the disabled craft.

If desired, the components of the control block 108, the interface block 102, and the reception components of the radio frequency block 104 can be used as a monitoring receiver. For example, a monitoring receiver can be used as a ground control station to monitor the airspace surrounding an airport. In the case of small airstrips or airports in developing areas, the present invention can be adapted to provide both airborne craft and ground control personnel the information necessary to coordinate air traffic. This application of the embodiment of the invention would allow for the system's use as a standard ATC system exclusive of expensive radar installations, and at extremely low cost.

It will be appreciated that while the described embodiment is the presently best known mode for carrying out the invention, those skilled in the pertinent art will understand that other components may be used to carry out the invention. For example, the functional blocks illustrated in FIG. 3 may be combined into a few, or just one, devices housed in a single enclosure. Moreover, as the art progresses, the inventive concepts of the present invention may be embodied in devices not yet available but carrying out functions equivalent to those described in connection with the preferred embodiment. All such alternative embodiments of the invention are intended to fall within the scope of the present invention.

Provided in FIGS. 4-1 through 4-11 is a more detailed flow chart describing the steps carried out by the above-described embodiment. In these figures, the boxed letter designations indicate the interconnections between the portions of the flow chart which are divided between figures.

The flow chart provides a description of the high-level functions carried out principally by the interface controller 102 represented in FIG. 3. In the presently preferred embodiment of the present invention, the

interface controller 102 is implemented by the above described Data-V-Com system. Also, as indicated earlier, the presently preferred programming code for the Data-V-Com system is included in Appendix A attached hereto.

It will be appreciated that the flow chart represented in FIGS. 4-1 through 4-11 represents just one preferred example of the programming steps that can be used to carry out the present invention and those skilled in the art will be able to devise other embodiments of the present invention using the teaching contained herein. Also, as will be appreciated after an examination of FIGS. 4-1 through 4-11, the Lap Top (a portable computer which functions as the computer 108A represented in FIG. 3) functions principally to display the data presented to it by other components and to provide input and output functions with a human user. The text contained in the dashed boxes are included to improve the clarity of the flow chart and do not represent steps in the method.

Also, as indicated earlier, the presently preferred programming code for the Lap Top is included in Appendix B attached hereto. Those skilled in the art will be able to provide embodiments of the present invention which perform functions similar or equivalent to those performed by the described embodiment, using the teachings contained herein.

The flow chart contained in FIGS. 4-1 through 4-11 is divided into several principal routines as indicated below.

Flow Chart Routine	FIG. NOS.
Initialization	FIGS. 4-1 through 4-4
Radio	FIGS. 4-5 through 4-6
Loran	FIG. 4-7
Altimeter	FIG. 4-8
Lap Top	FIGS. 4-9 through 4-11

Further information concerning each of the routines indicated above is provided in the Glossaries set forth below.

Ref. No.	Label	Description
<u>A. Initialization Routine Glossary</u>		
200	START	This is the beginning of the program. When the micro-processor is reset via hardware or software it is vectored to this point. At START, the Stack Pointers, Interrupt disabling and various other Housekeeping chores are performed.
202	SELFTEST	This is the beginning of the Selftest Module. At this point the Diagnostic Terminal will display that Selftest has started. Also at this point Selftest Interrupts will be set up and enabled.
206	RAM TEST PASS	The RAM (Random Access Memory) test is run. The RAM test writes data found in a ROM table to each RAM address. If the same data that was written is read back, the program continues through the table until the data 00H is written. This signifies the end of the test and leaves all RAM cleared and ready for use. If the test fails at any point the data read, data written and the address are saved and reported back to the Diagnostic Terminal. The RAM should be replaced at the address reported if failure occurs.
208	SET RAM	The Selftest Status is sent to the Lap

-continued

Ref. No.	Label	Description
	TEST FAIL BIT	Top and to other Aircraft via the Radio. If this bit is set it means that the RAM Test Failed and all data is unreliable.
210	ROM TEST PASS	All Used ROM (Read Only Memory) is added and the sum must equal zero. The Check Sum Adjust Byte at 3FFFH ensures that the sum (without carry) is zero. If the sum is not zero the test fails and ROM should be replaced. The test failure status and the erroneous check sum are reported to the Diagnostic Terminal.
212	SET ROM TEST FAIL BIT	The Selftest Status is sent to the Lap Top and to other Aircraft via the Radio. If this bit is set it means that the ROM Test Failed and all data is unreliable.
214	LAP TOP INTER. PASS	The USART (Universal Synchronous Asynchronous Receive Transmit) 8051 is Software reset and the MIF (Mode Instruction Format) set up. Status is then checked and verified.
216	SET LAP TOP INTERFACE FAIL BIT	The Selftest Status is sent to the Lap Top and to other Aircraft via the Radio. If this bit is set it means that the Lap Top Communication could be unreliable.
218	I/O TEST PASS	The PIA (Peripheral Interface Adapter) 8155 is initialized for the proper mode, which ports are Inputs and which are outputs and the Baud rate Clock set up. Outputs are initialized. Status is then checked and verified.
220	SET INPUT/OUTPUT FAIL BIT	The Selftest Status is sent to the Lap Top and to other Aircraft via the Radio. If this bit is set it means that the PIA is unreliable. The PIA interfaces mainly with the Altimeter.
222	RADIO INTER PASS	The USART (Universal Synchronous Asynchronous Receive Transmit) 8051 is Software reset and the MIF (Mode Instruction Format) set up. Status is then checked and verified.
224	SET RADIO INTERFACE FAIL BIT	The Selftest Status is sent to the Lap Top and to other Aircraft via the Radio. If this bit is set it means that the Radio Communication could be unreliable.
226	ADD AIRCRAFT ID	Each System has its own Aircraft Identification Via the Diagnostic Terminal the Aircraft Identification is put in NonVolatile Memory or EEPROM. (Electrically Erasable Read Only Memory). This Identification number will not be lost if the system is reset or powered down. When Selftest is run the system can sense that the Terminal is attached and allow the user the option of changing the Aircraft Identification This block asks the question, Do You Want to Enter Aircraft Identification?
228	GET AIRCRAFT ID	This block Prompts the Terminal, for the Identification, Error Checks it and buffers it.
230	ENTER AIRCRAFT ID	This block enters the buffered aircraft Identification into EEPROM.
232	PUT IN RANDOM XMIT TIMER	Because the preferred embodiment of the present invention is polite (will not transmit when another signal is already present on the frequency). To eliminate various systems transmitting simultaneously, each system will have different wait times before transmitting after a frequency has cleared. This random wait time is determined by the system Aircraft Identification in the preferred embodiment.

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Ref. No.	Label	Description
5	234 TOGGLE PRINTER STAT.	Changes the printer status.
10	236 CHG PRINTER STAT	To enable someone to monitor the entire system there is a port for a printer or some other type of monitor. At this time the Diagnostic Terminal can be used for changing the Status of the Printer or Monitor. The Status is kept in NonVolatile ROM or EEPROM so that if the System is powered down or reset that status will not change.
15	240 INITIA-LIZE POINTERS TIMERS ETC.	Pointers such as the Input and Output Pointers for the Rotating buffers used for the Lap Top Interface are set up at this time.
20	242 SET ALTIMETER DATA TO 9's	At this point in the Program it is not known if the Altimeter is functional or not, especially since the Altimeter does not function for several minutes after it is powered up. So that any data present is not mistaken for good Altimeter Data all 9's are inserted in the Altimeter use area.
25	244 SET LORAN DATA TO 9's	At this point in the Program it is not known if the Loran Receiver is functional or not, especially since the Loran Receiver must be initialized manually to send the proper information at the proper baud rate, transmit rate, and so forth. To ensure that any random data already present is not mistaken for good data, all 9's are inserted in the Loran Longitude and Latitude buffers.
30	246 SET # MESSAGE FLAG	The # message Flag is set so that when the system is fully initialized and functional, the # message will be sent to the Lap Top Computer.
35	250 EXECLP	This is the exit point for initialization that has been performed heretofore and the loop point for the rest of the program. When the rest of the program has completed it will come back to this point and begin again at this point.
40	252 CALL RADIO	The Radio Module (See step 276, FIG. 4-5) is interfaced at this point.
45	256 CALL LORAN	The Loran Module (See step 312, FIG. 4-7) is interfaced at this point.
50	260 CALL LAP TOP	The Lap Top Computer Module (See step 348, FIG. 4-9) is interfaced at this point.
55	264 RESET WATCH DOG	See step 266.
60	266 TRANSMITTING	If the system is transmitting the watch dog timer is not reset. This helps to ensure that the system does not transmit for too long a period. If the watch dog is not reset every 1.1 seconds the system is reset, the Program Counter goes to 0000H or the Start.
65	268 CALL ALTIMETER	The Altimeter Module is interfaced at this point.
		<u>B. Radio Routine Glossary.</u>
	272 RANDOM TIMER EXP	After the frequency has been busy this timer must expire before transmitting. (See step 232 (FIG. 4-2) and 278 (FIG. 4-5).
	274 'T' RELOAD VALUE NON 0?	
	276 RADIO	This is the entrance to the RADIO module. This module handles all the communication to/from the Radio. All communication to other aircraft is via

-continued

Ref. No.	Label	Description
278	RESET XMIT TIMER	the Radio. The value used to reset the XMIT timer is the value obtained from the 'T' message from the Lap Top Computer. (See steps 274 (FIG. 4-5), 350 (FIG. 4-9) & 376 (FIG. 4-10).
280	CHANNEL CLEAR	Because the system is polite (See step 232 (FIG. 4-2)). It will not transmit unless the channel is clear. The Carrier Active Sense (CAS) is checked at this point to see if the frequency is clear or not.
282	XMIT TIMER EXP	The location message for this aircraft must be transmitted to the other aircraft at a rate determined via a 'T' message from the Lap Top Computer. If the 'T' message time is 0 then (which is what is upon initialization) then the message is not transmitted at all.
286	GET THIS AIRCRAFT ID	The Aircraft Identification is read from EEPROM and stored in the Radio Transmit buffer. (See steps 230, 228 (FIG. 4-2), & 226).
290	RESET RANDOM TIMER	The Random timer must be reset with the value determined from the Aircraft Identification (See steps 232 (FIG. 4-2), 280 & 372 (FIG. 4-5).
292	GET LORAN LAT. LON. DATA	The most recent data from the Loran Receiver is placed in the Transmit buffer. (See steps 292 (FIG. 4-6), 314 & 320 (FIG. 4-7)).
296	MESSAGE RECEIVED?	By examining the input buffer it can be determined if a message has been received.
298	CHECK SUM CORRECT	Each byte of the message is added w/carry and it is determined if the message is error free or not.
300	PUT MESSAGE IN LAP TOP BUFFER	The message received above is rebuffered into the large rotating Lap Top Computer Buffer for transmittal to the Lap Top Computer.
302	GET LORAN STATUS	This byte indicates the Loran Status. (See step 330 (FIG. 4-7)). The Loran Receiver transmits in addition to the Latitude and Longitude the Status of the Receiver.
304	SET LAPTOP MESSAGE READY FLAG	This flag tells the Lap Top Module that a message is ready to go to the Lap Top Computer.
306	GET DATA V-COM STATUS	The Data-V-Com Status includes not only the Selftest status, (See steps 208, 212, 216, 220 & 224 (FIGS. 4-1 & 4-2)) but also status as to how current the Loran Information is.
308	GET MESSAGE CHECKSUM	After concluding with step 306 (FIG. 4-6) the message is formatted and the Check sum is calculated by adding up each byte (w/carry).
310	OUTPUT MESSAGE	The message formatted above is output (AFSK) via the USART. An attack delay (a period of no data to enable the transmitter to stabilize) as well as a time delay at the end is necessary.
<u>C. Loran Routine Glossary.</u>		
312	LORAN	This is the Entrance Point for the Loran Module. This module handles the interface with the Loran receiver. Besides receiving the Latitude and Longitude it also saves the Loran Status and keeps track of if an update occurs between Message transmissions to other aircraft and to the Lap Top.
314	SAVE LATITUDE	The Latitude less the spaces is saved. (See step 318 (FIG. 4-7)). A flag is set so that it can be determined that

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Ref. No.	Label	Description
5		data was received for Lat. and Lon. so that when processed will be data that is compatible.
316	LATITUDE	Checks for LAT (See step 318 (FIG. 4-7)) in message coming in from Loran Receiver.
10	318 MESSAGE COMING IN?	The entire message from the Loran Receiver is not saved. If a message is coming in the program looks for LAT for Latitude and LON for longitude and SD for Loran Status. If these labels are found the data less the spaces is saved.
15	320 SAVE STATUS	The Loran Status is saved for transmission to other Aircraft, as well as the Lap Top Computer. This makes it possible to know the accuracy of the Loran as well as other pertinent information. (See steps 388 (FIG. 4-11) & 302 (FIG. 4-6)).
20	322 LONGITUDE ?	Checks for LON (See step 318 (FIG. 4-7)) in message coming in from Loran Receiver.
	324 ALL DATA RECEIVED?	See step 318.
25	326 PUT LAT. IN LAPTOP BUFFER	The message received above is rebuffered into the large rotating Lap Top Computer Buffer for transmittal to the Lap Top Computer.
	330 SAVE STATUS	See step 320.
30	332 LORAN STATUS?	A flag is set for each of the 11 different types of status if that particular status is read from the Loran Receiver.
	334 PUT LON. IN LAPTOP BUFFER	See step 326.
35	336 PUT LORAN STATUS IN LAPTOP BUFFER	See step 332.
<u>D. Altimeter Routine Glossary.</u>		
40	338 ALTIMETER	This is the Entrance point to the Altimeter handler. It reads the 11 Discrete inputs and decodes the data to an ASCII altitude. If the altitude is bad a 998 is output.
	340 GET DISCRETE INPUTS (11)	Reads the 11 Discrete inputs from the Altimeter. If some are not used the program is modified so that they are assumed to be most significant and low.
45	342 GET ALT. FROM TABLE	The data is deemed good and saved for messages to the Lap Top and to other Aircraft.
50	344 INPUT DATA IN TABLE	A look up table is used to decode the 11 inputs to a three digit altitude reflecting 100's of feet. If the table reflects 998 then the data is deemed to be bad and is rejected.
	346 GET '998'	The data in the table was bad, so instead of unreadable data a '998' is used to reflect the error.
<u>E. Laptop Routine Glossary.</u>		
	348 LAPTOP	This is the input to the module. This module handles all communication to/from the Lap Top Computer.
60	350 PUT '#' IN MESSAGE CLEAR '#' FLAG	Put the '#' in the output buffer for this particular message to the Laptop.
	352 CLEAR '#' FLAG	See step 354 (FIG. 4-9).
	354 '#' FLAG SET?	This flag indicates that a # Message must be sent to the Lap Top Computer. This flag is set only during Initialization. (See step 246 (FIG. 4-3)).
65	358 GET THIS AIRCRAFT	The Aircraft Identification in Nonvolatile memory (EEPROM), is put

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Ref. No.	Label	Description
	ID	in the transmit buffer to go to the Lap Top Computer. (See step 228 (FIG. 4-2)).
360	GET MESSAGE FROM OTHER AIRCRAFT	The message that was received from another aircraft via. the RADIO module is rebuffered and sent to the Lap Top Computer.
362	'O' FLAG SET?	The 'O' determines if messages from other Aircraft are accepted or not. If the Flag is set, which must occur when a 'O' message is received from the Lap Top Computer, (See step 378 (FIG. 4-10)) then other aircraft messages are accepted.
364	GET SELFTEST STATUS	The Selftest Status (See steps 208, 212, 216, 220 (FIG. 4-1) & 224 (FIG. 4-2)) is put in the buffer to be transmitted to the Lap Top Computer.
366	RESET 'L' TIMER	The time value received from the Lap Top Computer, (See step 390 (FIG. 4-11)) is used to reset this timer.
368	'L' TIMER EXP	The 'L' timer determines how often the Location message for this aircraft is sent to the Lap Top Computer. If the value is set at 0, which occurs on initialization, it is not sent at all. The value that the 'L' timer is set to is determined by the 'L' message received from the Lap Top Computer. (See step 390 (FIG. 4-11)).
372	GET THIS AIRCRAFT ID	The Aircraft Identification in Nonvolatile memory (EEPROM), is put in the transmit buffer to go to the Lap Top Computer. (See step 228 (FIG. 4-2)).
376	GET THIS LONGITUDE LATITUDE	This information is obtained from the Loran module. (See steps 314 & 320 (FIG. 4-7)) and is put in the transmit buffer to go to the Lap Top Computer.
378	"O" MESSAGE?	Is the incoming message from the Lap Top Computer an 'O' message? If so the 'O' flag is set. (See step 362 (FIG. 4-10)).
380	SET "O" FLAG	See step 378 (FIG. 4-10).
382	GET THIS ALTITUDE	This information is obtained from the Altitude module. (See step 342 (FIG. 4-8)) and is put in the transmit buffer to go to the Lap Top Computer.
384	"T" MESSAGE?	Is the incoming message from the Lap Top Computer a 'T' message? (See steps 274 & 282 (FIG. 4-5)). If so the 'T' Reload value is saved. This value determines how often the Radio Transmits the Location of the Aircraft to other Aircraft.
386	GET "T" RELOAD VALUE	See step 384 (FIG. 4-11). This value is used at step 372 (FIG. 4-5).
388	GET THIS LORAN STATUS	This information is obtained from the Loran module. (See step 330 (FIG. 4-7)) and is put in the transmit buffer to go to the Lap Top Computer.
390	"L" MESSAGE?	Is the incoming message from the Lap Top Computer and 'L' message? (See steps 368 (FIG. 4-10) & 392 (FIG. 4-11)). If so the 'L' Reload value is saved for use at step 366 (FIG. 4-10).
392	GET "L" RELOAD VALUE	See step 390 (FIG. 4-11). This value is used at step 366 (FIG. 4-10).
394	GET CHECKSUM	After concluding with step 388 (FIG. 4-11) the message is formatted and the Check sum is calculated by adding up each byte (w/carry).
396	"R" MESSAGE?	Is the incoming message from the Lap Top Computer an 'R' message? If so the

-continued

Ref. No.	Label	Description
5		program is vectored through software to the start address. (See step 276 (FIG. 4-5)).
398	OUTPUT MESSAGE	The message formatted above is output via the USART to the Lap Top Computer.

3. Summary

In view of the foregoing, it will be appreciated that the present invention provides a system and method for announcing the position of a craft to a remote receiver such that the position of the craft can be continuously monitored and collisions with other craft avoided. The present invention also provides a system and method for collision avoidance which may be used in addition to, and without interference with, preexisting collision avoidance systems.

It will be further appreciated that the present invention also provides an emergency location system for distressed or disabled craft and a collision avoidance system which can be economically implemented in the general aviation industry. Still further, the present invention provides a system and method which allows the operator of a first craft to monitor the position of a plurality of other craft within a predetermined range of the first craft and to be alerted of a potential collision as early as possible.

The present invention also provides a system and method of monitoring craft proximity spacing, a critical function, while operating within the flight sector system in high density areas and while transiting or making approaches in VFR and IFR controlled airspace. Further, the system and method provide a similar function in the marine environment, particularly under adverse weather conditions and high density mooring or docking maneuvers.

Even further, after consideration of the foregoing, it will be understood that the present invention provides a system and method for announcing the position of any craft to other craft which are within a predetermined range which is reliable and which does not distract an operator of a craft from other duties unless operator attention is necessary. Moreover, the operator of a first craft is allowed to continually monitor the positions of a plurality of other craft within a predetermined range. Even further, the present invention provides a system and method of tracking the position of one or more craft within a predetermined range of a location anywhere on the earth such as at a remote airstrip or allows the position of each of a plurality of craft belonging to a fleet to be tracked. Still further, the present invention provides a collision avoidance system which does not become saturated in areas of dense traffic and which readily makes a permanent record of the movement of any craft within a predetermined range of a monitoring receiver.

The present invention may be embodied in other specific forms without departing from its spirit or essential characteristics. The described embodiment is to be considered in all respects only as illustrative and not restrictive. The scope of the invention is, therefore, indicated by the appended claims rather than by the foregoing description. All changes which come within the meaning and range of equivalency of the claims are to be embraced within their scope.

United States Patent Application
of
Edward J. Fraughton
and
Philip H. Berger
for
A UNIVERSAL DYNAMIC NAVIGATION,
SURVEILLANCE, EMERGENCY LOCATION, AND
COLLISION AVOIDANCE SYSTEM AND METHOD

APPENDIX A
PROGRAMMING CODE

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ORGANIZATION OF FIRMWARE FOR ACAS DATA-V-COM 16 Jan 90

Some of the descriptions in this document are operational and others are only organizational. The operational and other more important descriptions are highlighted in bold.

The following is a module by module description for the ACAS firmware for the Data-V-Com. Also included are all global buffers and flags for each module and how each will interface. Rotating buffers are used only where so stated (for output functions only). Where rotating buffers are used if, the input and output pointers are made equal while outputting the interrupts are disabled. The Global Flags and Buffers, RDTLTB/F, LRTLTB/F, LTDDVB/F and LRTDRB/F will function in the following manner. The module sending the message will set the flag, all 8 bits to 1's, and fill the buffer. Two modules will receive each message, one of which is the Printer module. The printer module will clear the LSB and the other module will clear the MSB when it receives the message. The module transmitting the message will wait till only the MSB is cleared before it will send another message. The LSB is only to tell the Printer module that it has received the message. The buffers and flags are described in the Section PRNHAN.SRC.

INITAL.SRC This module is the Initialization module. This module will be vectored to on power up. It will initialize stack pointers, set interrupt vectors etc. and then call the selftest module SLFTST.SRC. When selftest has completed, this module will enable interrupts, set interrupt priorities etc. The names for the initialization routines for each of the following modules that needs one will be the same as the first three letters with a INL at the end, such as ITIINL. Selftest will call some of the initialization routines, this module will call the others. This module will also house the interrupt vectors for the modules requiring interrupts. The final act of this module will be to jump to the Executive loop EXECLP.SRC.

SLFTST.SRC Selftest will be called by the Initialization Routine. It is the only other routine other than the Executive Loop that will be allowed to reset the Watchdog timer. Selftest is the only other module other than the Executive and Initialization that will be allowed to call other modules. The following selftests will be performed:

1. EPROM Check Sum Test. All EPROM must add up to a predetermined value or this test will fail.
2. RAM Checkerboard Test. Patterns of all 1's, all 0's and alternating 1's and 0's will be written to each RAM address. After all addresses have been written with the pattern, each is read. If the correct data is not read this test will fail.
3. Peripheral Interface Adapter Test. Status Check only, there are at present no provisions for a wraparound self test.
4. USART Test. Status Check only on all used USARTS, there are at present no provisions for a wraparound self test.

This module will sense if a terminal is attached. If a terminal is attached it will give the operator the option of bypassing or rerunning the test that failed. Each time a test fails, failure data, such as the address and bit pattern for the failure will be displayed on the terminal. When a terminal is attached it will also handle operator input for the following functions:

1. See if Operator wants to exit Selftest and continue with no changes.
2. See if the Operator wants to change the Primary Aircraft Id.
3. See if Operator wants to change the transmission delay time. (TRNTIM see the Radio module for details) This should be slightly different for each aircraft.

4. If the Diagnostic function is enabled at assembly time the operator, can also turn on this function. (See below) If this function is turned on it will be turned off the next time the system is reset either by the Lap Top Computer or a power reset.

5. If the Printer function is enabled at assembly time the operator can also toggle this function off or on. This toggle will be in nonvolatile EEPROM so that resetting does not change it.

EXECLP.SRC This is the Executive module and will call all the other modules except as noted above. After Selftest and Initialization are complete, this is the only module that will be allowed to call other modules, except of course for INITAL.SRC.

A second function of the Executive module is to declare global all Buffers, Flags etc. that are used by more than one module. Also any parameters that may need to be changed, such as memory limits etc. will be in the Executive. This does not include timers used only in one module. They will be declared global in the timer module and used in one other module.

Although the code for the Executive is minimal it will be most likely be the largest module.

Because this module is only a loop for calling the other modules, it will reset the watch dog timer, but only after calling all the other modules. If the radio module is transmitting (MESINP flag), or if the computer module has received a reset message (NWDRST flag), the watchdog will not be reset.

ITIMER.SRC The timer routine after initialization will be interrupt driven. This module will be in two parts as shown:

- A. ITIINT Label called by interrupt from the Initialization module.
- B. ITIINL Initialization part of module called from the Initialization module.

The user will set the timer location with a value and then the timer module will decrement it until it is zero. It will be the function of the user module to monitor the timer to see when it expires. There will be three groups of timers, those that are decremented every 1/4 sec., 1 sec. and every minute.

The rest of the modules will only be called from the Executive Loop.

PRNHAN.SRC This module can be enabled or disabled at assembly time. There will be a flag that can be set or cleared when the system is assembled and linked that will enable or disable this module. If disabled the USART will be turned off and the module will be disabled. Also to activate this module it will be necessary to set a non-volatile EEPROM location to an ASCII P. This can be performed during Selftest but only when a Diagnostic Terminal is attached. See Selftest for toggling this function. The parts of this module are as follows:

- A. PRNHAN Called by the Executive from the main loop. This module will monitor communication between the Data-V-Com and the radio, between the Data-V-Com and the Lap Top Computer as well as between the Loran Receiver and the Data-V-Com. This will check flags to see if a message needs to go out to the printer/terminal that it handles. This function will be rebuffered to the rotating buffer and add the characters for where the message is from/to as shown below.
- B. PRNITL This is the initialization for this module called by Selftest.

C. PRNINT This is the interrupt driver for this module for outputting data to the printer/terminal. The transmit function will have a 500 character rotating buffer. If the Buffer should become full it will not inhibit the transmission of information, but will print out Printer Buffer Full, clear just this buffer and then start over.

Each message will be on a single line with. At the end of the line there will be a space then the following to designate where the message is going or where it came from. Also included here are the message global buffers and flags. When this module reads the following Buffer it will clear the LSB (Least Sig. Byte) of the following flags:

	<u>Buffer</u>	<u>Flag</u>	<u>Printer Designation</u>
From Radio to Lap Top Computer	RDTLTB	RDTLTF	RDTLT
From Loran Receiver to Lap Top Computer, (includes # msg)	LRTLTB	LRTLTF	LRTLTL
From Lap Top Computer to Data-V-Com	LTTDVB	LTTDVF	LTTDV
From Loran Receiver to Radio	LRTRDB	LRTRDF	LRTRD

This module will be the last one that is written except for the Diagnostic module (DIAGNT.SRC) and any or all of the above may be left out if time is short.

DIAGNT.SRC This module is also for trouble shooting as is PRNHAN.SRC above. This module will not need initialization and will have no interrupt vectors. The handle used by the Executive to call this module will be DIAGNT. The terminal used will be the Selftest Terminal when attached. If not attached this function will be disabled. While the system is up and running this module will make it possible to do the following:

- A. Read/Write Memory.
- B. Read Code.

A volatile memory location in RAM must have an ASCII D for this module to function. It can also be inhibited during assembly and linking. This module when activated will seriously impede the processor since the software programmed serial interface must disable all interrupts when functioning. No interrupt handlers are needed for this module. This module will be the last one written and will be left out if time is short.

LAPHAN.SRC This module is the Lap Top Computer Handler (Computer Module). This module is divided up as shown:

A. LAPHAN This is the main part of the module that will be called from the Executive Loop. On power up it will send the # message which includes the aircraft id in EEPROM and the Selftest Information (00 All passed) to the Lap Top Computer. The interface to the rest of the modules will be with the Global Buffers and Flags defined above for the printer module.

B. LAPRIN This is the receive interrupt handler. After the first character is received, the rest of the message, which is terminated with a Carriage Return, will have 1 second to be received, before it is aborted and the pointer reset to the beginning of the buffer. It is not anticipated that the Lap Top Computer will send messages closer than 1 second apart, because it must receive a response, before it sends the next message. If a Reset message is received it will set the NWRST flag, which will cause the system to be reset. (See the Executive section) If a 'T' message is received the next two digits will be converted to HEX and stored in the buffer

VTMCNT. This will be the number of times that the Loran Receiver will send information before the info will be sent to the Radio and the Lap Top Computer. The only other module that will need to read the messages coming in, is the printer module.

C. LAPTIN This is the transmit interrupt handler. Messages going to the Lap Top Computer will be put in a 500 character rotating buffer. This means that approximately 20 messages could be buffered up from the Radio and Loran Modules. Each time called this module will monitor the appropriate flags. If a flag is set it will transfer the appropriate message to the rotating buffer.

D. LAPINL This is the initialization routine for the module. It will set the RSTMSG flag that will cause LAPHAN to send the # or reset message when called.

RADHAN.SRC The radio module will transmit and receive messages over the radio. This module will be four parts as shown:

A. RADHAN This module will be polite and wait for the frequency to clear before transmitting. If it must wait longer than 10 sec., the message will be blurted out even if the channel is busy. If the channel clears before the 10 sec. has expired, it will not transmit immediately, because there may be other systems waiting also. Instead it will wait for a period of time determined by reading Nonvolatile EEPROM (TRNTIM). It is suggested that the first unit have a 01 for this value the next a 02 and so on till a max. is reached. Each tick will be 1/4 second.

B. RADINL Initialization called by Selftest.

C. RADRIN Receive interrupt handler from the Radio. Messages received will go only to the Printer and Computer modules as explained above.

D. RADTIN Transmit interrupt handler to the Radio. These messages will come only from the Loran handler module.

LORHAN.SRC This module will handle communications from the Loran Receiver. The only communication between the Loran Receiver and the Data-V-Com will be the data burst that the Loran Receiver will send once every second. This module will be divided as shown:

A. LORHAN This module will be called by the EXECLP every loop. When the LORRIN routine (B below) receives data from the Loran Receiver, if the counter TIMCNT has reached the value received from the Lap Top VTMCNT, the message will be sent to the Lap Top Computer and to the Radio handler RADHAN.

B. LORRIN Receive interrupt handler. This interrupt driven routine will ignore all data received from the Loran Receiver except the Longitude data after LON and the Latitude data after LAT. When it has successfully received this data it will set the local flag LORRCV. This flag will be cleared by the LORHAN module above.

C. LORINL Loran Initialization routine called by the Initialization module.

ALTHAN.SRC This module will read the 9 discrete inputs from the altimeter and decode them and put the ASCII data in the 3 byte buffers RADALT (3) and LAPALT (3) the MSB first. LORALT and LAPALT will be inbeded in the messages going out to the respective ports to the Radio and the Lap Top Computer.

THE FOLLOWING IS A LIST OF GLOBAL FLAGS, BUFFERS ETC.

DIAGNT = Diagnostic Handler Module.
 DIAINL = Diagnostic Handler Initialization.
 EXECLP = Executive Loop.
 INITAL = Initialization Module.
 ITIMER = Interrupt for timer module.
 ITIINL = Initialization for timer module.
 LAPALT = 3 byte buffer for altitude written by ALTHAN
 LAPDAT = USART Data for Computer handler.
 LAPHAN = Lap Top Computer Handler Module.
 LAPINL = Initialization Routine for Lap Top Computer.
 LAPRIN = Receive Interrupt Handler for Lap Top Computer.
 LAPSTS = USART Status for Computer handler.
 LAPTIN = Transmit Interrupt Handler for Lap Top Computer.
 LORALT = 3 byte buffer for altitude written by ALTHAN
 LORDAT = USART Data for Loran Handler
 LORHAN = Loran Receiver Handler Module.
 LORINL = Loran Receiver Handler Initialization.
 LORRIN = Loran Receiver Receive Interrupt handler.
 LORSTS = USART Status for Loran Handler.
 LRTRDB = From Loran Receiver to Radio Buffer, length MSGLEN
 LRTRDF = " Flag, MSB & LSB
 LRTLTB = From Loran Receiver to Lap Top Buffer, length MSGLEN
 LRTLTE = End of LRTLTB buffer.
 LRTLTF = " Flag, MSB & LSB
 LTTDVB = From Lap Top Computer to DVC Buffer, length MSGLEN
 LTTDVE = End of LRTLTE buffer.
 LTTDVF = " Flag, MSB & LSB
 MSGLEN = Message Length without checksum and start of header
 MSGTLN = Total Message Length with checksum and start of header
 PRNDAT = USART Data for Printer Handler.
 PRNHAN = Print Handler Module.
 PRNSTS = Print Handler Status.
 PRNTIN = Print Handler Transmit Interrupt.
 PRNITL = Print Handler Initialization.
 MESINP = Radio Message Inprocess.
 NWDRST = Reset Flag when set, watchdog not hit.
 RADDAT = USART Data for Radio Handler.
 RADHAN = Radio Handler Module.
 RADINL = Radio Handler Initialization.
 RADRIN = Radio Handler Receive Interrupt handler.
 RADSTS = USART Status for Radio Handler.
 RADTIN = Radio Handler Transmit Interrupt handler.
 RCVINP = Receive in process for MESINP
 RDTLTB = From Radio to Lap Top Computer Buffer, length MSGLEN
 RDTLTE = End of RDTLTB buffer.
 RDTLTF = " Flag, MSB & LSB
 SLFTST = Selftest Module.
 TMAXTX = Radio Transmit Maximum time before time out
 TRNINP = Transmit in process for MESINP
 TRNTIM = Transmission Delay, when receiveing Timer 1/4 sec
 VMAXTX = Value for TMAXTX.
 VRNTIM = Value for TRNTIM Non Volatile EEPROM
 VTMCNT = Timer for Loran Status to the RADIO and the Lap Top
 Computer. When 0 it will not ever transmit.

```

CALL      COMOUT                ;OUT PUT MESSAGE
MOV       DPTR,#ROMMSG          ;OUTPUT ROM MESSAGE
CALL      COMOUT                ;OUTPUT MESSAGE
MOV       A,R4                  ;
ANL       A,#0FOH              ;GET MSB
SWAP     A                      ; ROTATE TO MSB

CALL      BINOUT                ;CONVERT TO ASCII AND OUTPUT
MOV       A,R4                  ;GET DATA AND CONVERT
CALL      BINOUT                ;CONVERT TO ASCII AND OUTPUT

MOV       DPTR,#SLFFLG          ; SET SELFTEST FAILURE FLAG
MOVX     A,@DPTR
ORL      A,#SFFAIL              ;
MOVX     @DPTR,A
  
```



```

SLFRM3:
    MOV        RO, #00                ;CLEAR RETRY COUNTER
;LAP TOP USART SELFTEST

SLFDCH:
    CALL       WDRST                  ;RESET WATCHDOG
    CALL       LAPINL                 ;INITIALIZE USART FOR LAP TOP

    MOV        DPTR, #LAPSTS          ;READ DATA
    MOVX       A, @DPTR
    CJNE       A, #LAPCMP, SLFDC1

SLFDC0:
    MOV        RO, #00                ;CLEAR RETRY COUNTER
    JMP        SLFPRT                 ; AND CONTINUE

SLFDC1:
    %COMP      (SLFDCH, 19)           ;RERUN 19 TIMES IF NEEDED
    MOV        DPTR, #DCHFL          ;GET FAILED MESSAGE
    CALL       COMOUT                 ;AND OUTPUT

    MOV        DPTR, #SLFFLG         ; SET SELFTEST FAILURE FLAG
    MOVX       A, @DPTR
    ORL        A, #SFFAIL
    MOVX       @DPTR, A
;

;PRINTER USART SELFTEST

SLFPRT:
    CALL       WDRST                  ;RESET WATCHDOG
    CALL       PRTIAL                 ;INITIALIZE PRINTER USART

    MOV        DPTR, #PRTSTS          ;READ DATA
    MOVX       A, @DPTR
    CJNE       A, #PRTCMP, SLFPR1

SLFPRO:
    MOV        RO, #00                ;CLEAR RETRY COUNTER
    JMP        SLFITP                 ; AND CONTINUE

SLFPR1:
    %COMP      (SLFPRT, 19)           ;RERUN 19 TIMES IF NEEDED
    MOV        DPTR, #PRTFLL          ;GET FAILED MESSAGE
    CALL       COMOUT                 ;AND OUTPUT

    MOV        DPTR, #SLFFLG         ; SET SELFTEST FAILURE FLAG
    MOVX       A, @DPTR
    ORL        A, #SFFAIL
    MOVX       @DPTR, A
;

;I/O SELFTEST

SLFITP:
    CALL       WDRST                  ;RESET WATCHDOG
    CALL       IOINIT                 ;I/O SELFTEST

;PIA #1 SELFTEST

    MOV        DPTR, #ICSRI           ;GET STATUS
    MOVX       A, @DPTR
    ORL        A, #ITPST1
;
;ARE TIMER AND INTERRUPT
;ENABLES SET?
    CJNE       A, #ITPST1, SLFIT1    ;DID IT PAS
    MOV        RO, #00                ;CLEAR RETRY COUNTER
    JMP        SLFRAD                 ; AND CONTINUE

SLFIT1:
    %COMP      (SLFITP, 3)            ;RERUN 3 TIMES IF NEEDED
    MOV        DPTR, #IT1FLL          ;GET FAILED MESSAGE
    CALL       COMOUT                 ;AND OUTPUT

    MOV        DPTR, #SLFFLG         ; SET SELFTEST FAILURE FLAG
    MOVX       A, @DPTR
    ORL        A, #SFFAIL
    MOVX       @DPTR, A
;

```

;RADIO INTERFACE SELFTEST

SLFRAD:

```
CALL    WDRST      ;RESET WATCHDOG
CALL    RADINL     ;INITIALIZE RADIO USART
```

```
MOV     DPTR,#RADSTS ;READ DATA
MOVX    A,@DPTR
```

```
CJNE   A,#RADCOM,SLFRD1
```

SLFRD0:

```
MOV     R0,#00     ;CLEAR RETRY COUNTER
JMP     SLFPSD     ; AND CONTINUE
```

SLFRD1:

```
%COMP   (SLFRAD,19) ;RERUN 19 TIMES IF NEEDED
MOV     DPTR,#DCHFLL ;GET FAILED MESSAGE
CALL    COMOUT      ;AND OUTPUT
JNZ     SLFRAD      ;AND HALT
```

```
MOV     DPTR,#SLFFLG ; SET SELFTEST FAILURE FLAG
MOVX    A,@DPTR
ORL     A,#SFFAIL
MOVX    @DPTR,A
```

;SELFTEST PASSED

SLFPSD:

```
MOV     DPTR,#SLFPAS ;OUT PUT SELFTEST PASSED
CALL    COMOUT       ;MESSAGE
```

```
CALL    WDRST      ;RESET WATCHDOG
```

```
CALL    LORINL     ;INITIALIZE LORAN RECEIVER
```

;THE FOLLOWING IS FOR TRANSMIT TEST, MAINTENANCE, SHORT TIME

SLFINT:

```
CALL    WDRST      ;RESET WATCHDOG
```

```
JNB     P3.2,SLFINE ;JUMP IF NO KEYBOARD
```

```
MOV     DPTR,#MENU ;OUT PUT MENUE
CALL    COMOUT     ;MESSAGE
```

```
CALL    RCV
```

SLFINE:

```
CJNE   A,#'F',SLFPED ; NOT A 'F' CHECK FOR MEMORY
CALL    DATOUT       ; GO TRANSMIT AND DON'T COMEBACK
```

SLFPED:

```
;SEE IF MEMORY READ/WRITE
```

```
CJNE   A,#'R',SLFCDR ;NOT READ/WRITE MESSAGE
```

```
MOV     R3,#'D'     ;SET FOR DATA
```

```
CALL    MEMRWD     ;MEMORY READ/WRITE
JMP     SLFINT     ;EXIT MEMORY READ WRITE
```

SLFCDR:

```
CJNE   A,#'C',SLFAIR ;NOT READ/WRITE MESSAGE
```

```
MOV     R3,#'C'     ;SET FOR CODE
```

```
CALL    MEMRWD     ;MEMORY READ/WRITE
JMP     SLFINT     ;EXIT MEMORY READ WRITE
```

SLFAIR:

```
CJNE   A,#'D',SLFTXW ; NOT AIRCRAFT ID ENTRY
CALL    AIRIDE      ; IS AIRCRAFT ID ENTRY
JMP     SLFINT     ; SEE IF MORE ENTRIES
```

SLFTXW:

```
CJNE   A,#'V',SLFMFP ;NOT WAIT VALUE ENTRY
CALL    WTVALE      ; IS XMIT WAIT VALUE ENTRY
JMP     SLFINT     ; SEE IF MORE ENTRIES
```

```

SLFMFP:      CJNE     A, #'L', SLFMFR      ;NOT FOR MSG FRQ. FROM LORAN
              CALL    MSGFLP              ; GO CHANGE MESSAGE FREQ
              JMP     SLFINT              ; SEE IF MORE ENTRIES

SLFMFR:      CJNE     A, #'T', SLFLRI      ;NOT FOR MSG FRQ. FROM RADIO
              CALL    MSGFRD              ; GO CHANGE MESSAGE FREQ
              JMP     SLFINT              ; SEE IF MORE ENTRIES

SLFLRI:      CJNE     A, #'M', SLFALT      ;NOT FOR ENTERRING LORRAN DATA
              CALL    ENTLOR              ; GO ENTER LORRAN DATA
              JMP     SLFINT

SLFALT:      CJNE     A, #'A', SLFPEPT     ;NOT FOR CHANGING ALTIMETER INFO
              CALL    ENTALT              ; GO ENTER ALTIMETER DATA
              JMP     SLFINT              ; GO CONTINUE

SLFPEPT:     CJNE     A, #'P', SLFEND      ;SEE IF NEED TO GET PRINTER DATA
              CALL    ENTPRT
              JMP     SLFINT

;           END OF SELFTEST

```

```

SLFEND:      CALL    WDRST                ;RESET WATCHDOG

              MOV     IE, #0              ;DISABLE INTERRUPTS

              RET

```

```

;           *****
;           *****
;           * *
;           * *   S U B R O U T I N E S   * *
;           * *
;           *****
;           *****

```

IOINIT:

;INITIALIZATION OF PIA (8155) #1 PORTS A, B, C

```

              MOV     DPTR, #ICTMB1      ;SET SQUARE WAVE OUTPUT FREQ.
              MOV     A, #VICTM1        ;D7,D6=0,1  TIMER MODE

                                              ;D5-D0=1,0,0,1,0  5 MSB OF FREQ

              MOVX    @DPTR, A          ;OUTPUT

              MOV     DPTR, #ICTML1     ; 868.06 Hz
              MOV     A, #VICTL1        ;GET 8 LSB OF FREQ.

              MOVX    @DPTR, A          ;OUTPUT

              MOV     DPTR, #ICSR1      ;GET COMMAND REGISTER ADDRESS

              MOV     A, #INMOD          ;D7,D6=TIMER MODE SQUARE WAVE=1,1
                                              ;D5=INT ENB PORT B=0
                                              ;D4=INT ENB PORT A=0
                                              ;D3,D2=I/O PORT C (INPUTS)=0,0
                                              ;D1=I/O PORT B (INPUTS)=0
                                              ;D0=I/O PORT A (INPUTS)=0

              MOVX    @DPTR, A          ;SET COMMAND REG.

              MOV     A, #SETHGH        ;INIT HIGH

;TURN OFF THE FOLLOWING

              MOV     DPTR, #RADIOO     ;RADIO OFF
              MOVX    @DPTR, A

```

```

MOV    DPTR,#PTT_TK    ;DEASSERT PUSH TO TALK
MOVX   @DPTR,A

```

```

IOINEX:    RET                ;EXIT SET UP

```

```

;THE FOLLOWING SUBROUTINE WAITS 20 MS AFTER A WRITE HAS BEEN
; MADE TO AN EEPROM

```

```

WAITEP:
    PUSH    1EH
    PUSH    1FH                ;SAVE R6 & R7
    MOV     R7,#15
WATEP1:   MOV     R6,#0FFH
WATEP2:
    NOP
    DJNZ   R6,WATEP2
    DJNZ   R7,WATEP1
    POP    1FH
    POP    1EH
    RET                ;EXIT WAIT SUBROUTINE

```

```

;This subroutine transmits until an '.' is output at that
; time it returns.

```

```

COMOUT:
    MOV     A,#00H                ;CLEAR ACC
    MOV    @A+DPTR                ;OUTPUT DATA POINTER
    INC    DPTR
    CJNE   A,#' ',XMTOTO         ;TIME TO RESET WATCHDOG
    CALL   WDRST                 ; YES RESET
XMTOTO:
    CJNE   A,#' ',XMTOT1         ;ARE WE DONE?
    JMP    INTRS3                ;YES EXIT
XMTOT1:
    CALL   TRNOUT                 ; NO TRANSMIT
    JMP    COMOUT
INTRS3:
    RET                ;RETURN ENABLE INTERRUPTS

```

```

;This subroutine take the ACC and converts to ASCII and outputs it
; to the console

```

```

BINOUT:
    ANL    A,#0FH                ;STRIP OFF MSB
    %JAG   (9,BINOT1)           ;GREATER THAN 9
    ORL    A,#30H                ;MAKE ASCII
    JMP    BINOT2                ;CONTINUE
BINOT1:
    ADD    A,#37H                ;MAKE ASCII
BINOT2:
    CALL   TRNOUT                 ;OUTPUT TO CONSOLE
    RET                ;EXIT SUBROUTINE

```

```

;THE FOLLOWING ROUTINE RESETS THE WATCHDOG IF NOT TRANSMITTING

```

```

WDRST:
    PUSH   ACC
    PUSH   PSW
    PUSH   DPL
    PUSH   DPH
    MOV    A,#SETHGH            ;TURN OFF RADIO
    MOV    DPTR,#RADIOO
    MOVX   @DPTR,A

```

```

MOV      DPTR,#PTT_TK      ;SET PUSH TO TALK HIGH
MOVX    @DPTR,A           ; OR OFF
MOV      DPTR,#WDTIM      ;RESET WATCHDOG CHIP
MOVX    @DPTR,A

```

```

POP      DPH
POP      DPL
POP      PSW
POP      ACC

```

```
RET
```

```
;THIS SUBROUTINE READS AND WRITES MEMORY AS INPUT FROM MAINT CONSOLE
```

```

MEMRWD:
MOV      DPTR,#MEMDAT      ;GET ENTER DATA MESSAGE
CALL    COMOUT             ;OUTPUT
MEMWR:
MOV      DPTR,#CRLFDP      ;CARRIAGE RETURN LF
CALL    COMOUT
CALL    GETDAT             ;GET TWO CHARACTERS
MOV      DPH,A
MOV      A,R4
CJNE    A,#CR,MEMRWO      ;GOOD DATA CONT
JMP     MEMRWD
MEMRWO:
CALL    GETDAT
MOV      DPL,A            ; SAVE DATA
MOV      A,R4
CJNE    A,#CR,MEMRW1      ;GOOD DATA CONT
JMP     MEMRWD
MEMRW1:
CALL    MEMEQE            ;GO DISPLAY MEMORY AT DPTR
MEMRW2:
MOV      A,#' '           ;OUTPUT A SPACE
CALL    TRNOUT
CALL    GETDAT            ;GET NEXT DATA
MOV      R7,A             ;SAVE
MOV      A,R4             ; GET COMMAND ENTRY
CJNE    A,#CR,MEMRW3      ;GOOD CONT
INC      DPTR             ;GO TO NEXT LOCATION
CALL    MEMDIS            ;GO DISPLAY MEMORY AT DPTR
JMP     MEMRW2
MEMRW3:
CJNE    A,#' ',MEMRW4     ;CONTINUE
JMP     MEMWR             ;REENTER ADDRESS
MEMRW4:
CJNE    A,#'/',MEMRW5     ;DATA WRITE
CALL    MEMDIS            ;GO DISPLAY MEMORY AT DPTR
JMP     MEMRW2
MEMRW5:
CJNE    A,#'G',MEMRW6     ;CONTINUE
JMP     MEMRWT
MEMRW6:
MOV      A,#'C'           ; IS IT CODE?
CJNE    A,1BH,MEMRW7
JMP     MEMRW8
MEMRW7:
MOV      A,R7
MOVX    @DPTR,A          ;WRITE DATA AT LOCATION
MEMRW8:
CALL    MEMDIS            ;GO DISPLAY MEMORY AT DPTR
JMP     MEMRW2
MEMRWT:
RET
;THE FOLLOWING SUBROUTINE OUTPUTS A MEMORY DISPLAY AT DPTR
MEMDIS:
MOV      R1,DPH
MOV      R2,DPL
;SAVE DPTR

```

```

MOV     DPTR,#CRLFPD      ;CARRIAGE RETURN LF
CALL    COMOUT

MOV     A,R1              ;GET DPH
CALL    HEXOUT            ;OUTPUT MSB
MOV     A,R2              ;
CALL    HEXOUT            ;OUTPUT LSB

MOV     A,#'= '          ;OUTPUT A =
CALL    TRNOUT

MOV     DPL,R2            ;GET DATA
MOV     DPH,R1           ;AT DPTR

MOV     A,#'C'           ;GET CODE DISIGNATOR
CJNE   A,1BH,MEMDTA     ; NOT CODE BUT DATA
CLR    A
MOVC   A,@A+DPTR        ;GET CODE
JMP    MEMDT1

MEMDTA:
MOVX   A,@DPTR
MEMDT1:
CALL   HEXOUT           ;OUTPUT

MOV    DPH,R1
MOV    DPL,R2

RET

```

;THE FOLLOWING SUBROUTINE OUTPUTS AN EQUAL DISPLAY AT DPTR
MEMEQE:

```

MOV     R1,DPH
MOV     R2,DPL           ;SAVE DPTR

MOV     A,#'= '          ;OUTPUT A =
CALL    TRNOUT          ;OUTPUT

MOV     DPL,R2          ;GET DATA
MOV     DPH,R1         ;AT DPTR

MOV     A,#'C'           ;GET CODE DISIGNATOR
CJNE   A,1BH,MQEDTA    ; NOT CODE BUT DATA
CLR    A
MOVC   A,@A+DPTR       ;GET CODE
JMP    MQEDT1

MQEDTA:
MOVX   A,@DPTR
MQEDT1:
CALL   HEXOUT          ;OUTPUT

MOV    DPH,R1
MOV    DPL,R2

RET

```

;The following subroutine reads in the data for the Transmit Routine
; and then echo's it to the keyboard and then goes forever into the
; routine.

```

DATOUT:
SLFPS1:
MOV     DPTR,#ENTDAT     ;GET ENTER DATA MESSAGE
CALL    COMOUT          ;OUTPUT

CALL    RCV             ;GET RECEIVE CHARACTER
CALL    TRNOUT          ;ECHO IT OUT
MOV     R1,A            ;SAVE

%JAL   ('0',SLFPS1)     ;TOO LOW
%JAG   ('9',SLFPS2)     ;NOT NUMERIC

ANL    A,#0FH           ;MAKE NUMERIC
JMP    SLFPS3          ;CONTINUE

```

```

SLFPS2:
  %JAL      ('A',SLFPS1)      ;TOO LOW CONTINUE
  %JAG      ('F',SLFPS1)      ;TOO HIGH CONTINUE

  CLR       C
  ADD       A,#0AH            ;
  SUBB     A,#'A'            ;MAKE HEX

SLFPS3:
  SWAP     A                    ;ROTATE MSB LSB
  MOV      R5,A                ;SAVE

  CALL     RCV                  ;GET CHARACTER
  CALL     TRNOUT                ;ECHO IT OUT

  MOV      R2,A                ;SAVE

  %JAL     ('0',SLFPS1)        ;TOO LOW
  %JAG     ('9',SLFPS5)        ;NOT NUMERIC

  ANL     A,#0FH                ;MAKE NUMERIC
  JMP     SLFPS6                ;CONTINUE

SLFPS5:
  %JAL     ('A',SLFPS1)        ;TOO LOW CONTINUE
  %JAG     ('F',SLFPS1)        ;TOO HIGH CONTINUE

  CLR       C
  ADD       A,#0AH            ;
  SUBB     A,#'A'            ;MAKE HEX

SLFPS6:
  ORL     A,R5                    ;GET ALL OF DATA
  MOV     DPTR,#MODBYT          ;SET AT RAM
  MOVX    @DPTR,A                ;

  MOV     DPTR,#MSGTRN          ;SEND OUT TRANSMIT MESSAGE
  CALL    COMOUT

  MOV     A,R1                    ;
  CALL    TRNOUT                ;OUTPUT FIRST DIGIT

  MOV     A,R2                    ;OUTPUT SECOND DIGIT
  CALL    TRNOUT                ;

DATOTL:
  CALL    MODBUS                ; AND OUTPUT FOREVER
  JMP     DATOTL

;The following subroutine requests and enters the aircraft Id.

AIRIDE:
  MOV     DPTR,#AIRMSG          ;OUTPUT AIRCRAFT ID MESSAGE
  CALL    COMOUT
  MOV     DPTR,#AIRCID          ;GET AIRCRAFT ID LOCATION
  MOV     R2,#AIRIDN           ;GET NUMBER OF MESSAGES

AIRD01:
  CALL    RCV                    ;GET NUMBER OR WHAT EVER
  MOVX    @DPTR,A                ; SAVE IN EEPROM
  INC     DPTR                    ; GO TO NEXT LOCATION
  CALL    WAITEP                 ;WAIT FOR EEPROM
  CALL    TRNOUT                 ;ECHO TO SCREEN
  DJNZ   R2,AIRD01              ; RECEIVE WHOLE MESSAGE

  MOV     DPTR,#TKYMSG          ;SAY THANKYOU
  CALL    COMOUT

  RET                             ;EXIT SUBROUTINE

;The following subroutine requests and enters the Altimeter data.

ENTALT:
  MOV     DPTR,#ALTMSG          ;OUTPUT AIRCRAFT ID MESSAGE

```

```

CALL      COMOUT
MOV       DPTR, #ALTBUF      ;GET AIRCRAFT ID LOCATION
MOV       R2, #ALTLEN       ;GET NUMBER OF MESSAGES
ENTL1:
CALL      RCV                ;GET NUMBER OR WHAT EVER
MOVX     @DPTR, A           ; SAVE IN EEPROM
INC      DPTR                ; GO TO NEXT LOCATION
CALL     WAITEP              ;WAIT FOR EEPROM
CALL     TRNOUT              ;ECHO TO SCREEN
DJNZ    R2, ENL1            ; RECEIVE WHOLE MESSAGE

MOV       DPTR, #TKYMSG      ;SAY THANKYOU
CALL     COMOUT

MOV       DPTR, #ALTINH      ;SET ALTIMETER INHIBIT
CLR      A
CPL     A
MOVX     @DPTR, A

RET                               ;EXIT SUBROUTINE

```

;The following subroutine requests and enters the LORAN Id.

```

ENTLOR:
MOV       DPTR, #LNGMSG      ;OUTPUT LONGITUDE MESSAGE
CALL     COMOUT
MOV       DPTR, #PLNBUF      ;GET LONGITUDE LOCATION
MOV       R2, #PLNLEN       ;GET NUMBER OF CHAR
ENTLN1:
CALL      RCV                ;GET NUMBER OR WHAT EVER
MOVX     @DPTR, A           ; SAVE IN EEPROM
INC      DPTR                ; GO TO NEXT LOCATION
CALL     WAITEP              ;WAIT FOR EEPROM
CALL     TRNOUT              ;ECHO TO SCREEN
DJNZ    R2, ENLN1           ; RECEIVE WHOLE MESSAGE

MOV       DPTR, #LATMSG      ;OUTPUT LATITUDE MESSAGE
CALL     COMOUT
MOV       DPTR, #PLTBUF      ;GET LONGITUDE LOCATION
MOV       R2, #PLTLEN       ;GET NUMBER OF CHAR
ENTLT1:
CALL      RCV                ;GET NUMBER OR WHAT EVER
MOVX     @DPTR, A           ; SAVE IN EEPROM
INC      DPTR                ; GO TO NEXT LOCATION
CALL     WAITEP              ;WAIT FOR EEPROM
CALL     TRNOUT              ;ECHO TO SCREEN
DJNZ    R2, ENLT1           ; RECEIVE WHOLE MESSAGE

MOV       DPTR, #TKYMSG      ;SAY THANKYOU
CALL     COMOUT

RET                               ;EXIT SUBROUTINE

```

;The following subroutine requests and enters the wait value.

```

WTVALE:
MOV       DPTR, #WTVMSG      ;OUTPUT WAIT TILL CHANNEL CLEAR
CALL     COMOUT              ; MESSAGE
MOV       DPTR, #VRNTIM      ;GET EEPROM LOCATION

CALL     GETDAT              ;GET HEX CHARACTER
MOVX     @DPTR, A           ; SAVE IN EEPROM
CALL     WAITEP              ;WAIT FOR EEPROM

MOV       DPTR, #TKYMSG      ;SAY THANK YOU
CALL     COMOUT

RET                               ;EXIT SUBROUTINE

```

;The following subroutine requests and enters the message frequency
; for the transmission of the aircraft frequency to the Lap Top


```

MSGFLP:
  MOV     DPTR,#FRLMSG      ;OUTPUT TIME IN SECONDS
                                ; TO OUTPUT TO LAP TOP
                                ; MESSAGE
  CALL    COMOUT
  MOV     DPTR,#VLRcnt      ;GET RAM LOCATION
  CALL    GETDAT            ;GET HEX CHARACTER
  MOVX   @DPTR,A           ; SAVE IN RAM
  MOV     DPTR,#TKYMSG      ;SAY THANK YOU
  CALL    COMOUT
  RET                                ;EXIT SUBROUTINE

```

;The following subroutine requests and enters the message frequency
; for the transmission of the aircraft frequency to the Other Aircraft
; over the radio.

```

MSGFRD:
  MOV     DPTR,#FRMSG      ;OUTPUT TIME IN SECONDS
                                ; TO OUTPUT TO AIRCRAFT
                                ; MESSAGE
  CALL    COMOUT
  MOV     DPTR,#VTRCNT      ;GET RAM LOCATION
  CALL    GETDAT            ;GET HEX CHARACTER
  MOVX   @DPTR,A           ; SAVE IN RAM
  MOV     DPTR,#TKYMSG      ;SAY THANK YOU
  CALL    COMOUT
  RET                                ;EXIT SUBROUTINE

```

;THE FOLLOWING SUBROUTINE CONVERTS THE DATA IN THE ACC TO ASCII AND
; OUTPUTS IT

```

HEXOUT:
  PUSH   ACC
  SWAP   A                  ;GET MSB
  CALL   BINOUT             ;OUTPUT IT
  POP    ACC
  CALL   BINOUT             ;GET LSB
                                ; AND OUTPUT IT
  RET

```

;The following subroutine requests and enters data to enable the printer.

```

ENTPRT:
  MOV     DPTR,#PRTMSG      ;OUTPUT PRINTER ID MESSAGE
  CALL    COMOUT
  CALL    RCV               ;GET NUMBER OR WHAT EVER
  MOV     DPTR,#PRTFLG      ;GET EEPROM ADDRESS
  MOVX   @DPTR,A           ; SAVE IN EEPROM
  CALL    WAITEP            ;WAIT FOR EEPROM
  CALL    TRNOUT           ;ECHO TO SCREEN
  MOV     DPTR,#TKYMSG      ;SAY THANKYOU
  CALL    COMOUT
  RET                                ;EXIT SUBROUTINE

```

;THIS SUBROUTINE RETURNS THE TWO BYTE DATA ENTERED

```

GETDAT:
  MOV     R4,#0             ;CLEAR R4
  CALL    RCV               ;READ DATA
  %JAL   ('0',MEMPS1)       ;TOO LOW
  %JAG   ('9',MEMPS2)       ;NOT NUMERIC

```

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```

CALL      TRNOUT          ;OUTPUT

ANL      A,#OFH          ;MAKE NUMERIC
JMP      MEMPS3          ;CONTINUE

MEMPS2:
%JAL     ('A',MEMPS1)    ;TOO LOW CONTINUE
%JAG     ('F',MEMPS1)    ;TOO HIGH CONTINUE

CALL     TRNOUT          ;OUTPUT

CLR      C
ADD      A,#0AH          ;
SUBB     A,#'A'          ;MAKE HEX

MEMPS3:
SWAP     A                ;ROTATE MSB,LSB
MOV      R5,A            ;SAVE

CALL     RCV             ;GET DATA

%JAL     ('0',MEMPS1)    ;TOO LOW
%JAG     ('9',MEMPS5)    ;NOT NUMERIC

CALL     TRNOUT          ;OUTPUT

ANL      A,#OFH          ;MAKE NUMERIC
JMP      MEMPS6          ;CONTINUE

MEMPS5:
%JAL     ('A',MEMPS1)    ;TOO LOW CONTINUE
%JAG     ('F',MEMPS1)    ;TOO HIGH CONTINUE

CALL     TRNOUT          ;OUTPUT

CLR      C
ADD      A,#0AH          ;
SUBB     A,#'A'          ;MAKE HEX

MEMPS6:
ORL      A,R5            ;GET ALL OF DATA

JMP      MEMEXT          ;EXIT

MEMPS1:
MOV      R4,A

MEMEXT:
RET                      ;EXIT SUBROUTINE

```

;THE FOLLOWING SUBROUTINE OUTPUTS THE DATA IN THE ACC

```

TRNOUT:
PUSH     ACC
PUSH     1AH             ;SAVE

MOV      R2,#8           ;SET UP FOR 8 DATA BITS

CLR      P1.2           ;SET START BIT LOW
CALL     WAIT2          ; WAIT

LPTRN:
RRC      A               ;GET BIT IN CARRY
JNC      LPTRN1         ;SET BIT
SETB     P1.2           ;CARY SET OUTPUT HIGH
JMP      LPTRN2         ; CONTINUE

LPTRN1:
CLR      P1.2           ;SET LOW

LPTRN2:
CALL     WAIT2          ; GO WAIT
DJNZ    R2,LPTRN        ; LOOP UNTILL DONE

SETB     P1.2           ;STOP BITS

MOV      R2,#0CH        ;WAIT FOR STOP BITS

```

```

LPTRN3:
    CALL    WAIT2
    DJNZ   R2,LPTRN3

    POP    1AH          ;RESTORE
    POP    ACC

    RET                    ;EXIT SUBROUTINE

```

```

;
;The following is the RAM memory test. It does the following:

```

```

; Write each pattern found at RAMTAB to memory
; from address RAMBG to RAMED then read it back.
; if there is ever a change it will stop and put
; the error address and written and read pattern
; at RAMADD, RAMWRT and RAMRED respectively. Will always
; end with writing 00H to each address

```

```

RAMTAB:
    DB      55H,0AAH,0FFH,0H

```

```

RAMTST:

```

```

;CHECKERBOARD TEST AT USED RAM LOCATIONS

```

```

RAMIT3:
    MOV     R1,#0          ;GET BEGINING OF TABLE
    MOV     DPTR,#RAMTAB  ;GET BYTE TO WRITE
    MOV     A,R1          ;GET TABLE ENTRY
    MOVC   A,@A+DPTR     ;

    MOV     R5,A          ;SAVE
    INC    R1            ;GET NEXT VALUE IF NEEDED

    MOV     R2,#RAMBG    ;GET START ADDRESS TO CHECK
    MOV     R3,#RAMED    ;GET STOPPING ADDRESS

    CALL   RAMCK         ;GO CHECK EACH LOCATION

    JC     EXIT          ;ERROR EXIT

    MOV     A,R5          ;SEE IF DONE
    JNZ   RAMIT3        ; CONTINUE UNTIL DONE

```

```

;SET DIAG. REPORTING TO ALL FF'S

```

```

    MOV     DPTR,#RAMADD ;GET RAM PASS , FAIL ADDRESS
    MOV     A,#0FFH      ;SET ALL DATA TO 0FFH
    MOVX   @DPTR,A       ;SAVE
    INC    DPTR          ;GET NEXT ADDRESS
    MOVX   @DPTR,A       ;
    INC    DPTR          ;SAVE DATA WRITEN
    MOVX   @DPTR,A       ;
    INC    DPTR          ;SAVE DATA READ
    MOVX   @DPTR,A       ;

```

```

EXIT:

```

```

    RET                    ;EXIT TEST

```

```

;THE FOLLOWING IS TO THE CODE TO FIND THE CHECKSUM. IT ADDS
; UP ALL VALUES FROM ROMBG TO ROMED AND PUTS THE VALUE AT
; CKSMAD

```

```

CKSADJ:    DB      0ACH          ;CHECKSUM ADJUST TO 00

```

```

ROMCKS:

```

```

    MOV     DPH,#ROMBG ;GET BEGINING ADDRESS
    MOV     DPL,#00H   ;

```

```

MOV      A,#0H      ;CLEAR USED BUFFERS
MOV      R4,#00H   ;

ROM1:
MOV      A,#0H      ;CLEAR ACC
CLR      C          ;CLEAR CARRY BIT
MOVC     A,@A+DPTR  ;GET DATA
INC      DPTR       ; GO TO NEXT ADDRESS
ADD      A,R4       ; AND ADD
MOV      R4,A       ; AND SAVE

MOV      A,#ROMED   ;SEE IF THROUGH
CJNE     A,DPH,ROM1 ; NOT THROUGH JUMP

MOV      A,R4       ;GET CHECKSUM
MOV      DPTR,#CKSMAD ; AND STORAGE ADDRESS
MOVX     @DPTR,A    ; AND SAVE

RET      ;EXIT TEST

;
;
; This subroutine writes the value of the ACC from high order address
; at R2 to but not including the high order address at R3.
;
RAMCK:
;WRITE TO ALL LOCATIONS

MOV      DPH,R2     ;GET STARTING ADDRESS
MOV      DPL,#00H   ;

RAMCK1:
MOV      A,R5       ;GET VALUE
MOVX     @DPTR,A    ;WRITE ACC
INC      DPTR
MOV      A,R3       ;GET HIGH ORDER BYTE
CJNE     A,DPH,RAMCK1 ;NOT THROUGH JUMP

;READ AT ALL LOCATIONS

MOV      DPH,R2     ;GET STARTING ADDRESS
MOV      DPL,#00H   ;

RAMCK3:
MOVX     A,@DPTR    ;GET DATA AT ADDRESS
CJNE     A,1DH,ERR  ;ERROR EXIT
INC      DPTR       ; INC ADDRESS
MOV      A,R3       ;GET END
CJNE     A,DPH,RAMCK3 ;NOT THROUGH JUMP
CLR      C          ;SET DONE
RET      ;EXIT SUBROUTINE

ERR:
MOV      R6,A       ;SAVE BAD DATA
MOV      A,DPH      ;GET HIGH ORDER
MOV      R1,DPL     ; SAVE LOW ORDER
MOV      DPTR,#RAMADD ;GET RAM ADDRESS
MOVX     @DPTR,A    ;
INC      DPTR       ;SAVE BAD ADDRESS
MOV      A,R1       ;
MOVX     @DPTR,A    ;
INC      DPTR       ;SAVE DATA WRITTEN
MOV      A,R5       ;
MOVX     @DPTR,A    ;
INC      DPTR       ;SAVE DATA READ
MOV      A,R6       ;
MOVX     @DPTR,A    ;

SETB     C          ;INDICATE ERROR
RET      ;EXIT THIS MODULE

```

```

;THE FOLLOWING SUBROUTINE WAITS ONE FULL BIT
WAIT2:

```

```

      CALL SWAIT      ;WAIT 1/2
WAIT1: CALL SWAIT      ;WAIT 1/2
      RET            ;EXIT WAIT

```

```
;THIS SUBROUTINE WAITS 1/2 OF BIT LENGTH
```

```

SWAIT:
  PUSH 1FH      ;
  PUSH 1EH
      MOV R7,#14H ;
SWAIT1:
  MOV R6,#3H
SWAIT2:
      DJNZ R6,SWAIT2
      NOP
      DJNZ R7,SWAIT1
      POP 1EH
      POP 1FH
      RET

```

```
;THE FOLLOWING IS FOR THE RECEIVE ON THE COM PORT
; AS WELL AS THE INTERRUPT HANDLER
```

```
CONINT:
```

```

RCV:
  PUSH PSW      ;SAVE PSW
  PUSH 1AH      ;SAVE R2
  PUSH DPL      ;SAVE DPTR
  PUSH DPH
      MOV R2,#08H      ;SET FOR 8 DATA BITS
RCV1:  JB P1.3,RCV1    ;WAIT UNTIL STOP BIT
      CALL WAIT1      ;WAIT 1/2 BIT
RCV2:  CALL WAIT2      ;WAIT 1 BIT
      JB P1.3,RCV3    ;IS DATA BIT LOW OR HIGH
      CLR C           ;CLEAR CARRY
      JMP RCV4        ;CONTINUE
RCV3:  SETB C          ;SET CARRY
RCV4:  RRC A          ;ROTATE IN
      DJNZ R2,RCV2
      MOV R2,#4       ;WAIT FOR 2 STOP BITS AND A
RCV5:  ; LITTLE EXTRA
      CALL WAIT2      ;
      DJNZ R2,RCV5
      POP DPH        ;RESTORE DPTR
      POP DPL
      POP 1AH
      POP PSW
      RET
      END

```

```
$TITLE (INITIALIZATION 8052 BASED)
NAME INITAL
```

```
;ACAS VA001 ORIGINAL RELEASE 26 JAN 90 14:12
```

```
;LAST CHANGE VA001 02 FEB 90 10:43
```

```
;Initialization Used for serial, D/A and discrete I/O
```

```
;Basic Description of Module
```

```
;This module houses the interrupt vectors, including start, calls Selftest
; initializes that not done in Selftest, and returns control to the main
; Executive Loop.
```

```
$EJECT
```

```
;
; *****
; *****
; * * * * *
; * * P U B L I C S * *
; * * * * *
; *****
; *****
PUBLIC INITAL ;Input to this module
```

```
$EJECT
```

```
;
; *****
; *****
; * * * * *
; * * M A C R O S * *
; * * * * *
; *****
; *****
```

```
;%DEFINE (JALE (CDATA, DEST)) LOCAL LABEL(
```

```
PUSH ACC
CLR C
SUBB A, %%CDATA
POP ACC
JC %DEST
CJNE A, %%CDATA, %LABEL
JMP %DEST
```

```
%LABEL:
```

```
)
```

```
;WAIT FOR EEPROM TO BE READY
```

```
;%DEFINE (WAITE) LOCAL LABEL
```

```
(
%LABEL:
JNB P1.5, %LABEL
)
```

```
;%DEFINE (INTHND(TYPSTS, TYPCNT, TYPVAL, TYPIT, USRTIN, XMPT)) LOCAL LABEL1 LABEL2
```

```
(
MOV DPTR, %%TYPSTS ;'SEE IF INTERRUPT REQUEST TRANSMIT'
MOVX A, @DPTR
ANL A, %%XMPT
JZ %LABEL2 ;' NO JUMP'
MOV DPTR, %%TYPCNT ;'GET INTERRUPT COUNT'
MOVX A, @DPTR
INC A
MOVX @DPTR, A ;'INC INTERRUPT COUNT'
CJNE A, %%TYPVAL, %LABEL1 ;'COUNT AT MAX?'
CLR A
MOVX @DPTR, A ;'CLEAR COUNT'
CALL %TYPIT ;'GO REINITIALIZE'
JMP %LABEL2
```

```

%LABEL1:
CALL    %USRTIN          %'GO HANDLE TRN. INTERRUPT'
%LABEL2:
)

$EJECT
;
; *****
; *****
; * * * * *
; * * * * *
; * * * * *
; * * * * *
; *****
; *****

; Interrupt handlers for different modules
;
EXTRN CODE (TIMIN)          ; SPARE
                           ; TIMER

EXTRN CODE (CONINT)        ;KEYBOARD (DIAGNOSTICS) INTERRUPT

EXTRN CODE (LAPRIN,LAPTIN) ; LAP TOP RECEIVE & TRANSMIT INTERRUPT
EXTRN CODE (LORRIN)        ; LORAN RECEIVE INTERRUPT HANDLER
EXTRN CODE (RADRIN,RADTIN) ; RADIO INTERRUPT HANDLERS
EXTRN CODE (PRTTIN)        ; INTERRUPT HANDLER FOR PRINTER

;Selftest Entry Point

EXTRN CODE (SLFTST)

; Initialization for USARTS

EXTRN CODE (LAPINL)        ; LAP TOP COMPUTER INITIALIZATION
EXTRN CODE (RADINL)        ; RADIO INITIALIZATION
EXTRN CODE (PRTIAL)        ; INITIALIZATION FOR PRINTER

; Interrupt counters

EXTRN XDATA (LAPTCT,LAPRCT) ;FOR LAP TOP COMPUTER
EXTRN XDATA (RADTCT,RADRCT) ;FOR RADIO
EXTRN XDATA (PRTTCT)

;EXTERNAL CODE FOR BEGIN OF PROGRAM

EXTRN CODE (EXECLP)

;USART Status addresses

EXTRN NUMBER (LAPSTS,RADSTS) ;LAP TOP, RADIO
EXTRN NUMBER (PRTSTS,TERSTS) ;PRINTER, TERMINAL

;Interrupt counter values

EXTRN NUMBER (LAPRVL,LAPTVL) ;LAP TOP COMPUTER
EXTRN NUMBER (RADRVL,RADTVL) ;RADIO
EXTRN NUMBER (TERRVL,TERTVL) ;TERMINAL
EXTRN NUMBER (PRTTVL)        ;PRINTER

;USART Ready or Empty flags

EXTRN NUMBER (TXRDY,TXEMPT,RXRDY) ;DATA EMPTY FLAGS

;INTERNAL INITIALIZATION

EXTRN NUMBER (VTH1,VTH0,VTLO) ;TIMER 0 & 1 VALUES

EXTRN NUMBER (VTMOD)          ;TIMER MODE
EXTRN NUMBER (PCON)          ;PCON IN INTERNAL MEMORY

;Reset Control

EXTRN NUMBER (VRSTCT)        ;RESET COUNT MAX
EXTRN NUMBER (SETHGH)        ;TURN OFF RADIO

```

```
EXTRN NUMBER (SETLOW)      ;TURN ON RADIO
EXTRN NUMBER (RADIOO)     ;RADIO CONTROL
```

```
$EJECT
```

```
; *****
; *****
; * * * * *
; * * * * * P R O G R A M * * * * *
;
; * * * * *
; *****
; *****
```

```
CSEG
```

```
ORG 00H
```

```
START:
```

```
JMP EXECLP ;GO BEGIN CODE
```

```
ORG 03H
```

```
JMP CONINT ;DIAGNOSTIC START BIT
```

```
ORG 0BH
```

```
JMP TIMIN ;TIMER INTERRUPT
```

```
ORG 013H
```

```
JMP INT1A ;INTERRUPT VECTOR
```

```
ORG 23H
```

```
JMP LORRIN ;GO HANDLE NAVIGATION INTERRUPT
```

```
;
PROG_S segment CODE
RSEG PROG_S
```

```
;THE FOLLOWING IS THE INTERRUPT VECTOR FOR THE USARTS
```

```
;
INT1A:
```

```
CLR EA ;DISABLE INTERRUPTS
```

```
PUSH PSW
```

```
PUSH DPH
```

```
PUSH DPL
```

```
PUSH ACC
```

```
;RADIO TRANSMIT USART
```

```
%INTHND (RADSTS,RADTCT,RADTVL,RADINL,RADTIN, TXRDY)
```

```
;RADIO RECIEVE USART
```

```
%INTHND (RADSTS,RADRCT,RADRVL,RADINL,RADRIN, RXRDY)
```

```
;LAP TOP TRANSMIT USART
```

```
%INTHND (LAPSTS,LAPTCT,LAPTVL,LAPINL,LAPTIN, TXRDY)
```

```
;LAP TOP RECEIVE USART
```

```
%INTHND (LAPSTS,LAPRCT,LAPRVL,LAPINL,LAPRIN, RXRDY)
```

```
;TERMINAL TRANSMIT USART
```

```
; INTHND (TERSTS,TERTCT,TERTVL,TRINIT,TRTINT, TXRDY)
```

```
;TERMINAL RECEIVE USART
```

```
; INTHND (TERSTS,TERRCT,TERRVL,TRINIT,TRRINT, RXRDY)
```

```
;PRINTER RECEIVE USART
```

```
%INTHND (PRTSTS,PRTTCT,PRTTVL,PRTIAL,PRTTIN, TXEMPT)
```



```
POP    ACC
POP    DPL
POP    DPH
POP    PSW
```

```
SETB   EA           ;ENABLE INTERRUPTS
RETI
```

;

; THE FOLLOWING INITIALIZES ALL OF THE FOLLOWING:

INITIAL:

```
ORL     PSW,#18H      ;SELECT REG. BANK 3

SETB    P1.4         ;TURN ON PROCESS FAIL LED

;INITIALIZATION AND TEST OF USARTS AND ROM AND RAM

CALL    SLFTST       ;GO DO SELFTEST

;SET UP INTERNAL CPU REGISTERS

;TIMER 0 FOR 16 BIT TIMER FOR TIMIN
;TIMER 1 FOR BAUD RATE GEN FOR COM PORT

MOV     TMOD,#VTMOD   ;SET TMOD FOR
;TIMER 0 = TIMER
;TIMER 1 = BAUD RATE GENERATOR

MOV     TH1,#VTH1     ;SET BAUD RATE FOR COM PORT

MOV     TH0,#VTH0     ;SET TIMER 0
MOV     TLO,#VTLO

SETB    TRO           ;START INTERNAL TIMER
SETB    TR1           ;START BAUD RATE GENERATOR

SETB    SM1           ;MODE 2 IN SCON
SETB    REN           ;ENABLE RECEIVE

MOV     PCON,#80H     ;SET SMOD

SETB    PT1           ;SET INTERRUPT 1 HIGHEST PRIORITY

;ENABLE INTERRUPTS
SETB    EX1           ;EX1 = EXTERNAL INTERRUPT 1 USARTS
CLR     EX0           ;EX0 = EXTERNAL INTERRUPT 0 DIAG.
SETB    ES            ;ENABLE SERIAL PORT
SETB    ETO           ;ETO = INTERNAL TIMER INTERRUPT 0
CLR     P1.4         ;CLEAR PROCESS FAIL LED
SETB    EA            ;EA = ENABLE ALL

MOV     DPTR,#RADIOO  ;TURN ON RADIO
MOV     A,#SETLOW
MOVX    @DPTR,A

RETI                ;EXIT INITIALIZATION
```

;END OF PROGRAM

;
END

```
$TITLE (EXECUTIVE 8052 BASED)
NAME EXECLP ;
```

;ORIGIONAL RELEASE A001 20 JAN 90

;
;LAST CHANGE 22 MAY 90 08:14

```
;VA002 15 FEB 90 ADDED LORAN STATUS
;VA003 23 FEB 90 CHANGED LORAN METHOD OF UPDATE
```

```

;VA004 15 MAY 90  XMIT 1/2 AFTER RCV
;                RECEIVE LED
;                ATTACK DELAY 180MS
;VA005 21 MAY 90  ELIMINATED DOUBLE RESET
;VA006 22 MAY 90  FIXED ABOVE

```

```

;Basic Description of Module:

```

```

; This module is the main loop that calls all other assembly
; language routines in the main program. It also defines all
; public definitions and equates.
;
;
;

```

```

;*****
;*****
; * * * * *
; * * * * *
; * * * * *
; * * * * *
; * * * * *
;*****
;*****

```

```

PUBLIC      EXECLP      ;INPUT FROM INIT
;
;

```

```

;*****
;*****
; * * * * *
; * * * * *
; * * * * *
; * * * * *
;*****
;*****

```

```

EXTRN XDATA      (TWDTM)                ;WATCHDOG TIMER COUNTER
EXTRN XDATA (TRERUN)                ;RERUN TIMER

```

```

$EJECT
;
;
;
;
;

```

```

;*****
;*****
; * * * * *
; * * * * *
; * * * * *
;*****
;*****

```

```

;*****
;*****
;

```

```

;WAIT FOR EEPROM TO BE READY

```

```

%*DEFINE (WAITE) LOCAL LABEL
(

```

```

%LABEL:
    JNB P1.5,%LABEL
)

```

```

;CALL TO LOOPED MODULES

```

```

%*DEFINE (CALLIT (MODULE))
(

```

```

EXTRN CODE (%MODULE)
    LCALL    %MODULE
)

```

```

;JUMP (LONG JUMP) IF A> CDATA

```

```

%*DEFINE (JAG (CDATA,DEST)) LOCAL LABEL(
    PUSH ACC
    CLR C
    SUBB A,%CDATA
    JZ %LABEL
    JC %LABEL
)

```

```

        POP    ACC
        JMP    %DEST
%LABEL:
        POP    ACC
    )

%*DEFINE (DSE (TLABEL,DSV))
    (
PUBLIC    %TLABEL
%TLABEL: DS    %DSV
    )
;

; EQUATE AND MAKE PUBLIC
;
%*DEFINE (EQE (TLABEL,VALUE))
    (
PUBLIC    %TLABEL
%TLABEL EQU    %VALUE
    )
;
$EJECT
;
; *****

; *****
; * * * * *
; * * * * *   D A T A   * * * * *
; * * * * *
; *****
; *****

D_ABS    segment    XDATA
D_NONV   segment    XDATA
DATA_AREA segment    XDATA
PROG_S   segment    CODE

;*****ABSOLUTE DATA AREA*****

RSEG     D_ABS

;THE FOLLOWING IS A DATA AREA THAT IS USED EXTERNALLY AND SHOULD
; NOT BE CHANGED

%DSE (CKSMAD,1)          ;ROM CHECKSUM ADDRESS
%DSE (RAMADD,5)          ;RAM PASS FAIL AND ADDRESS
%DSE (MODBYT,1)          ;RADIO TEST ADDRESS
%DSE (RERUNA,1)          ;RERUN ADDRESS

;*****NONVOLATILE RAM AREA*****

RSEG     D_NONV

%EQE (AIRIDN,6)          ;NUMBER OF DIGITS IN AIRCRAFT ID
;*****A CHANGE OF ABOVE EMGLEN IN RADINT MUST ALSO BE CHANGED
%DSE (AIRCID,AIRIDN)     ;AIRCRAFT ID
%DSE (VRNTIM,1)          ;RANDOM WAIT TIMER

;*****Printer Flags*****
%DSE (PRTFLG,1)          ;VOLATILE RAM
%EQE (PRTVAL,'P')        ;VALUE FOR PRINTER ATTACHED

;*****RAM AREA*****

RSEG     DATA_AREA

;*****INTERRUPT COUNTERS*****

%DSE (LAPRCT,1)          ;LAP TOP RECEIVE
%DSE (LAPTCT,1)          ;LAP TOP TRANSMIT

%DSE (RADRCT,1)          ;RADIO RECEIVE
%DSE (RADTCT,1)          ;RADIO TRANSMIT

```

```

%DSE (LRRCNT,1) ;LORAN INTERRUPT COUNT
%DSE (PRTTCT,1) ;PRINTER TRANSMIT
%DSE (TERRCT,1) ;TERMINAL RECEIVE
%DSE (TERTCT,1) ;TERMINAL TRANSMIT

;*****Altimeter Receiver buffer and lengths

%EQE (ALTLEN,3) ;Altimeter buffer length
%DSE (ALTBUF,ALTLEN) ;Altimeter buffer

%DSE (ALTIINH,1) ;Altimeter inhibit

%EQE (ALT9IN,01H) ;FOR 9 STROBE ALTIMETER
%EQE (ALT11N,07H) ;FOR 11 STROBE ALTIMETER

;*****Lorran Receiver buffer and lengths

%EQE (PLNLEN,8) ;Longitude length
%DSE (PLNBUF,PLNLEN) ;Longitude buffer
%DSE (TLNBUF,PLNLEN) ;Temporary Longitude buffer

%EQE (PLTLEN,7) ;Latitude length
%DSE (PLTBUF,PLTLEN) ;Latitude buffer
%DSE (TLTBUF,PLTLEN) ;Temporary attitude buffer

%EQE (STSLEN,11) ;Loran Status Length
%DSE (STSBUF,STSLEN) ;Loran Status buffer
%EQE (TLORST,3) ;Actual transmit length of above

%DSE (LRTLTF,1) ;Loran Data Ready for Lap Top
%DSE (LRTRDF,1) ;Loran Data Ready for Radio

;*****Radio Handler Flags and buffers*****
;*****RECEIVE*****
%DSE (RDTLTF,1) ;Message from Radio to Lap Top Flag
SOHLEN EQU 02 ;START OF HEADER LENGTH

%EQE (MSGLEN,AIRIDN+PLTLEN+PLNLEN+ALTLEN+TLORST+SOHLEN) ;Message Length
%DSE (RDTLTB,MSGLEN) ;Message from Radio to Lap Top
%DSE (MINCKS,1) ;Check sum address
%DSE (RDTLTE,1) ;End of RDTLTB

;*****Radio Handler Flags and buffers*****
;*****Transmit*****

%EQE (TRNMAX,1) ;Number of times to transmit

%EQE (MSGTLN,MSGLEN+1+SOHLEN) ;ENTIRE TRANSMIT LENGTH

;*****CAUTION DO NOT ALTER THE FOLLOWING ORDER*****
%DSE (STTRAN,SOHLEN) ;AREA FILLED WITH START OF HEADERS
%DSE (RADOTB,MSGLEN) ;BUFFER FOR MESSAGE
%DSE (TRNCKS,1) ;CHECKSUM ADDRESS

DS (TRNMAX-1)*MSGTLN

%DSE (EDTRAN,0) ;END OF BUFFER
;*****

%DSE (VTRCNT,1) ;COUNT FOR MESSAGES TO RADIO

;*****Lap Top Handler Flags*****

%DSE (VLRCNT,1) ;COUNT BETWEEN MESSAGES
%DSE (NWDRST,1) ;NO WATCH DOG RESET

;*****SelfTest FLAGS*****

%DSE (SLFFLG,1) ;SELFTEST FLAG
%EQE (SFFAIL,01H) ;DVC SELFTEST FAILURE FLAG
%EQE (NOLRRD,02H) ;NO LORAN UPDATE FOR RADIO
%EQE (NOLRLT,04H) ;NO LORAN UPDATE FOR LAPTOP

```

```
%EQE (TRRDST,08H) ;NOT TRANSMITTED TO RADIO
%EQE (TRLTST,10H) ;NOT TRANSMITTED TO LAPTOP
%EQE (OUTFLG,20H) ;NOT ACCEPTING OTHER AIRCRAFT MSG'S
```

\$EJECT

```
; *****
; *****
; * * * * *
; * * EQUATES * *
; * * * * *
; *****
; *****
```

```
%EQE (VRSION,0A006H) ;VERSION NUMBER
```

```
;***** 8155 PIA #1*****
```

```
%EQE (ICSR1,0A010H) ;INT. CMD STATUS REG
%EQE (ICTMB1,0A015H) ;MSB FOR SQUARE WAVE FREQ.
```

```
; (1800 BAUD)
;SET SQUARE WAVE OUTPUT FREQ.
;D7,D6=0,1 TIMER MODE
;D5-D0=1,0,0,1,0 5 MSB OF FREQ
```

```
%EQE (VICTM1,41H) ;VALUE FOR MSB ABOVE (1200 BAUD)
%EQE (ICTML1,0A014H) ;LSB FOR SQUARE WAVE FREQ.
%EQE (VICTL1,038H) ;VALUE FOR LSB ABOVE (1200 BAUD)
%EQE (ITPST1,0CFH) ;STATUS VERIFICATION
```

```
%EQE (OUTMOD,0CCH) ;PORT C OUTPUT
%EQE (INMOD,0COH) ;PORT C INPUT
;D7,D6=TIMER MODE SQUARE WAVE=1,1
;D5=INT ENB PORT B=0
;D4=INT ENB PORT A=0
;D3,D2=I/O PORT C (INPUTS)=0,0
;D3,D2=I/O PORT C (OUTPUTS)=1,1
;D1=I/O PORT B (INPUTS)=0
;D0=I/O PORT A (INPUTS)=0
```

```
; PORT A (INPUTS)
%EQE (PORTA1,0A011H)
```

```
;ADD FOR PORT A
```

```
%EQE (ALT_A1,01H) ;01H - ALTIMETER INPUT A1 (PA0)
%EQE (ALT_D4,02H) ;02H - ALTIMETER INPUT D4 (PA1)
%EQE (ALT_D2,04H) ;04H - ALTIMETER INPUT D2 (PA2)
;08H - SPARE (PA3)
;10H - SPARE (PA4)
%EQE (DPTTIN,20H) ;20H - INPUT FOR DAT_PTT (PA5)
%EQE (CARR,40H) ;40H - CARRIER ACTIVE SENCE (PA6)
;80H - NOT USED (PA7)
```

```
; PORT B (INPUTS)
%EQE (PORTB1,0A012H)
```

```
;PORT B ADD
```

```
%EQE (ALT_C4,01H) ;01H - ALTIMETER INPUT C4 (PB0)
%EQE (ALT_C2,02H) ;02H - ALTIMETER INPUT C2 (PB1)
%EQE (ALT_C1,04H) ;04H - ALTIMETER INPUT C1 (PB2)
%EQE (ALT_B4,08H) ;08H - ALTIMETER INPUT B4 (PB3)
%EQE (ALT_B2,10H) ;10H - ALTIMETER INPUT B2 (PB4)
%EQE (ALT_B1,20H) ;20H - ALTIMETER INPUT B1 (PB5)
%EQE (ALT_A4,40H) ;40H - ALTIMETER INPUT A4 (PB6)
%EQE (ALT_A2,80H) ;80H - ALTIMETER INPUT A2 (PB7)
```

```
; PORT C (INPUTS & OUTPUTS)
%EQE (PORTC1,0A013H)
```

```
;
```

```
%EQE (RADIO0,0A051H) ;LOW RADIO ON
%EQE (PTT_TK,0A052H) ;LOW PUSH TO TALK OUTPUT TO RADIO
```

```
;*****Loran Receiver*****
;
;CPU USART
```

*****LAP TOP COMPUTER SERIAL CONTROL***

```

%EQE (LAPSTS,0A061H) ;MODE/COMMAND/STATUS
%EQE (LAPDAT,0A060H) ;TRANSMIT AND RECEIVE DATA

%EQE (LAPMIF,0CEH) ;SET MODE INSTRUCTION FORMAT
;D7=1,D6=1 2 STOP BITS
;D5=0 ODD PARITY
;D4=0 PARITY DISABLED
;D3=1,D2=1 8 DATA BITS
;D1=1,D0=0 16X BAUD RATE FACTOR

%EQE (LAPTRS,035H) ;ENABLE TRANSMIT AND RECEIVE

;D5=1 SET RTS
;D4=1 RESET ERROR FLAGS
;D2=1 SET RECEIVE ENABLE
;D0=1 SET TRANSMIT ENABLE

%EQE (LAPRCS,034H) ;ENABLE RECEIVE (NOT TRANSMIT)
;D5=1 SET RTS
;D4=1 RESET ERROR FLAGS
;D2=1 SET RECEIVE ENABLE
;D0=0 DISABLE TRANSMIT

%EQE (LAPCMP,0C5H) ;STATUS COMPARISON

%EQE (LAPRVL,0CH) ;RECEIVE INTERRUPT COUNT MAX
%EQE (LAPTVL,0CH) ;TRANSMIT INTERRUPT COUNT MAX
%EQE (LAPRET,06H) ;USART RESET TRIES

```

*****RADIO COMMUNICATION CONTROLLER*****

```

%EQE (RADSTS,0A031H) ;MODE/COMMAND/STATUS
%EQE (RADDAT,0A030H) ;TRANSMIT AND RECEIVE DATA
%EQE (RADMIF,0CEH) ;SET MODE INSTRUCTION FORMAT
;D7=1,D6=1 2 STOP BITS
;D5=0 ODD PARITY
;D4=0 PARITY DISABLED
;D3=1,D2=1 8 DATA BITS
;D1=1,D0=0 16X BAUD RATE FACTOR

%EQE (RADTRS,031H) ;ENABLE TRANSMIT
;D5=1 SET RTS
;D4=1 RESET ERROR FLAGS
;D0=1 SET TRANSMIT ENABLE

%EQE (RADRCS,034H) ;ENABLE RECEIVE (NOT TRANSMIT)
;D5=1 SET RTS
;D4=1 RESET ERROR FLAGS
;D2=1 SET RECEIVE ENABLE
;D0=0 DISABLE TRANSMIT

%EQE (RADCOM,0C5H) ;STATUS COMPARISON

%EQE (RADRES,50H) ;RADIO RESET

%EQE (RADRVL,0CH) ;RECEIVE INTERRUPT COUNT MAX
%EQE (RADTVL,0CH) ;TRANSMIT INTERRUPT COUNT MAX
%EQE (RADSVL,03H) ;SYNCH INTERRUPT COUNT MAX
%EQE (RADRET,06H) ;USART RESET TRIES

```

*****PRINTER COMMUNICATION CONTROLLER*****

```

%EQE (PRTSTS,0A071H) ;MODE/COMMAND/STATUS
%EQE (PRTDAT,0A070H) ;TRANSMIT AND RECEIVE DATA

%EQE (PKTMIF,0CEH) ;SET MODE INSTRUCTION FORMAT
;D7=0,D6=1 1 STOP BIT
;D5=0 ODD PARITY
;D4=0 PARITY DISABLED
;D3=1,D2=1 8 DATA BITS
;D1=1,D0=0 16X BAUD RATE FACTOR

```

```

%EQE (PRTRRS,11H) ;ENABLE TRANSMIT
;D5=0 SET RTS
;D4=1 RESET ERROR FLAGS
;D2=0 SET RECEIVE ENABLE
;D0=1 SET TRANSMIT ENABLE

%EQE (PRTCMP,0C5H) ;STATUS COMPARISON

%EQE (PRTTVL,0CH) ;TRANSMIT INTERUPT COUNT MAX

;*****TERMINAL*****

%EQE (TERSTS,0A061H) ;MODE/COMMAND/STATUS
%EQE (TERDAT,0A060H) ;TRANSMIT AND RECEIVE DATA
%EQE (TERMIF,0CH) ;SET MODE INSTRUCTION FORMAT
;D7=0,D6=1 1 STOP BIT
;D5=0 ODD PARITY
;D4=0 PARITY DISABLED
;D3=1,D2=1 8 DATA BITS
;D1=1,D0=0 16X BAUD RATE FACTOR

%EQE (TERTRS,50H) ;ENABLE TRANSMIT AND RECEIVE
;D5=1 SET RTS
;D4=1 RESET ERROR FLAGS
;D2=1 SET RECEIVE ENABLE
;D0=1 SET TRANSMIT ENABLE

%EQE (TERRCS,50H) ;ENABLE RECEIVE (NOT TRANSMIT)
;D5=1 SET RTS
;D4=1 RESET ERROR FLAGS
;D2=1 SET RECEIVE ENABLE
;D0=0 DISABLE TRANSMIT

%EQE (TERCOM,0DFH) ;STATUS COMPARISON
%EQE (TERRVL,0CH) ;RECEIVE INTERRUPT COUNT MAX
%EQE (TERTVL,0CH) ;TRANSMIT INTERUPT COUNT MAX

;*****USART EQUATES*****
; READ

%EQE (RXRDY,02H) ;READY FOR RECEIVE
%EQE (TXRDY,01H) ;READY FOR TRANSMIT
%EQE (TXEMPT,04H) ;TRANSMITTER EMPTY
%EQE (TRSERR,38H) ;FRAMMING, OVERRUN OR PARITY ERROR

; WRITE

%EQE (TRSDIS,10H) ;DISABLE USART & ERROR RESET
%EQE (TRSRES,50H) ;RESET USART & ERROR RESET
%EQE (TRANON,011H) ;ENABLE TRANSMIT AND RESET ERROR
%EQE (TRANOF,0EEH) ;DISABLE TRANSMIT
%EQE (RECVON,014H) ;ENABLE RECEIVE AND RESET ERROR
%EQE (RECVOF,0EBH) ;DISABLE RECEIVE

;*****WATCHDOG EQUATES*****

%EQE (WDTIM,0F000H) ;WATCHDOG TIMER ADDRESS

%EQE (VWDHTC,225) ;MAX WATCH DOG RESETS OUT OF LOOP
%EQE (VRSTCT,5) ;MAX RESETS
%EQE (VWDTM,01) ;VALUE FOR WATCHDOG TIMER

;*****RAM AND ROM LIMITS*****

%EQE (RAMBG,80H) ;MSB OF BEGINING OF RAM
%EQE (RAMED,90H) ;MSB OF END OF RAM CHECK
%EQE (ROMBG,00H) ;MSB OF BEGINNING OF ROM CHECK
%EQE (ROMED,40H) ;MSB OF END OF RAM CHECK

;*****Message Length Deffinition*****

%EQE (MSGTOT,35) ;TOTAL LENGTH OF MESSAGE
;NUMBER OF SPACES IN LAPTOP MESSAGE
%EQE (LATSPC,MSGTOT-PLTLEN-PLNLEN-AIRIDN-ALTLEN-TLORST-6)
%EQE (RSTSPC,MSGTOT-AIRIDN-7) ;NUMBER OF SPACES IN RESET MESSAGE

```

```

*****CPU INTERNAL TIMER & INTERRUPT SETUP *****
;
;   TIMERS 0 & 1
%EQE   (VTMOD,21H)           ;VALUE FOR TMOD
;                                     ;TIMER 0 = 16 BIT AUTO RELOAD TIMER
;                                     ;TIMER 1 = 8 BIT AUTO RELOAD
;                                     ;   BAUD RATE GEN.

%EQE   (VTH1,0F3H)           ;RELOAD VALUE FOR TIMER 1
;                                     ;CPU USART FOR 1200/2400 BAUD
%EQE   (VTLO,00H)           ;RELOAD VALUE FOR TIMER 0 LSB
%EQE   (VTH0,0FH)           ;RELOAD VALUE FOR TIMER 0 MSB
;                                     ;TIMER INTERRUPT 4/SECOND
%EQE   (PCON,87H)           ;PCON IN INTERNAL MEMORY

*****OTHER COMMONLY USED SYMBOLS*****
%EQE   (SETLOW,0H)           ;SET OUTPUT LOW
%EQE   (SETHGH,01H)         ;SET OUTPUT HIGH
%EQE   (SC1,0A5H)           ;FIRST SYNC CHARACTER
%EQE   (SC2,0D2H)           ;SECOND SYSC CHARACTER
%EQE   (CR,0DH)             ;CARRIAGE RETURN
%EQE   (LF,0AH)             ;LINE FEED
%EQE   (TRUE,0FFH)          ;POSITIVE
%EQE   (FALSE,00)           ;NEGATIVE
%EQE   (STBRK,5BH)          ;START BRACKET [
%EQE   (ENDBRK,5DH)         ;END BRACKET OF ]
%EQE   (RERUNV,'1')         ;RERUN VALUE
VRERUN EQU 02               ;RERUN TIMER

$EJECT

;
; *****
; *****
; * * * * *
; * * * * *
; * * * * *
; * * * * *
; * * * * *
; *****
; *****

RSEG   PROG_S

;Beginning of Main Program

EXECLP:

      MOV     SP,#30H         ;SET STACK POINTER
      MOV     IE,#0          ; CLEAR INTERRUPTS

EXE1:  %CALLIT (INITAL)       ;Initialization Routine
      %CALLIT (RADHAN)       ;RADIO Handler
      %CALLIT (LAPHAN)       ;LAP TOP Handler
      %CALLIT (LORHAN)       ;Loran Receiver Handler
      %CALLIT (ALTHAN)       ;Altimeter Handler

      CALL    WDRST          ;GO RESET WATCH DOG TIMER

      MOV     DPTR,#NWDRST   ; NO WATCH DOG RESET
      MOVX   A,@DPTR
      JNZ    EXECLP          ; CONTINUE

      ; RERUN IF NEEDED

      MOV     DPTR,#RERUNA    ; NEED TO RERUN
      MOVX   A,@DPTR
      CJNE   A,#RERUNV,EXECL1;
      JMP    EXE1             ;Continue to Loop

EXECL1: CJNE   A,#TRUE,EXECL2 ; RERUN INITIALIZED
      JMP    EXECL4

EXECL2: MOV     A,#TRUE
      MOVX   @DPTR,A         ;SET RERUN INITIALIZED

```



```

EXECL3:      MOV      DPTR,#TRERUN    ; SET RERUN TIMER
              MOV      A,#VRERUN
              MOVX     @DPTR,A

EXECL4:      MOV      DPTR,#TRERUN    ; RERUN TIMER EXPIRED?
              MOVX     A,@DPTR
              JNZ      EXE1

EXECL5:      MOV      DPTR,#RERUNA
              MOV      A,#RERUNV
              MOVX     @DPTR,A

              JMP      EXECLP          ;LOOP

```

```

;          *****
;          *****
;          * * * * *
;          * *   S U B R O U T I N E S   * *
;          * * * * *
;          *****
;          *****

```

```

;THE FOLLOWING SUBROUTINE WAITS 20 MS AFTER A WRITE HAS BEEN
; MADE TO AN EEPROM

```

```

WAITEP:     PUSH     1EH
              PUSH     1FH          ;SAVE R6 & R7

              MOV      R7,15

WATEP1:     MOV      R6,0FFH

WATEP2:     NOP
              DJNZ    R6,WATEP2
              DJNZ    R7,WATEP1

              POP      1FH
              POP      1EH

              RET          ;EXIT WAIT SUBROUTINE

```

```

;THE FOLLOWING SUBROUTINE RESETS THE WATCHDOG TIMER
; IF NOT INTERRUPT BOUND

```

```

WDRST:
CLR         A          ; CLEAR COUNT

MOV         DPTR,#RADRCT ;CLEAR RADIO RECEIVE INTERRUPT
MOVX        @DPTR,A

MOV         DPTR,#RADTCT ;CLEAR RADIO TRANSMIT INTERRUPT
MOVX        @DPTR,A

MOV         DPTR,#LAPRCT  ;CLEAR LAP TOP RECEIVE INTERRUPT
MOVX        @DPTR,A

MOV         DPTR,#LAPTCT  ;CLEAR LAP TOP TRANSMIT INTERRUPT
MOVX        @DPTR,A

MOV         DPTR,#PRTTCT  ;CLEAR PRINTER TRANSMIT INTERRUPT
MOVX        @DPTR,A

MOV         DPTR,#TWTDTM  ;WATCH DOG TIMER EXP
MOVX        A,@DPTR
JNZ        WDRS4        ; NO CONTINUE

```

```

MOV      DPTR,#PORTA1    ;DATPTT LOW
MOVX     A,@DPTR        ;
ANL      A,#DPTTIN      ;
JNZ      WTDRS4         ; YES DON'T SET WATCHDOG

MOV      DPTR,#TWTDTM    ;GET WATCH DOG TIMER
MOV      A,#VWTDTM      ; YES RESET TIMER
MOVX     @DPTR,A        ;

MOV      DPTR,#WTDTIM    ;RESET WATCHDOG CHIP
MOVX     A,@DPTR

WTDRS4:  RET

END

```

\$TITLE (TIMER 8052 BASED)

NAME TIMER ;DDECREMENTS TIMERS EVERY 1/4 SEC

;ACAS ORIGINAL RELEASE 23 JAN 90 15:53

;LAST CHANGE 15 MAY 1990 8:48

;Basic Description of Module

;This module consists of two groups of timers:

;GROUP 1

; This group has timers from TSTAR1 to TEND1. They are decremented
; until they are zero. They are decremented every 1/4 second
; which is determined by the initialization of the clock
; calendar chip using the PIE output.

;GROUP 2

; This group has timers from TSTAR2 to TEND2. They are decremented
; until they are zero. They are decremented every 1 second
; which is determined by counting four ticks or decrements of the
; timers in group 1.

;GROUP 3

; This group has timers from TSTAR3 to TEND3. They are decremented
; until they are zero. They are decremented every 1 minute
; which is determined by counting sixty ticks or decrements of the
; timers in group 2.

```

; *****
; *****
; * * * * *
; * * PUBLICS * *
; * * & * *
; * * EXTERNALS * *
; * * * * *
; *****
; *****
;
;

```

PUBLIC TIMIN ;Input to this module

EXTRN NUMBER (VTH0,VTLO) ;TIMER 0 RELOAD VALUES

```

; *****
; *****
; * * * * *

```

```

;          * *          M A C R O S          * *
;          * *          * *
;          *****
;          *****

; DEFINE AND MAKE PUBLIC
;
%DEFINE (DSE (TLABEL))
PUBLIC (%TLABEL
%TLABEL: DS 1
)

$EJECT
;
;          *****
;          *****
;          * *          * *
;          * *          D A T A          * *
;          * *          * *
;          *****
;          *****

;THIS IS DATA USED INTERNAL TO THIS FILE ONLY

DATA_AREA      segment XDATA
RSEG           DATA_AREA

; GROUP 1 DECREMENTED EVERY 1/4 SECOND

GROUP1:        DS 1 ;GROUP 1 COUNTER (SET BY VGRUP1)
VGRUP1         EQU 1 ;GROUP 1 COUNTER VALUE

%DSE (TSTAR1) ;START OF TIMERS GROUP 1 (SPARE)
%DSE (TWTDTM) ;WATCH DOG TIMER COUNTER
%DSE (TLPRT0) ;LAP TOP RECEIVE TIME OUT COUNTER
%DSE (TLPART) ;LAP TOP UART REFRESH TIMER
%DSE (TLRRT0) ;LORAN RECEIVE TIME OUT COUNTER

%DSE (TWATCH) ; RADIO WATCHDOG TIMER
%DSE (TRCVE) ; RADIO RECEIVE TIME OUT TIMER
%DSE (TWATXT) ; RADIO WAIT FOR TRANSMIT
%DSE (TRDRNT) ; RADIO RECEIVE HANG UP TIMER
%DSE (TRERUN) ; RERUN TIMER
%DSE (TRCLED) ; RECEIVE ON LED TIMER (LAPHAN)

%DSE (TEND1) ;END OF TIMERS GROUP 1 (SPARE)

; GROUP 2 DECREMENTED EVERY SECOND

%DSE (GROUP2) ;GROUP 2 COUNTER (SET BY VGRUP2)
VGRUP2 EQU 3 ;GROUP 2 COUNTER VALUE

%DSE (TSTAR2) ;START OF TIMERS GROUP 2 (SPARE)
%DSE (TMAXTX) ;RADIO MAX. WAIT TO TRANSMIT
%DSE (TTRCNT) ;WAIT TO TRANSMIT TO RADIO
%DSE (TLRCNT) ;WAIT TO SEND MESSAGE TO LAPTOP
%DSE (TEND2) ;END OF TIMERS GROUP 2 (SPARE)

; GROUP 3 DECREMENTED EVERY MINUTE

%DSE (GROUP3) ;GROUP 3 COUNTER (SET BY VGRUP3)
VGRUP3 EQU 59 ;GROUP 3 COUNTER VALUE

%DSE (TSTAR3) ;START OF TIMERS GROUP 3 (SPARE)
%DSE (TEND3) ;END OF TIMERS GROUP 3 (SPARE)

$EJECT

```

```

; *****
; *****
; * * * * *
; * *          P R O G R A M          * *
; * * * * *
; *****
; *****
;
;
PROG_S      segment          CODE
RSEG PROG_S

TIMIN:
    ORL      PSW,#18H        ;SET RB
    PUSH     PSW              ;SAVE PSW
    PUSH     DPH
    PUSH     DPL
    PUSH     ACC
    PUSH     1DH              ; (REG5)
    PUSH     1CH              ; (REG4)

;GROUP 1 FAST TIMERS 1/4 SECOND / TICK

    MOV      DPTR,#GROUP1    ;GET GROUP 1 COUNTER
    MOVX     A,@DPTR        ;
    JZ       TIMG1S          ;COUNTED DOWN GO DEC.
    DEC      A                ; NOT DONE DEC AND CONT.
    MOVX     @DPTR,A        ;
    JMP      TIMG3E          ;

TIMG1S:
    MOV      A,#VGRUP1        ;RESET GROUP 1 COUNTER
    MOVX     @DPTR,A        ;

    MOV      DPTR,#TEND1+1    ;GET LAST VALUE +1
    MOV      R5,DPL          ; TIMER
    MOV      R4,DPH          ;

TIMG11:
    MOV      DPTR,#TSTAR1    ;GET FIRST TIMER
    MOVX     A,@DPTR        ;GET VALUE
    JZ       TIMG12          ; CONTINUE IF ZERO
    DEC      A                ;
    MOVX     @DPTR,A        ;

TIMG12:
    INC      DPTR            ;GO TO NEXT ONE
    MOV      A,DPH          ;SEE IF DONE
    CJNE    A,1CH,TIMG11    ; CONTINUE IF NOT EQUAL
    MOV      A,DPL          ;
    CJNE    A,1DH,TIMG11    ;

;GROUP 2 SLOW TIMERS 1 SECOND / TICK

    MOV      DPTR,#GROUP2    ;GET GROUP 2 COUNTER
    MOVX     A,@DPTR        ;
    JZ       TIMG2S          ;COUNTED DOWN GO DEC.
    DEC      A                ; NOT DONE DEC AND CONT.
    MOVX     @DPTR,A        ;
    JMP      TIMG3E          ;

TIMG2S:
    MOV      A,#VGRUP2        ;START OF GROUP 2 TIMERS
    MOVX     @DPTR,A        ;RESET GROUP 2 COUNTER
    ;

    MOV      DPTR,#TEND2+1    ;GET LAST VALUE +1
    MOV      R5,DPL          ; TIMER
    MOV      R4,DPH          ;

TIMG21:
    MOV      DPTR,#TSTAR2    ;GET FIRST TIMER
    MOVX     A,@DPTR        ;GET VALUE
    JZ       TIMG22          ; CONTINUE IF ZERO
    DEC      A                ;
    MOVX     @DPTR,A        ;

```

```

TIMG22:
    INC     DPTR           ;GO TO NEXT ONE
    MOV     A,DPH         ;SEE IF DONE
    CJNE   A,1CH,TIMG21  ; CONTINUE IF NOT EQUAL
    MOV     A,DPL         ;
    CJNE   A,1DH,TIMG21  ;

;GROUP 3 SLOW TIMERS 1 MINUTE / TICK

    MOV     DPTR,#GROUP3 ;GET GROUP 3 COUNTER
    MOVX   A,@DPTR       ;
    JZ     TIMG3S        ;COUNTED DOWN GO DEC.
    DEC    A             ; NOT DONE DEC AND CONT.
    MOVX   @DPTR,A       ;
    JMP    TIMG3E        ;

TIMG3S:
    MOV     A,#VGRUP3    ;START OF GROUP 3 TIMERS
    MOVX   @DPTR,A       ;RESET GROUP 3 COUNTER
    ;

    MOV     DPTR,#TEND3+1 ;GET LAST VALUE +1
    MOV    R5,DPL        ; TIMER
    MOV    R4,DPH        ;

    MOV     DPTR,#TSTAR3 ;GET FIRST TIMER

TIMG31:
    MOVX   A,@DPTR       ;GET VALUE
    JZ     TIMG32        ; CONTINUE IF ZERO
    DEC    A             ;
    MOVX   @DPTR,A       ;

TIMG32:
    INC     DPTR           ;GO TO NEXT ONE
    MOV     A,DPH         ;SEE IF DONE
    CJNE   A,1CH,TIMG31  ; CONTINUE IF NOT EQUAL
    MOV     A,DPL         ;
    CJNE   A,1DH,TIMG31  ;

TIMG3E:
    ;END OF GROUP2

    MOV     TH0,#VTHO    ;SET TIMER 0
    MOV     TLO,#VTLO

    POP    1CH           ;RESTORE SAVED SFR, ECT.
    POP    1DH
    POP    ACC           ;
    POP    DPL           ;
    POP    DPH           ;
    POP    PSW          ;SAVED ON INTERRUPT

    RETI                ;EXIT INTERRUPT

;END OF PROGRAM
END

```

```

$TITLE (RADHAN 8052 BASED)
NAME RADHAN ;RADIO INTERUPT HANDLER
;
;VA001 ORIGINAL RELEASE
;LAST CHANGE VA004 15 MAY 90 08:14

;VA002 14 FEB 90 ADDED LORAN STATUS
;VA003 23 FEB 90 CHANGED LORAN UPDATE METHOD
;VA004 15 MAY 90 CHANGED ATTACK DELAY TO 180 MS

```

;The purpose of this module is to handle and buffer I/O
; between the RADIO module and the radio modem.

;Basic Description of Module:

```

;RADHAN: This module will interface with the RADIO by looking at a flag
;         LRTRDF and transmitting what is in the buffers for altitude, latitude
;         and longitude when the flag is set. When the message is sent out it
;         will clear the flag LRTRDF. When it has received a message it will

```

```

; put it in RDTLTB buffer and set flag RDTLTF. It will not accept
; another message until this flag is cleared. This module will have two
; inputs. One called by EXECLP and two called from interrupts, one for
; transmit and one for receive. This module will be responsible
; for MESINP with TRNINP so that the watchdog timer will not be reset
; when transmitting.
;
; This module will handle only one kind of message format which is
; MSGLEN characters long.
;
; RADHAN module will be polite and wait for the frequency to clear
; before transmitting. If it must wait longer than 10 sec., the
; message will be blurred out even if the channel is busy, If
; the channel clears before the 10 sec. timer has expired, it will
; transmit immediately, because there may be other systems that were
; waiting and begin to transmit at the same time. It will wait an
; additional period of time as determined by reading EEPROM VRNTIM
;
; RADINL This is the Initialization Routine called by Selftest
;
; RADRIN This is the receive interrupt routine
;
; RADTIN This is the transmit interrupt routine

```

```

$EJECT

```

```

; *****
; *****
; * * * * *
; * * * * *
; * * * * *
; * * * * *
; * * * * *
; *****
; *****

```

```

PUBLIC RADTIN ;Input to this module
PUBLIC RADINL ;Receive interrupt
PUBLIC RADTIN ;Transmit interrupt
PUBLIC RADINL ;Initialization
EXTRN CODE (PRTCOM) ;TEMP

```

```

$EJECT

```

```

; *****
; *****
; * * * * *
; * * * * *
; * * * * *
; * * * * *
; * * * * *
; *****
; *****

```

```

;THIS DATA IS USED BY OTHER MODULES
; AND IS DECLARED IN EXECLP
;

```

```

; RECEIVE BUFFER ADDRESSES
EXTRN XDATA (RDTLTB,MINCKS)
;

```

```

; TIMERS

```

```

EXTRN XDATA (TWATCH) ;WATCHDOG TIMER
EXTRN XDATA (TRCVE) ;RECEIVE TIME OUT TIMER
EXTRN XDATA (TWATXT) ;WAIT FOR TRANSMIT
EXTRN XDATA (TMAXTX) ;MAX. WAIT TO TRANSMIT
;BEFORE BLERT OUT
EXTRN XDATA (TRDRNT) ;RENABLE RECEIVE

```

```

; TIMER VALUES

```

```

EXTRN XDATA (VRNTIM) ;WAIT TO TRANSMIT TIMER

```

```

; INPUT AND OUTPUT

```

```

EXTRN NUMBER (SETHGH,SETLOW) ;OUTPUTS BEFORE TRANS.
EXTRN NUMBER (PTT_TK)

```

```

;OUTPUTS BEFORE TRANS.

```

```

EXTRN NUMBER (PORTAL,CARR) ;CARRIER ACTIVE SENSOR

```

```

;          USART CONTROL
EXTRN NUMBER (RADSTS,RADDAT)          ;MODEM CONTROL
EXTRN NUMBER (RADRCS,RADTRS)          ;ENABLE RECEIVE & TRANSMIT
EXTRN NUMBER (RADMIF)                 ;MODE INSTRUCTION FORMAT
EXTRN NUMBER (TRSDIS,RADRES)          ;DISABLE USART
EXTRN NUMBER (TRSERR)                 ;FRAMING, OVERRUN OR PARITY ERROR
EXTRN NUMBER (TXEMPT)                 ;TRANSMIT EMPTY

;          RECEIVE AND TRANSMIT BUFFER LENGTHS
EXTRN NUMBER (TRNMAX)                  ;# OF TIMES TO TRANSMIT
EXTRN NUMBER (MSGLEN)                  ;MESSAGE LENGTH
EXTRN NUMBER (MSGTLN)

;          SYNC CHARACTORS
EXTRN NUMBER (SC1,SC2)                 ;SC1 = 1ST SC2 = 2ND

;          LAP TOP INTERFACE

EXTRN XDATA (RDTLTF)                   ;MESSAGE READY FROM RADIO TO LAP TOP

EXTRN XDATA (AIRCID)                   ;AIRCRAFT ID IN NONVOLATILE RAM
EXTRN NUMBER (AIRIDN)                  ;LENGTH OF AIRCRAFT ID

;          LORAN RECEIVER INTERFACE

EXTRN XDATA (LRTRDF)                   ;DATA READY FOR LAP TOP COMPUTER
EXTRN XDATA (PLTBUF,PLNBUF)           ;LATTITUDE AND LONGITUDE BUFFERS
EXTRN NUMBER (PLTLEN,PLNLEN)          ;LATTITUDE AND LONGITUDE LENGTH
EXTRN XDATA (STSBUF)                   ;LORAN STATUS BUFFER
EXTRN NUMBER (STSLN)                   ;LORAN STATUS BUFFER LENGTH

;          ALTIMETER INTERFACE

EXTRN XDATA (ALTBUF)                   ;ALTIMETER DATA
EXTRN NUMBER (ALTLEN)                  ;ALTIMETER DATA LENGTH

;          SELFTEST INTERFACE

EXTRN XDATA (SLFFLG)                   ;SELFTEST FLAG
EXTRN NUMBER (NOLRRD)                  ;NO LORAN UPDATE FOR RADIO
EXTRN NUMBER (TRRDST)                  ;NOT TRANSMITTED TO RADIO
EXTRN NUMBER (TRLTST)                  ;NOT TRANSMITTED TO LAPTOP

;          TRANSMIT BUFFER AND LENGTHS

EXTRN XDATA (STTRAN)                   ;START OF TRANSMIT BUFFER
EXTRN XDATA (RADOTB)                   ;MESSAGE IN TRANSMIT BUFFER
EXTRN XDATA (EDTRAN)                   ;END OF TRANSMIT BUFFER

;THIS IS DATA USED INTERNAL TO THIS FILE ONLY
;
$EJECT

DATA_AREA      segment      XDATA
PROG_S         segment      CODE
RSEG          DATA_AREA

RCVFLG:      DS      1      ;RECEIVE FLAGS
SCIRCD      EQU      01H    ;FIRST SYNC CHARACTER FLAG
RINPFG      EQU      02H    ;RECEIVE IN PROCESS
RECIND      EQU      04H    ;RECEIVE INITIALIZED
MUTPTR:     DS      2      ;TRANSMIT MESSAGE POINTER
MINPTR:     DS      2      ;INPUT MESSAGE POINTER

;          MESSAGE IN PROCESS BUFFER

MESINP:     DS      1      ;MESSAGE INPROCESS FLAGS
RCVINP      EQU      01H    ; RECEIVE IN PROCESS
RCVINP      EQU      0FFH - RCVINP ;RCVINP/
TRNINP      EQU      02H    ; TRANSMIT IN PROCESS
TRINPN      EQU      0FFH - TRNINP ;TRINP/

VRDRNT      EQU      04      ;REENABLE RECEIVE

```

```

$EJECT
;
; *****
; *****
; * * * * *
; * * * * * M A C R O S * * * * *
; * * * * *
; *****
; *****

;JUMP (LONG JUMP) IF A> CDATA

%*DEFINE (JAG (CDATA,DEST))LOCAL LABEL(
    PUSH ACC
    CLR C
    SUBB A,%CDATA
    JZ %LABEL
    JC %LABEL
    POP ACC
    JMP %DEST
%LABEL:
    POP ACC
)

$EJECT
;
; *****
; *****
; * * * * *
; * * * * * E Q U A T E S * * * * *
; * * * * *
; *****
; *****

;
;THE FOLLOWING IS INTERNAL EQUATES ONLY
;
;Receive flags (RCVFLG) see DATA above
;
;Message Inprocess flags (MESINP) see DATA above
VWATCH EQU 05H ;WATCH DOG TIMER VALUE
VRCVE EQU 05H ;RECEIVE TIME OUT VALUE

;MISC.

XWAIT EQU 22 ;WAIT XWAIT*10 ms before transmit
XWAITS EQU 10 ;WAIT 10 MS
VMAXTX EQU 10 ;MAX WAIT BEFORE BLERT OUT
;VALUE FOR TMAXTX

$EJECT
;
; *****
; *****
; * * * * *
; * * * * * P R O G R A M * * * * *
; * * * * *
; *****
; *****

RSEG PROG_S

;This input RADHAN is polled by the main exec EXECLP and it
; checks to see if a message needs to go out and if so enables
; the appropriate interrupt and rebuffers the message and
; calls the interrupt handler to output it. If a message has
; successfully been received it sets the appropriate flags to
; Lap Top Module so it can get the message.

RADHAN:
    MOV DPTR,%RADSTS ;THROUGH TRANSMITTING
    MOVX A,@DPTR ;GET COUNTER

```



```

ANL      A, #TXEMPT
JNZ      RADINO          ; YES CONTINUE

MOV      DPTR, #TWATCH  ;WATCH DOG TIMER EXPIRED?
MOVX     A, @DPTR
JZ       RADOA          ; YES TURN OFF

JMP      REXIT          ; NO EXIT
RADINO:
MOV      DPTR, #MESINP  ;A MESSAGE IN PROCESS
RADIOA:
MOVX     A, @DPTR       ;SEE IF TX IN PROCESS
ANL      A, #TRNINP
JZ       RADO          ; NO CONTINUE

MOV      DPTR, #TWATCH  ;WATCH DOG TIMER EXPIRED?
MOVX     A, @DPTR

JZ       RADOA         ; YES TURN OFF
JMP      REXIT         ; EXIT
RADOA:
MOV      DPTR, #RADSTS  ;DISABLE TRANSMIT INTRERRUPT
MOV      A, #TRSDIS
MOVX     @DPTR, A

MOV      DPTR, # LRTRDF ; CLEAR TRANSMIT FLAG
CLR      A
MOVX     @DPTR, A
MOV      DPTR, #MESINP  ; AND MESSAGE IN PROCESS
MOVX     @DPTR, A      ; FLAGS
RADO:
MOV      DPTR, #PTT_TK  ;GET PUSH TO TALK
MOV      A, #SETHGH    ;SET HIGH
MOVX     @DPTR, A

MOV      DPTR, #TWATCH  ;GET WATCH DOG TIMER COUNTER
MOVX     A, @DPTR      ;HAS IT EXPIRED?
JNZ      RAD1          ; NO CONTINUE
MOV      A, #VWATCH    ; YES UPDATE
MOVX     @DPTR, A

RAD1:
MOV      DPTR, #RDITLF  ;MESSAGE STILL PROCESSING?
MOVX     A, @DPTR
JNZ      RECXIT        ; YES CONTINUE

MOV      DPTR, #PORTA1  ;IS CARRIER ACTIVE SENSE HIGH
MOVX     A, @DPTR
ANL      A, #CARR
JNZ      RAD1B         ;YES CAN RECEIVE
;;;;;;;;;;;;;TEMPORARY;;;;;;;;;;;;;
JMP      RAD1B
;;;;;;;;;;;;;END OF TEMPORARY;;;;;;;;;;;;;
MOV      A, #TRSDIS    ; DISABLE USART
MOV      DPTR, #RADSTS
MOVX     @DPTR, A

CLR      A
MOV      DPTR, #RCVFLG  ;CLEAR RECEIVE FLAGS
MOVX     @DPTR, A
MOV      DPTR, #TRCVE   ;RESET RECEIVE TIMER
MOVX     A, @DPTR

JMP      RAD2          ; CONTINUE
RAD1B:
MOV      DPTR, #TRCVE   ;RECEIVE TIMER TIMER OUT?
MOVX     A, @DPTR
JNZ      RECXIT        ; NO CONTINUE

MOV      DPTR, #MESINP  ;RECEIVE IN PROCESS

MOVX     A, @DPTR      ;
ANL      A, #RCVINP    ; YES GO INITIALIZE
JNZ      RAD1A        ;

```

```

MOV     DPTR,#RCVFLG      ;ALREADY INITIALIZED?
MOVX    A,@DPTR
ANL     A,#RECIND
JZ      RAD1A

MOV     DPTR,#TRDRNT      ;SEE IF RECEIVING INTERRUPTS
MOVX    A,@DPTR
JZ      RAD1A             ; NO REINITIALIZE

JMP     RAD2              ; NO CONTINUE

; CHECK RECEIVE

RAD1A:
MOV     DPTR,#RCVFLG
MOV     A,#RECIND         ;SET RECEIVE INITIALIZED FLAG
MOVX    @DPTR,A

MOV     DPTR,#MESINP      ;CLEAR MESSAGE INPROCESS FLAG
CLR     A
MOVX    @DPTR,A

;ENABLE INTERRUPT RECEIVE

CALL    RADINL            ;REINITIALIZE

MOV     DPTR,#RADSTS      ;ENABLE RECEIVE
MOV     A,#RADRCS
MOVX    @DPTR,A

MOV     DPTR,#TRDRNT      ;RESET RECEIVE TIMER
MOV     A,#VRDRNT
MOVX    @DPTR,A

MOV     DPTR,#RDTLTB      ;SET POINTER TO BEG.
MOV     R4,DPL            ; OF MESSAGE INPUT BUFFER
MOV     A,DPH
MOV     DPTR,#MINPTR
MOVX    @DPTR,A
MOV     A,R4
INC     DPTR
MOVX    @DPTR,A

REXIT:
JMP     REXIT             ;EXIT MODULE
;END OF RECEIVE

;BEGINNING OF TRANSMIT

; SEE IF MESSAGE IS READY TO GO OUT

RAD2:
MOV     DPTR,#LRTRDF      ;MESSAGE READY TO GO OUT
MOVX    A,@DPTR
JZ      REXIT            ; NO CONTINUE

MOV     DPTR,#RCVFLG      ;IS RECEIVE IN PROCESS
MOVX    A,@DPTR          ;LOOK AT 1ST SYNC CHAR RECEIVED
ANL     A,#SC1RCD
JNZ     REXIT            ; YES DO NOT TRANSMIT

MOV     DPTR,#PORTA1      ;IS CARRIER ACTIVE SENSE HIGH
MOVX    A,@DPTR
ANL     A,#CARR
JZ      RAD2B            ;NO GO TRANSMIT IF TIMER EXPIRED?
;:::::::::::::TEMPORARY:::::::::::::
JMP     RAD2B
;:::::::::::::END OF TEMPORARY:::::::::::::

MOV     DPTR,#TWATXT      ; IS WAIT TIMER SET
MOVX    A,@DPTR
JNZ     RAD2C            ; YES DON'T RESET MAX. WAIT TIMER

```

```

MOV     DPTR,#TMAXTX      ;RESET MAX WAIT TIMER
MOV     A,#VMAXTX
MOVX    @DPTR,A
RAD2C:  MOV     DPTR,#TMAXTX      ;MAX WAIT TIMER EXPIRED?
MOVX    A,@DPTR
JZ      RAD4

MOV     DPTR,#VRNTIM      ;GET ENTERED VALUE FROM EEPROM
MOVX    A,@DPTR
MOV     DPTR,#TWATXT      ;SET WAIT TIMER
MOVX    @DPTR,A
JMP     RECXIT            ;CONTINUE
RAD2B:  MOV     DPTR,#TWATXT      ;WAIT TIMER EXPIRED?
MOVX    A,@DPTR
JNZ     RECXIT            ; NO EXIT

;      SET UP TO TRANSMIT

RAD4:   CALL    LOADMG          ;LOAD MESSAGE IN RADOTB
;      AND ENABLE INTERRUPTS

MOV     DPTR,#RCVFLG      ;CLEAR RECEIVE FLAGS (ALL OF THEM)
CLR     A
MOVX    @DPTR,A

MOV     DPTR,#TWATCH      ;GET WATCH DOG TIMER COUNTER
MOV     A,#VWATCH        ; RESET
MOVX    @DPTR,A

;ASSERT PUSH TO TALK AND TRANSMIT SWITCH

MOV     DPTR,#PTT_TK      ;GET PUSH TO TALK
MOV     A,#SETLOW
MOVX    @DPTR,A          ;SET LOWI

MOV     DPTR,#RADOTB      ;GET START ADDRESS
MOV     R3,DPL            ;SAVE LSD'S

MOV     DPTR,#STTRAN      ;GET START OF TRANSMIT BUFFER

RAD3:   ;PUT SYNC CHARACTERS IN BUFFER

MOV     A,#SC1            ;GET FIRST SYNCH CHARACTER
MOVX    @DPTR,A          ;SAVE
CALL    DINSERT           ;INSERT IN ALL MESSAGES
INC     DPTR              ;INC.
MOV     A,#SC2            ;GET SECOND SYNCH CHARACTER
MOVX    @DPTR,A          ;SAVE
CALL    DINSERT           ;INSERT IN ALL MESSAGES
INC     DPTR              ;INC.
MOV     A,R3              ;ARE WE DONE
CJNE   A,DPL,RAD3        ; NO JUMP

;COPY MESSAGE
RAD5:   MOV     R5,#MSGLEN      ;GET JUST LENGTH OF MESSAGE

RAD6:   MOVX    A,@DPTR        ; GET DATA
INC     DPTR
CALL    DINSERT           ;INSERT DATA
DJNZ   R5,RAD6           ;TILL END OF MESSAGE

;COMPUTE AND INSERT CHECKSUM

MOV     R5,#MSGLEN        ;GET TOTAL LENGTH OF MESSAGE

MOV     DPTR,#RADOTB      ;GET START OF MESSAGE
CALL    CHKADD            ;ADD CHECKSUM TO ACC
MOVX    @DPTR,A          ;PUT IN MESSAGE BUFFER
CALL    DINSERT           ;INSERT DATA

```

```
;SET OUTPUT POINTER
```

```
MOV    DPTR,#STTRAN    ;GET START OF TRANSMIT BUFFER
MOV    R4,DPL
MOV    A,DPH
MOV    DPTR,#MUTPTR    ;GET MESSAGE POINTER
MOVX   @DPTR,A         ;SAVE MSB
MOV    A,R4            ;SAVE LSB
INC    DPTR
MOVX   @DPTR,A
;
```

```
;ENABLE INTERRUPT TRANSMIT
```

```
MOV    R6,#XWAIT
RAD3L: MOV    R3,#XWAITS    ;WAIT ABOUT XWAIT * 10 MS
RAD2L: MOV    R4,#9BH
RAD2LL: NOP
      DJNZ   R4,RAD2LL
      DJNZ   R3,RAD2L
      DJNZ   R6,RAD3L
```

```
MOV    DPTR,#MESINP    ;SET MESSAGE IN PROCESS
MOV    A,#TRNINP       ;TRANSMIT
MOVX   @DPTR,A
;
```

```
CLR    EA              ;DISABLE ALL INTERRUPTS
```

```
CALL   RADINL         ;REINITIALIZE
```

```
MOV    DPTR,#RADSTS    ;ENABLE RADIO TRANSMIT
MOV    A,#RADTRS
MOVX   @DPTR,A
```

```
SETB   EA              ;ENABLE INTERRUPTS
```

```
REXIT:
```

```
RET                                ;END OF MAIN PROGRAM
```

```
;END OF MAIN PROGRAM
```

```
$EJECT
```

```
;
; *****
; *****
; * * * * * I N I T I A I Z A T I O N * * * * *
; * * * * *
; *****
; *****
```

```
;The following is for the initialization of the Synchronous
; USART
```

```
RADINL:
```

```
;RESET FOR MODE INSTRUCTION FORMAT
```

```
MOV    DPTR,#RADSTS    ;GET STATUS REGISTER
MOV    A,#RADRES       ;RESET FOR MODE INSTRUCTION
MOVX   @DPTR,A
```

```
MOV    R7,#3FH
DJNZ   R7,$            ;WAIT A BIT
```

```
; MODE INSTRUCTION FORMAT
```

```
MOV    A,#RADMIF       ;SET MODE INSTRUCTION FORMAT
MOVX   @DPTR,A
```

```
MOV    R7,#3FH
DJNZ   R7,$            ;WAIT A BIT
```

```
RET                                ;EXIT TEST
```

```

;
$EJECT

```

```

*****
*****
* *
* *   S U B R O U T I N E S   * *
* *
*****
*****

```

```

;The following subroutine loads the message in the buffer RADOTB
; from the altimeter, Loran Receiver and the nonvolatile EEPROM

```

```
LOADMG:
```

```

MOV     DPTR,#RADOTB      ;GET START OF BUFFER
MOV     R3,DPH            ; SAVE DPTR
MOV     R4,DPL

MOV     DPTR,#AIRCID     ;GET AIRCRAFT ID LOCATION
MOV     R7,#AIRIDN       ;GET NUMBER OF DIGITS IN AIRCRAFT ID
CALL    R7DOUT           ;OUTPUT THE AIRCRAFT ID.

MOV     DPTR,#PLTBUF     ;GET THE LATITUDE
MOV     R7,#PLTLEN       ;GET LENGTH FOR LATITUDE
CALL    R7DOUT           ;OUTPUT THE LATITUDE

MOV     DPTR,#PLNBUF     ;GET THE LONGITUDE
MOV     R7,#PLNLEN       ;GET LENGTH FOR LONGITUDE
CALL    R7DOUT           ;OUTPUT THE LONGITUDE

MOV     DPTR,#ALTBUF     ;GET ALTITUDE BUFFER
MOV     R7,#ALTLEN       ;GET ALTITUDE LENGTH
CALL    R7DOUT           ;OUTPUT THE ALTITUDE

MOV     DPTR,#SLFPLG     ;GET SELFTEST FLAG
MOVX   A,@DPTR
CALL    HEXOUT           ; OUTPUT

MOVX   A,@DPTR           ;GET SELFTEST FLAGS
ORL    A,#NOLRRD         ;SET NO LORAN UPDATE
ORL    A,#TRLTST         ;SET NO MESSAGE TO LAP TOP
MOV    R7,A              ;SAVE

MOV     A,#TRRDST        ;CLEAR MSG. SENT RADIO
CPL    A
ANL    A,1FH
MOVX   @DPTR,A

MOV     DPTR,#STSBUF     ;GET STATUS BUFFER
MOV     R7,#STSLEN       ;GET LENGTH OF BUFFER
CALL    LORCON           ;OUTPUT LORAN STATUS

RET                                           ;EXIT SUBROUTINE

```

```

;The following subroutine converts the data in the buffer STSBUF of
; length STSLEN and puts the hex characters in the Transmit buffer.

```

```

LORCON:
MOV     R1,#4            ;GET NUMBER OF BITS TO GO OUT
MOV     R2,#0            ; CLEAR MEMORY

LORCN1:
MOVX   A,@DPTR           ;GET CHARACTER
INC    DPTR
CJNE   A,#2DH,LORCN2    ;IS FLAG SET
CLR    C                 ; NO CLEAR
JMP    LORC2A

LORCN2:
SETB   C

LORC2A:
MOV    A,R2
RLC    A
MOV    R2,A

```

```

        DJNZ R7,LORCN3      ; WHEN DONE EXIT
        DEC R1
        JMP LORCN4
LORCN3: DJNZ R1,LORCN1      ; CONTINUE TIL BYTE IS SET

        CALL BINOUT        ;OUTPUT WHEN FULL
        JMP LORCON         ; AND CONTINUE
LORCN4: CLR C
        RLC A
        DJNZ R1,LORCN4     ; CONTINUE TIL BYTE IS SET
        CALL BINOUT        ;OUTPUT FULL

        RET                ;EXIT SUBROUTINE

```

```

; The following subroutine outputs data at dptr for the number
; of times found in R7.

```

```

R7DOUT: MOVX A,@DPTR        ;GET ID NUMBER
        CALL RADBLD       ; OUTPUT
        INC DPTR
        DJNZ R7,R7DOUT    ; UNTIL DONE

        RET                ;EXIT SUBROUTINE

```

```

;The following subroutine puts the data in the ACC when called
; in the RADOTB buffer pointed to by the pointer found in RADOPT

```

```

RADBLD: PUSH DPL          ;SAVE DPTR
        PUSH DPH

        MOV DPH,R3        ;GET NEW DPTR FOR RADOTB
        MOV DPL,R4
        MOVX @DPTR,A
        INC DPTR
        MOV R3,DPH
        MOV R4,DPL

        POP DPH           ;RESTORE DATA POINTER
        POP DPL

        RET                ;EXIT SUBROUTINE

```

```

;THE FOLLOWING SUBROUTINE CONVERTS THE DATA IN THE ACC TO ASCII AND
; OUTPUTS IT

```

```

HEXOUT: PUSH ACC

        SWAP A            ;GET MSB
        CALL BINOUT      ;OUTPUT IT

        POP ACC
        CALL BINOUT      ;GET LSB
                          ; AND OUTPUT IT

        RET

```

```

;This subroutine take the ACC and converts to ASCII and outputs it
; to the console

```

```

BINOUT: ANL A,#0FH        ;STRIP OFF MSB

        JAG (9,BINOT1)   ;GREATER THAT 9

        ORL A,#30H       ;MAKE ASCII
        JMP BINOT2       ;CONTINUE

```

```

BINOT1:
      ADD      A,#37H          ;MAKE ASCII
BINOT2:
      CALL    RADBLD          ;OUTPUT TO BUFFER
      RET      ;EXIT SUBROUTINE

```

```

;THE FOLLOWING SUBROUTINE ADDS UP THE DATA STARTING AT DPTR FOR
; THE CHECKSUM CALCULATION.

```

```

CHKADD:
      PUSH    1FH             ;SAVE R7
      PUSH    1EH             ;SAVE R6

      CLR     C               ;CLEAR CARRY
      CLR     A               ;CLEAR ACC
      MOV     R6,A            ;CLEAR R6
      MOV     R7,A            ;CLEAR R7
CHKLOP:
      MOVX   A,@DPTR         ;GET DATA
      INC    DPTR            ;INC POINTER
      ADDC   A,R7            ;ADD DATA TO CHECKSUM
      MOV    R7,A            ;SAVE

      JNC    CHKLP1          ;CONTINUE IF CARRY NOT SET
      MOV    R6,#OFFH        ; SET R6
CHKLP1:
      DJNZ   R5,CHKLOP      ;
      MOV    A,R6            ;SEE IF ALL ZERO'S
      ORL   A,R7             ;
      JNZ   CHKLP2          ; NO CONTINUE
      MOV    R7,#OF0H        ;CHANGE R7
CHKLP2:
      MOV    A,R7            ;GET CHECKSUM
      POP    1EH             ;GET R6
      POP    1FH             ;
      RET      ;EXIT SUBROUTINE

```

```

;THE FOLLOWING SUBROUTINE INSERTS THE DATA AT THE DIFFERENT MESSAGE

```

```

; LEVELS TO BE TRANSMITTED TRNMAX NUMBER OF TIMES.

```

```

DINSERT:
      MOV     R6,#TRNMAX      ;GET NUMBER OF TIMES TO TRANSMIT
      DJNZ   R6,DINLOP       ;NOT ZERO CONTINUE
      JMP    DINLP4          ; DO NOT INSERT ANY

      PUSH   DPL              ;SAVE DPTR
      PUSH   DPH              ;
DINLOP:
      MOV    R4,#MSGTLN      ;GET TOTAL MESSAGE LENGTH
DINLP3:
      INC    DPTR            ;GET NEXT ADDRESS
      DJNZ   R4,DINLP3       ;NOT DONE LOOP
      MOVX   @DPTR,A         ;SAVE IN OUTPUT BUFFER

      DJNZ   R6,DINLOP       ;NOT DONE LOOP
      POP    DPH              ;GET DPTR
      POP    DPL              ;
DINLP4:
      RET

```

\$EJECT

```

;
;
; *****
; *****
; * *
; * * I N T E R R U P T   H A N D L E R   * *
; * *
; *****
; *****
;
;
; TRANSMIT
; 2. Determine if an error has occurred. If so ALERT
; RADIO.
; 3. Determine if the message has completed, if so:
; A. Clear flag (LRTRDF) so that RADIO can enter
; another message.
; B. Disable interrupts.

```

```

RADTIN:
MOV     DPTR,#MESINP      ;TRANSMIT IN PROCESS
MOVX   A,@DPTR           ;
ANL    A,#TRNINP        ;
JNZ    TRNINT            ;TRANSMIT INTERRUPT

MOVX   A,@DPTR           ;RECEIVE IN PROCESS
ANL    A,#RCVINP        ;
JNZ    RDNTX1            ; YES CONTINUE
MOV    A,#RADRCS        ;
JMP    RDNTX2            ;RESET USART

RDNTX1:
MOV    A,#RADRCS        ;SET RECEIVE MODE

RDNTX2:
MOV    DPTR,#RADSTS      ;
MOVX   @DPTR,A

MOV    DPTR,#RADDAT     ; DUMMY WRITE
MOVX   @DPTR,A

JMP    INEXAL           ;BAD INTERRUPT EXIT

TRNINT:
PUSH   1DH              ;REG 5
PUSH   1CH              ;REG 4

MOV    DPTR,#EDTRAN     ;GET END OF BUFFER
MOV    R5,DPL

MOV    DPTR,#MUTPTR     ;GET MESSAGE POINTER
MOVX   A,@DPTR         ; MSB
MOV    R4,A             ; AND SAVE
INC    DPTR             ; GET LSB
MOVX   A,@DPTR         ;
MOV    DPL,A           ;GET NEW DPTR
MOV    DPH,R4           ; FOR MESSAGE
CJNE  A,1DH,TOUT       ;MORE CONT.

;NO MORE DATA QUIT

MOV    DPTR,#RADSTS     ;DISABLE TRANSMIT INTRERRUPT
MOV    A,#TRSDIS
MOVX   @DPTR,A

MOV    DPTR,#LRTRDF    ; CLEAR TRANSMIT
CLR    A
MOVX   @DPTR,A

MOV    DPTR,#MESINP    ;CLEAR ALL MESSAGE INPROCES FLAGS
CLR    A
MOVX   @DPTR,A

JMP    INEXIT

```



```
TOUT:
MOVX  A,@DPTR          ;GET DATA
MOV   R5,A             ;SAVE
INC   DPTR             ;GO TO NEXT BYTE
MOV   R4,DPL          ;SAVE DATA POINTER
MOV   A,DPH           ; IN MESSAGE POINTER
MOV   DPTR,#MUTPTR    ;
MOVX  @DPTR,A         ;
MOV   A,R4            ;
INC   DPTR            ;
MOVX  @DPTR,A         ;
MOV   DPTR,#RADDAT    ; OUTPUT DATA
MOV   A,R5            ;
MOVX  @DPTR,A         ;
JMP   INEXIT          ; AND EXIT
```

```
;END OF TRANSMIT INTERRUPT HANDLER
```

```
;RECEIVE
```

```
; 1. After the interrupt routine RDINTS has set the sync flags
; this routine is enabled to start receiving the message.
; 2. It will read in and buffer the message.
; 3. The message will be ignored if the checksum is bad.
; 4. Determine end of message, when the message has completed
; do the following:
;   A. Set flag (RADFIN) to alert RADIO that message has
;      arrived.
;   B. Disable interrupts so that message will not be
;      overrun.
```

```
RADRIN:
```

```

PUSH  1DH              ;
PUSH  1CH              ;
PUSH  1FH              ;

MOV   DPTR,#MESINP    ; MESSAGE INPROCESS
MOVX  A,@DPTR
ANL   A,#TRNINP      ; TRANSMIT EXIT
JNZ   RDITR2         ;

MOV   DPTR,#PORTA1    ;GET CHANNEL BUSY OR CARRIER
MOVX  A,@DPTR        ; ACTIVE SENSE
ANL   A,#CARR        ;

;:::::::::::::TEMPORARY:::::::::::::
JMP   TEMP
;:::::::::::::END OF TEMPORARY:::::::::::::
JZ    RDITR1         ; NO DON'T TRY TO RECEIVE
;:::::::::::::TEMPORARY:::::::::::::
TEMP:
;:::::::::::::END OF TEMPORARY:::::::::::::
MOV   DPTR,#RADSTS    ;FRAMING OVERRUN OR
MOVX  A,@DPTR        ; PARITY ERROR
ANL   A,#TRSERR
JZ    RDINGD         ; NO GET DATA

RDITR1:
MOV   A,#RADRCS       ;REFRESH USART
MOV   DPTR,#RADSTS
MOVX  @DPTR,A
JMP   RDITR3

RDITR2:
MOV   A,#RADTRS       ;REFRESH USART
MOV   DPTR,#RADSTS
MOVX  @DPTR,A

RDITR3:
MOV   DPTR,#RADDAT    ;GET DATA
MOVX  A,@DPTR
;
JMP   INEXIR         ;ERROR GO EXIT

; GOOD DATA PROCESS
RDINGD:
MOV   DPTR,#TRDRNT    ;RESET RECEIVE TIMER
```

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```

MOV      A,#VRDRNT
MOVX     @DPTR,A

MOV      DPTR,#RADDAT      ;GET DATA
MOVX     A,@DPTR          ;
MOV      R5,A              ;SAVE

MOV      DPTR,#RCVFLG      ;GET RECEIVE FLAGS
CJNE     A,#SC1,RCVCK2    ;IS DATA 1ST SYNC CHARACTER
; YES SET 1SYNC CHARACTER REC

MOVX     A,@DPTR
ORL      A,#SC1RCD
MOVX     @DPTR,A

RCVCK1:  MOVX     A,@DPTR      ;SEE IF MESSAGE ALREADY INPROCESS
ANL      A,#RINPFG
RCVI2D  ; YES CONTINUE
JMP      INEXIR           ;NO EXIT

RCVCK2:  MOVX     A,@DPTR      ; GET FLAGS
ANL      A,#SC1RCD        ;1ST SYNC CHARACTER RECEIVED?
JZ       RCVCK1          ; NO CONTINUE

MOV      A,#SC1RCD        ;YES CLEAR 1ST REC FLAG
CPL      A
MOV      R4,A
MOVX     A,@DPTR
ANL      A,1CH
MOVX     @DPTR,A

MOV      A,R5              ; GET CHARACTER
CJNE     A,#SC2,RCVCK1    ;SECOND SYNC CHARACTER?
MOVX     A,@DPTR          ; YES

ORL      A,#RINPFG        ; SET IN PROCESS FLAG
MOVX     @DPTR,A

MOV      DPTR,#MESINP      ; SET RECEIVE MESSAGE INPROCESS
MOV      A,#RCVINP
MOVX     @DPTR,A

MOV      DPTR,#TRCVE       ;SET RECEIVE TIMER
MOV      A,#VRCVE
MOVX     @DPTR,A

JMP      RCVI3A           ;CLEAR AND EXIT

; DATA IS GOOD AND RECEIVED SYNC CHARACTORS

RCVI2D:  MOV      DPTR,#TRCVE      ;SET RECEIVE TIMER
MOV      A,#VRCVE
MOVX     @DPTR,A

MOV      DPTR,#MINPTR      ;GET MESSAGE POINTER
MOVX     A,@DPTR          ; MSB
MOV      R4,A              ;
INC      DPTR              ; LSB
MOVX     A,@DPTR          ;
MOV      DPL,A            ;AND SET IN DPTR
MOV      DPH,R4           ;
MOV      A,R5              ;
CALL     PRTCOM            ;*****TEMP
MOVX     @DPTR,A          ;SAVE DATA IN BUFFER
INC      DPTR              ;INCREMENT POINTER
MOV      A,DPL             ; SAVE POINTER
MOV      R4,DPH           ;

MOV      DPTR,#MINCKS+1    ; SEE IF AT END OF MESSAGE
CJNE     A,DPL,RCVIN4     ; NO CONTINUE

RCVI2E:  MOV      R7,#00        ;
MOV      R5,#MSGLEN       ;GET CHECKSUM MESSG LENGTH
MOV      DPTR,#RDTLTB     ;

```

```

CALL      CHKADD      ;ADD UP CHECKSUM

MOV       R5,A        ;SAVE CHECKSUM
MOV       DPTR,#MINCKS ;GET RECEIVED CHECKSUM
MOVX      A,@DPTR     ;SEE IF RIGHT

RCVI2F:   CJNE        A,1DH,RCVIN3      ; NOT RIGHT EXIT

RCVI2K:   MOV         DPTR,#RDTLTF      ;SET MESSAGE IN FLAG
MOV       A,#OFFH      ;
MOVX      @DPTR,A      ;

MOV       DPTR,#TRCVE   ;CLEAR RECEIVE TIMER
CLR       A
MOVX      @DPTR,A

RCVIN3:   MOV         DPTR,#RADSTS      ;DISABLE ALL INTERRUPTS
MOV       A,#TRSDIS
MOVX      @DPTR,A

MOV       DPTR,#RCVFLG  ;CLEAR RECEIVE FLAGS
CLR       A
MOVX      @DPTR,A

RCVI3A:   MOV         DPTR,#RDTLTB     ;SET POINTER TO BEG.
MOV       A,DPL        ; OF MESSAGE INPUT BUFFER
MOV       R4,DPH      ;

RCVIN4:   MOV         DPTR,#MINPTR     ;SAVE POINTER
MOV       R5,A        ;
MOV       A,R4        ;
MOVX      @DPTR,A    ;
MOV       A,R5        ;
INC       DPTR        ;
MOVX      @DPTR,A    ;

INEXIR:   POP         1FH

INEXIT:   POP         1CH      ;
POP       1DH      ;

INEXAL:   RET          ;EXIT INTERUPT

;
END

```

```

10 REM LORRAN SIMULATOR FEB 23 90 12:13
20 PRINT"ENTER LON DIR"
30 LN$ = INPUT$(1)
40 PRINT "ENTER LONGITUDE (7 DIGITS)"
50 INPUT N
60 PRINT "ENTER LAT DIR"
70 LT$ = INPUT$(1)
80 PRINT "ENTER LATITUDE (6 DIGITS)"
90 INPUT T
100 PRINT "ENTER CHANGE FOR LONGITUDE"
110 INPUT NC
120 PRINT "ENTER CHANGE FOR LATITUDE"
130 INPUT TC
160 OPEN "COM:" FOR OUTPUT AS#1
200 PRINT "LAT ";LT$;USING "#####"; CDBL(T)
210 PRINT "LON ";LN$;USING "#####"; CDBL(N)
220 IF T <100000 THEN PRINT#1,"LAT ";LT$;USING "0#####";CDBL(T)
230 IF T >99999 THEN PRINT#1,"LAT ";LT$;USING "#####";CDBL(T)
240 IF N <1E+06 THEN PRINT#1,"LON ";LN$;USING "0#####";CDBL(N)
250 IF N >999999 THEN PRINT#1,"LON ";LN$;USING "#####";CDBL(N)
260 PRINT#1,"WRN SD-----"
270 FOR I = 0 TO 1000

```

```

280 NEXT I
290 LET N=N+NC
300 LET T=T+TC
310 GOTO 200
320 END

```

```

$TITLE (LORHAN 8052 BASED)
NAME LORHAN

```

```
;VA001 2 Nov 89 ORIGINAL RELEASE
```

```
;LAST MODIFIED 23 FEB 90 10:51
```

```
;VA002 15 FEB 90 ADDED LORAN STATUS
```

```
;VA003 23 FEB 90 CHANGED LORAN METHOD OF UPDATE
```

```
;Basic Description of Module
```

```

;
; This module will monitor and control the transmitting of messages
; to and from the Loran Receiver. The following is a brief description
; of how this module is organized and the function of each section.
;
; LORHAN This is the main section or the part that is called from the
; Executive Loop. When the DIN flags are set (see LORRIN) it will
; rebuffer the data in the TLXBUF buffers to the PLXBUF buffers.
;
; LORRIN This is the receive interrupt handler from the Loran Receiver.
; Messages coming from the Loran Receiver will ignore spaces and
; look for the words "LAT" and "LON" and put the Data in the buffers
; TLTBUF and TLNBUF. When it is through receiving lat. and lon.
; it will set the LATDIN and LONDIN flags for LORHAN.
;
; LORINL This is the initialization routine for this module. It is
; called once by the Initialization or Selftest module, and is called
; internally by this module if the receive lock up during a lat. or
; a lon. input. This module Sets up the CPU UART and timer as well
; as clears the MESINP flags.

```

```
$EJECT
```

```

;
; *****
; *****
; * * * * *
; * * PUBLICS * * * * *
; * * * * *
; *****
; *****
;
; PUBLIC LORHAN ;INPUT TO THIS MODULE
PUBLIC LORINL ;INITIALIZATION OF MODULE
PUBLIC LORRIN ;RECEIVE INTERRUPT HANDLER

```

```
$EJECT
```

```

;
; *****
; *****
; * * * * *
; * * EXTERNALS * * * * *
; * * * * *
; *****
; *****
;
EXTRN XDATA (LRRCNT) ;RECEIVE INTERRUPT COUNT
EXTRN XDATA (TLRRT0) ;RECEIVE TIMEOUT TIMER
; LORAN RECEIVER INTERFACE
EXTRN XDATA (LRTLTF) ;Data ready for Lap Top
; RADIO INTERFACE

```

```

EXTRN XDATA (LRTRDF)           ;Data ready for radio
;      POSITION BUFFERS
EXTRN XDATA (PLTBUF,PLNBUF)    ;LONGITUDE AND LATITUDE
EXTRN XDATA (TLTBUF,TLNBUF)    ;TEMP. LONGITUDE AND LATITUDE
EXTRN NUMBER (PLTLEN,PLNLEN)   ;LENGHT OF LATITUDE AND LONGITUDE
;      STATUS BUFFER AND LENGTH
EXTRN XDATA (STSBUF)           ;STATUS BUFFER
EXTRN NUMBER (STSLEN)          ;STATUS BUFFER LENGTH
;      LORAN ACTIVE (DO NOT PROCESS)
EXTRN XDATA (TTRCNT,VTRCNT)    ;VALUE & TIMER FOR HOW LONG TO WAIT
;                                ; TO TRANSMIT TO RADIO
EXTRN XDATA (TLRCNT,VLRCNT)    ;VALUE & TIMER FOR HOW LONG TO WAIT
;                                ; TO SEND MESSAGE TO LAP TOP
;      SELFTEST INTERFACE
EXTRN XDATA (SLFFLG)           ;SELFTEST FLAG
EXTRN NUMBER (NOLRRD)          ;NO LORAN UPDATE FOR RADIO
EXTRN NUMBER (NOLRLT)          ;NO LORAN UPDATE FOR LAPTOP
;      MISC
EXTRN NUMBER (TRUE)            ;INPUT TRUE

```

```

;      *****
;      *****
;      * * * * *
;      * *      D A T A      * *
;      * * * * *
;      *****
;      *****

```

```

;THIS IS DATA USED INTERNAL TO THIS FILE ONLY

```

```

DATA_AREA      segment XDATA
PROG_S         segment CODE
RSEG           DATA_AREA

```

```

MESINP:        DS      01H      ;MESSAGE INPROCESS FLAGS
;              ; DONE FLAGS MUST BE 10H*INPROCESS FLAGS

```

```

; NOTE INPORCESS FLAGS ARE 1ST NIBBLE

```

```

LONINP        EQU      01H      ;LONGITUDE DATA COMMING IN
LATINP        EQU      02H      ;LATITUDE DATA COMMING IN
STSinP        EQU      04H      ;STATUS DATA COMMING IN

```

```

; NOTE DONE FLAGS ARE ONLY 2ND NIBBLE

```

```

LONDIN        EQU      10H*LONINP ;LONGITUDE DATA COMPLETE
LATDIN        EQU      10H*LATINP ;LATITUDE DATA COMPLETE
STSDIN        EQU      10H*STSinP  ;STATUS DATA COMPLETE

```

```

LORIPT:        DS      02H      ;LORAN RECEIVE INPUT POINTER

```

```

LORNCH:        DS      01H      ;NUMBER OF CHARACTORS TO READ

```

```

FILOLN        EQU      03H      ;LENGTH OF FILO BUFFER

```

```

FILODT:        DS      FILOLN   ;FIRST IN LAST OUT BUFFER

```

```

; *****
; *****
; * * * * *
; * *   E Q U A T E S   * *
; * * * * *
; *****
; *****

;   FOR MESINP FLAGS SEE DATA ABOVE

VLRRT0   EQU          04           ;REINITIALIZE AFTER > 1/2 SECOND

$EJECT
;

; *****
; *****
; * * * * *
; * *   T A B L E S   * *
; * * * * *
; *****
; *****

RSEG     PROG_S

; The following tables are trigger compare tables for Loran
; receiver inputs, they must be backwards from how received:

LATTRG:  DB          'TAL'
LONTRG:  DB          'NOL'
STSTRG:  DB          'NRW'

; *****
; *****
; * * * * *
; * *   P R O G R A M   * *
; * * * * *
; *****
; *****

LORHAN:

;   CHECK FOR MESSAGE FROM LORAN

MOV      DPTR,#MESINP           ;SEE IF MSG COMPL. FROM LORAN
MOVX     A,@DPTR
ANL      A,#LONDIN+LATDIN+STSDIN ;CHECK FOR BOTH LAT. LON. AND STATU
CJNE     A,#LONDIN+LATDIN+STSDIN,CHKRAD
CPL      A                       ;GOT THEM SO CLEAR
MOV      R2,A
MOVX     A,@DPTR
ANL      A,R2
MOVX     @DPTR,A                ; AND SAVE

MOV      DPTR,#SLFFLG           ;CLEAR SELFTEST FLAG
MOV      A,#NOLRLT              ; FOR LORAN
ORL      A,#NOLRRD              ; NOT RECEIVED
CPL      A
MOV      R7,A
MOVX     A,@DPTR
ANL      A,1FH
MOVX     @DPTR,A

;   REBUFFER DATA FOR REST OF MODULES
;   ;LONGITUDE

MOV      DPTR,#PLNBUF           ;GET START OF DESTINY
MOV      R0,DPL
MOV      R2,DPH
MOV      DPTR,#TLNBUF           ;GET START OF SOURCE

```

```

MOV      R7,#PLNLEN      ;GET LENGTH
CALL     TRNDAT          ;TRANSFER DATA

;LATITUDE

MOV      DPTR,#PLTBUF    ;GET START OF DESTINY
MOV      R0,DPL
MOV      R2,DPH
MOV      DPTR,#TLTBUF    ;GET START OF SOURCE
MOV      R7,#PLTLEN     ;GET LENGTH
CALL     TRNDAT          ;TRANSFER DATA

; TIME TO SEND MESSAGES TO RADIO

CHKRAD:
MOV      DPTR,#VTRCNT    ;FIRST CHECK TO SEE IF DISABLED
MOVX     A,@DPTR
MOV      R3,A           ; SAVE
JZ       CHKLAP         ; DISABLED GO CHECK LAPTOP

MOV      DPTR,#TTRCNT    ;TIMER EXPIRED?
MOVX     A,@DPTR
ANL     PSW,#18H        ; CLEAR CARRY ETC
DA       A              ; DECIMAL ADJUST
MOVX     @DPTR,A
JNZ     CHKLAP         ; NO CONTINUE
MOV      A,R3
MOVX     @DPTR,A       ; RESET TIMER

MOV      A,#TRUE
MOV      DPTR,#LRTRDF    ;SET FLAG FOR MESSAGE TO RADIO
MOVX     @DPTR,A

; SEE IF MESSAGE TO LAP TOP COMPUTER

CHKLAP:
MOV      DPTR,#VLRcnt    ;FIRST CHECK TO SEE IF DISABLED
MOVX     A,@DPTR
MOV      R3,A           ; SAVE
JZ       INPCHK        ; DISABLED GO CHECK LAPTOP

MOV      DPTR,#TLRCNT    ;TIMER EXPIRED?
MOVX     A,@DPTR
ANL     PSW,#18H        ; CLEAR CARRY ETC
DA       A              ; DECIMAL ADJUST
MOVX     @DPTR,A
JNZ     INPCHK        ; NO CONTINUE
MOV      A,R3
MOVX     @DPTR,A       ; RESET TIMER

MOV      A,#TRUE
MOV      DPTR,#LRTLTF    ;SET FLAG FOR MESSAGE TO LAP TOP
MOVX     @DPTR,A

INPCHK:
; CHECK TO SEE IF INPROCESS TO LONG

MOV      DPTR,#MESINP    ; SEE IF INPROCESS
MOVX     A,@DPTR
ANL     A,#LATINP+LONINP+STSINP ;LOOK AT LAT, LON & STATUS
JZ       LOREXT        ; NOT INPROCESS EXIT

MOV      DPTR,#TLRRTO    ;SEE IF INPROCESS TIMER HAS
MOVX     A,@DPTR        ; EXPIRED
JNZ     LOREXT        ; NO EXIT

CALL     LRINL3         ; REINITIALIZE LORAN RECEIVER
; HANDLER INPUT

LOREXT:
RET

$EJECT

```

```

; *****
; *****
; * * * * *
; * *   I N I T I A I Z A T I O N   * * *
; * * * * *
; *****
; *****

```

;This is the initialization routine for this module. It is called once by the Initialization or Selftest module, and is called internally by this module if the receive lock up during a lat. or a lon. input. This module Sets up the CPU UART and timer as well as clears the MESINP flags.

;The following is for the initialization of the serial port used for transmitting to the Computer. It is initialized, and enabled.

```

;*****
;Baud Rate - 9600
;Start Bits - 1
;Data Bits - 8
;Stop Bits - 2
;Parity - None
;*****

```

```

LORINL:      MOV     A,#'9'
             MOV     R7,#PLNLEN      ; GET LONGITUDE LENGTH
             MOV     DPTR,#PLNBUF    ; FILL WITH '9''S
LRINL1:      MOVX    @DPTR,A         ; SAVE
             INC     DPTR
             DJNZ   R7,LRINL1       ; LOOP UNTIL DONE
             MOV     R7,#PLTLEN     ; GET LATITUDE LENGTH
             MOV     DPTR,#PLTBUF    ; FILL WITH '9''S
LRINL2:      MOVX    @DPTR,A         ; SAVE
             INC     DPTR
             DJNZ   R7,LRINL2       ; LOOP UNTIL DONE
LRINL3:      MOV     DPTR,#MESINP    ; CLEAR INPROCESS FLAGS
             CLR     A
             MOVX    @DPTR,A
             RET
             ;EXIT INITIALIZATION

```

\$EJECT

```

; *****
; *****
; * * * * *
; * *   S U B R O U T I N E S   * * *
; * * * * *
; *****
; *****

```

;This subroutine transfers the data at the DPTR to the destiny in R0,R2 for the length found in R7.

```

TRNDAT:
MOVX  A,@DPTR      ;GET SOURCE
INC   DPTR         ; GO TO NEXT DPTR
MOV   R1,DPL       ;SAVE SOURCE
MOV   R3,DPH
MOV   DPL,R0       ;RETRIEVE DESTINY
MOV   DPH,R2
MOVX  @DPTR,A     ;SAVE DATA
INC   DPTR
MOV   R0,DPL      ;SAVE DESTINY
MOV   R2,DPH
MOV   DPL,R1

```



```

MOV     DPH,R3           ;RETRIEVE SOURCE
DJNZ   R7,TRNDAT       ; LOOP UNTIL DONE
RET                               ;EXIT SUBROUTINE

```

;This subroutine determines if the last R6 number of characters
; match the trigger value found at the Code address or the DPTR.
; If so Acc = 0, if not Acc = 0FFH

```

FILOCK:
  PUSH   1BH           ; SAVE R3
  PUSH   DPH           ; SAVE DPTR
  PUSH   DPL

  MOV    DPTR,#FILODT ;GET DATA FROM LAST INTERRUPTS
  MOV    R5,DPH
  MOV    R7,DPL

FILOLP:
  CLR    A
  POP    DPL
  POP    DPH           ;RESTORE DPTR
  MOVC   A,@A+DPTR    ;GET DATA FROM TABLE
  MOV    R3,A         ; SAVE IN REG 3
  INC    DPTR
  PUSH   DPH           ; SAVE DPTR
  PUSH   DPL

  MOV    DPL,R7
  MOV    DPH,R5       ;GET FILO DATA POINTER
  MOVX   A,@DPTR     ; GET FILO DATA
  INC    DPTR
  MOV    R5,DPH
  MOV    R7,DPL

  CJNE   A,1BH,FILONE ;NOT EQUAL GO EXIT

  DJNZ   R6,FILOLP
  CLR    A           ;CLEAR ACC BECAUSE IT IS EQUAL
  JMP    FILOXT     ; AND EXIT

FILONE:
  MOV    A,#0FFH    ;SET ACC NOT EQUAL

FILOXT:
  POP    DPL
  POP    DPH         ;RESTORE DPTR
  POP    1BH        ; RESTORE R3

  RET                               ;EXIT SUBROUTINE

```

\$EJECT

```

; *****
; *****
; * * * * *
; * *   INTERRUPT   HANDLER   * *
; * * * * *
; *****
; *****
;

```

; This interrupt handler handles the data coming into
; the CPU from the Loran Receiver.

```

LORRIN:
  CLR    EA           ;DISABLE INTERRUPTS
  PUSH   PSW         ;SAVE PROGRAM STATUS WORD
  PUSH   ACC         ;SAVE ACCUMULATOR
  PUSH   DPH         ;SAVE DPTR
  PUSH   DPL
  PUSH   1FH         ;SAVE R7
  PUSH   1EH         ;SAVE R6
  PUSH   1DH         ;SAVE R5
  PUSH   1CH         ;SAVE R4

```

133

```

MOV     DPTR,#LRRCNT      ;INC COUNT
MOVX   A,@DPTR           ;FOR INTERRUPTS
INC    A
MOVX   @DPTR,A

```

```

;SEE IF RECEIVE OR TRANSMIT INTERRUPT

```

```

JB     RI,LORRN1         ;CONTINUE IF RECEIVE INTERRUPT
JMP    LORRXT

```

```

LORRN1:

```

```

; READ DATA AND PUT IN FIRST IN LAST OUT BUFFER

```

```

MOV     DPTR,#FILODT     ;GET FILO 3 POS BUFFER
MOV     A,SBUF           ;GET COM PORT DATA
MOV     R5,A            ; AND SAVE FOR LATER
MOV     R7,A            ; AND SAVE FOR FILO BUFFER
MOV     R6,#FILOLN      ;GET NUMBER OF CHARACTERS IN BUFFER

```

```

FILOFL:

```

```

MOVX   A,@DPTR          ;GET DATA IN FIRST POSITION
MOV     R4,A            ; SAVE
MOV     A,R7            ;GET NEW DATA
MOVX   @DPTR,A          ;SAVE
MOV     1FH,R4          ; PUT DATA FROM R4 IN TO R7
INC    DPTR
DJNZ   R6,FILOFL        ;CONTINUE FOR WHOLE BUFFER

```

```

; CHECK TO SEE IF A MESSAGE IS ALREADY IN PROCESS

```

```

MOV     DPTR,#MESINP     ;MESSAGE IN PROCESS
MOVX   A,@DPTR
ANL    A,#LATINP+LONINP+STSINP ; CHECK LAT LON AND STATUS
JZ     LATCHK           ; NO GO CHECK TO SEE IF COMMING

```

```

; THE FOLLOWING READS IN THE DATA FOR A MESSAGE INPROCESS

```

```

MOV     A,R5             ;GET DATA, IS IT A SPACE
CJNE   A,#' ',DATSAV    ; NO GO SAVE DATA
JMP    LORRXT           ; YES EXIT

```

```

DATSAV:

```

```

MOV     DPTR,#LORIPT     ;GET DATA POINTER
MOVX   A,@DPTR          ; GET DPH
MOV     R7,A            ; SAVE
INC    DPTR             ; GET DPL
MOVX   A,@DPTR
MOV     DPL,A
MOV     DPH,R7          ;SET UP DPTR

```

```

MOV     A,R5             ;STORE DATA
MOVX   @DPTR,A

```

```

INC    DPTR             ;INC. POINTER AND STORE
MOV     R7,DPL
MOV     A,DPH
MOV     DPTR,#LORIPT    ; SAVE
MOVX   @DPTR,A
MOV     A,R7
INC    DPTR
MOVX   @DPTR,A

```

```

MOV     DPTR,#LORNCH     ;NUMBER OF CHARACTORS = 0
MOVX   A,@DPTR          ; AFTER DEC.
DEC    A
MOVX   @DPTR,A

```

```

JNZ    LORRXT           ; NO CONTINUE

```

```

MOV     DPTR,#MESINP     ;GET WHICH MESSAGE IT IS
MOVX   A,@DPTR
ANL    A,#0FOH          ;SAVE ANY DONE FLAGS
MOV     R7,A            ; IN R7

```

```

MOVX  A,@DPTR          ;GET FLAGS AGAIN
ANL   A,#0FH          ;SAVE ONLY INPROCESS FLAGS
SWAP  A
ORL   A,R7            ; GET ALL FLAGS
MOVX  @DPTR,A         ; AND SAVE

JMP   LORRXT          ; EXIT MODULE

; CHECK TO SEE IF A MESSAGE IS IN

LATCHK:
MOV   DPTR,#LATTRG    ; GET LATITUDE TRIGGER
MOV   R6,#FILOLN      ;GET NUMBER OF CHARACTERS TO COMPARE
CALL  FILOCK          ; CHECK TO SEE IF LATTITUDE
JNZ   LONCHK

; IS LATITUDE MESSAGE SET FLAGS AND EXIT

MOV   R5,#LATINP      ;GET LATITUDE IN PROCESS MSG.
MOV   R7,#PLTLEN      ;GET MESSAGE LENGTH
MOV   DPTR,#TLTBUF    ;GET POINTER FOR OUTPUT
JMP   CHKEXT          ; GO QUIT

LONCHK:
MOV   DPTR,#LONTRG    ; GET LONGITUDE TRIGGER
MOV   R6,#FILOLN      ;GET NUMBER OF CHARACTERS TO COMPARE
CALL  FILOCK          ; CHECK TO SEE IF LATTITUDE
JNZ   STSCHK          ; NO GO CHECK STATUS

; IS LONGITUDE MESSAGE SET FLAGS AND EXIT

MOV   R5,#LONINP      ;GET LONGITUDE IN PROCESS MSG.
MOV   R7,#PLNLEN      ;GET MESSAGE LENGTH
MOV   DPTR,#TLNBUF    ;GET POINTER FOR OUTPUT
JMP   CHKEXT          ; GO QUIT

STSCHK:
MOV   DPTR,#STSTRG    ; GET STATUS TRIGGER
MOV   R6,#FILOLN      ;GET NUMBER OF CHARACTERS TO COMPARE
CALL  FILOCK          ; CHECK TO SEE IF STATUS
JNZ   LORRXT          ; NO GO EXIT

; IF STATUS GO SET FLAGS AND EXIT

MOV   R5,#STSINP      ;GET STATUS IN PROCESS MSG.
MOV   R7,#STSLEN      ;GET MESSAGE LENGTH
MOV   DPTR,#STSBUFF   ;GET POINTER FOR OUTPUT

CHKEXT:
MOV   R6,DPL          ;SAVE DPTR
MOV   A,DPH
MOV   DPTR,#LORIPT    ; IN LORAN INPUT POINTER
MOVX  @DPTR,A
INC   DPTR
MOV   A,R6
MOVX  @DPTR,A

MOV   DPTR,#LORNCH    ;STORE # OF CHAR. IN CHAR BUFFER
MOV   A,R7
MOVX  @DPTR,A

MOV   DPTR,#MESINP    ;SET MESSAGE INPROCESS
MOVX  A,@DPTR
ORL   A,R5
MOVX  @DPTR,A

TIMSET:
MOV   DPTR,#TLRRTO    ;SET LORAN GOOD MESSAGE TIMER
MOV   A,#VLRRTO
MOVX  @DPTR,A

LORRXT:
CLR   RI              ;CLEAR RECEIVE INTERRUPT

```

```

CLR      TI                ;CLEAR TRANSMIT INTERRUPT IF ANY
SETB    EA                ;ENABLE INTERRUPTS

POP      1CH              ;RESTORE R4
POP      1DH              ;RESTORE R5
POP      1EH              ;RESTORE R6
POP      1FH              ;RESTORE R7
POP      DPL              ;
POP      DPH              ;RESTORE DPTR
POP      ACC              ;RESTORE ACCUMULATOR
POP      PSW              ;RESTORE PROGRAM STATUS WORD
SETB    EA                ;ENABLE INTERRUPTS

```

RETI

END

\$TITLE (ALTHAN 8052 BASED)
NAME ALTHAN

;ORIGINAL RELEASE 12 JAN 90 VERSION A001
;LAST MODIFIED 02 FEB 90 10:49

;VA001 12 JAN 90

;Basic Description of Module

; +note+ This module and the label ALTLEN must change if the number
; ALTLEN is changed from 3.
; This module gets the 9 bit input from the Altimeter multiplies it times
; 3 to and adds to the base of the table to get the 3 digit ASCII code in
; the table ALTTAB. It then puts the three digit ASCII code in the buffer
; ALTUD for use by other modules.

; *****
; *****
; * * * * *
; * * P U B L I C S * * * * *
; * * * * *
; *****
; *****

PUBLIC ALTHAN ;Input to this module

\$EJECT
; *****
; *****
; * * * * *
; * * E X T E R N A L S * * * * *
; * * * * *
; *****
; *****

EXTRN XDATA (ALTBUF) ;ADDRESS FOR ALTITUDE
EXTRN XDATA (ALTIINH) ;ALTIMETER INHIBIT

EXTRN NUMBER (PORTA1,PORTB1) ;ALTITUDE INPUTS
EXTRN NUMBER (ALT9IN) ;9 INPUT ALTIMETER

; *****
; *****
; * * * * *
; * * M A C R O S * * * * *
; * * * * *
; *****
; *****

; DEFINE BYTE WITH LISTING

*DEFINE (DBE (CODEDATA)) (

DB %CODEDATA
)

\$EJECT

; *****
; *****
; * * * *
; * * ALTITUDE TABLE * *
; * * * *
; *****
; *****

PROG_S segment CODE
RSEG PROG_S

ALTAB:

%DBE ('998998-10998-08998-09998998-03-05-04')
%DBE ('-07998-06998998007005006003998004998')
%DBE ('998-02000-01002998001998998027025026')
%DBE ('023998024998998018020019022998021998')
%DBE ('998008010009012998011998998017015016')
%DBE ('013998014998998067065066063998064998')
%DBE ('998058060059062998061998998048050049')
%DBE ('052998051998998057055056053998054998')
%DBE ('998028030029032998031998998037035036')
%DBE ('033998034998998047045046043998044998')
%DBE ('998038040039042998041998998147145146')
%DBE ('143998144998998138140139142998141998')
%DBE ('998128130129132998131998998137135136')
%DBE ('133998134998998108110109112998111998')
%DBE ('998117115116113998114998998127125126')
%DBE ('123998124998998118120119122998121998')
%DBE ('998068070069072998071998998077075076')
%DBE ('073998074998998087085086083998084998')
%DBE ('998078080079082998081998998107105106')
%DBE ('103998104998998098100099102998101998')
%DBE ('998088090089092998091998998097095096')
%DBE ('093998094998998307305306303998304998')
%DBE ('998298300299302998301998998288290289')
%DBE ('292998291998998297295296293998294998')
%DBE ('998268270269272998271998998277275276')
%DBE ('273998274998998287285286283998284998')
%DBE ('9982782802792829982819989982828230229')
%DBE ('232998231998998237235236233998234998')
%DBE ('998247245246243998244998998238240239')
%DBE ('242998241998998267265266263998264998')
%DBE ('998258260259262998261998998248250249')
%DBE ('252998251998998257255256253998254998')
%DBE ('998148150149152998151998998157155156')
%DBE ('153998154998998167165166163998164998')
%DBE ('998158160159162998161998998187185186')
%DBE ('183998184998998178180179182998181998')
%DBE ('998168170169172998171998998177175176')
%DBE ('173998174998998227225225223998224998')
%DBE ('998218220219222998221998998208210209')
%DBE ('212998211998998217215216213998214998')
%DBE ('998188190189192998191998998197195196')
%DBE ('193998194998998207205206203998204998')
%DBE ('998198200199202998201998998627625626')
%DBE ('623998624998998618620619622998621998')
%DBE ('998608610609612998611998998617615616')
%DBE ('613998614998998588590589592998591998')
%DBE ('998597595596593998594998998607605606')
%DBE ('603998604998998598600599602998601998')
%DBE ('998548550549552998551998998557555556')
%DBE ('553998554998998567565566563998564998')
%DBE ('998558560559562998561998998587585586')
%DBE ('583998584998998578580579582998581998')
%DBE ('998568570569572998571998998577575576')
%DBE ('573998574998998468470469472998471998')

```

%DBE ('998477475476473998474998998487485486')
%DBE ('483998484998998478480479482998481998')
%DBE ('998507505506503998504998998498500499')
%DBE ('502998501998998488490489492998491998')
%DBE ('998497495496493998494998998547545546')
%DBE ('543998544998998538540539542998541998')
%DBE ('998528530529532998531998998537535536')
%DBE ('533998534998998508510509512998511998')
%DBE ('998517515516513998514998998527525526')
%DBE ('523998524998998518520519522998521998')
%DBE ('998308310309312998311998998317315316')
%DBE ('313998314998998327325326323998324998')
%DBE ('998318320319322998321998998347345346')
%DBE ('343998344998998338340339342998341998')
%DBE ('998328330329332998331998998337335336')
%DBE ('333998334998998387385386383998384998')
%DBE ('998378380379382998381998998368370369')
%DBE ('372998371998998377375376373998374998')
%DBE ('998348350349352998351998998357355356')
%DBE ('353998354998998367365366363998364998')
%DBE ('998358360359362998361998998467465466')
%DBE ('463998464998998458460459462998461998')
%DBE ('998448450449452998451998998457455456')
%DBE ('453998454998998428430429432998431998')
%DBE ('998437435436433998434998998447445446')
%DBE ('443998444998998438440439442998441998')
%DBE ('998388390389392998391998998397395396')
%DBE ('393998394998998407405406403998404998')
%DBE ('998398400399402998401998998427425426')
%DBE ('423998424998998418420419422998421998')
%DBE ('998408410409412998411998998417415416')
%DBE ('413998414998')

```

\$EJECT

```

; *****
; *****
; * * * * *
; * * * * * P R O G R A M * * * * *
; * * * * *
; *****
; *****

```

;The following gets the 11 bit input from the altimeter and puts
; it in the registers R6 and R7. The most significant bit is bit 2
; of R6 in descending order, the eight in descending order in R7.

ALTHAN:

```

;SEE IF ALTIMETER INHIBIT IS ON

MOV DPTR,#ALTHIN ;GET ALTIMETER INHIBIT FLAG
MOVX A,@DPTR
JZ ALTHN1 ; NO CONTINUE
JMP ALTEXT

```

ALTHN1:

```

;THE FOLLOWING READS AND FORMATS THE ALTIMETER INPUTS

MOV DPTR,#PORTA1 ;GET PORT A1 FOR MOST SIG
MOVX A,@DPTR ; INPUTS
CPL A ; INVERT
MOV R6,A

MOV DPTR,#PORTB1 ;GET PORT B1 FOR REST
MOVX A,@DPTR
CPL A ; INVERT
MOV R7,A

; THE FOLLOWING CONDITIONS THE DATA

MOV A,R6 ;GET MSB PART OF INPUT
ANL A,#ALT9IN ;GET USED BITS
MOV R6,A ; AND SAVE

```

```

JZ      GET_DATA          ;CONTINUE IF TOTAL # <1024

MOV     R5,#'9'          ;SET ERROR CODE '999'
MOV     R6,#'9'
MOV     R7,#'9'
JMP     STORE_DATA      ; GO SEND ERROR

```

; The following checks adds R6 + R7 to DPL three times, then gets
; the 3 digit altitude from the table and stores it in the buffer ALTTUD.

```

GET_DATA:
MOV     DPTR,#ALTTAB    ; GET TABLE ADDRESS
MOV     R3,#3          ;GET HOW MANY TIMES TO REPEAT
GET_DAT1:
MOV     A,R7           ;GET LEAST SIG PART OF ADDRESS
ADD     A,DPL
MOV     DPL,A         ; ADD AND STORE

MOV     A,R6           ;GET NEXT MOST SIG DIGIT
ADDC   A,DPH          ; AND ADD W/CARRY TO DPH
MOV     DPH,A

DJNZ   R3,GET_DAT1    ;ADD THREE TIMES TO LOOP

CLR     A
MOVC   A,@A+DPTR      ;GET MOST SIG DIGIT
MOV     R5,A          ; AND SAVE

INC     DPTR
CLR     A
MOVC   A,@A+DPTR      ;GET MIDDLE DIGIT
MOV     R6,A          ; AND SAVE

INC     DPTR
CLR     A
MOVC   A,@A+DPTR      ;GET LEAST SIG DIGIT
MOV     R7,A          ; AND SAVE

```

```

STORE_DATA:
;THE FOLLOWING STORES THE DATA FOR THE RADIO AND
; THE LAP TOP COMPUTER

MOV     DPTR,#ALTBUF   ;GET OUTPUT ADDRESS
MOV     A,R5           ;GET MOST SIG DIGIT
MOVX   @DPTR,A        ; OUTPUT

INC     DPTR           ; GET NEXT ADDRESS
MOV     A,R6           ;GET MOST SIG DIGIT
MOVX   @DPTR,A        ; OUTPUT

INC     DPTR           ;GET MOST SIG DIGIT
MOV     A,R7           ; GET MOST SIG DIGIT
MOVX   @DPTR,A        ; OUTPUT

```

```

ALTEXT:
RET     ; EXIT MODULE

```

END

\$TITLE (LAPHAN 8052 BASED)
NAME LAPHAN

;VA001 ORIGINAL RELEASE
;STARTED 16 OCT 89

;LAST CHANGE VA005 21 MAY 90 16:13

```

;VA002 14 FEB 90  ADDED LORAN STATUS
;          CHANGED RESET
;VA003 23 FEB 90  CHANGED LORAN METHOD OF UPDATE
;VA004 15 MAY 90  ADDED RECEIVE LED
;          RESET TRANSMIT TIMER WHEN RECEIVE
;VA005 21 MAY 90  CHANGED RESET MESSAGE

```

;Basic Description of Module

; MESSAGES IN:

;This module will interface with the Lap Top Computer, both transmit
; and receive. When LAPCOM is called it will put the character in the
; ACC in the transmit buffer and begin transmitting if not already
; transmitting. If a reset message "R" is received from the computer,
; this module will set the NWDRST flag which will cause the watchdog
; to expire and reset the system. If a Time message "T" is received
; the VTMCNT value for the TIMCNT counter will be adjusted. This counter
; reflects how many counts must expire before the location data goes to
; the radio and to the Lap Top computer. This module will be responsible
; for generating the # message to the Computer. This message will be sent
; just after initialization.

; MESSAGES OUT:

; From Loran Receiver:

; This module will monitor the flag LATOFT to see if a message is to go to
; the Lap Top Computer, then it will send the start bracket and the message
; found at AIRCID, PLTBUF, PLNBUF, ALTBUF, Data-V-Com flags and Loran Status
; insert the spaces and then send the check sum and End Bracket.

; From Radio:

; This module will monitor the RDTLTF flag when it is set it will send the
; message found at RDTLTB and add a carriage return.

\$EJECT

; *****
; *****
; * * * * *
; * * P U B L I C S * *
; * * * * *
; *****
; *****

PUBLIC LAPHAN ;INPUT TO THIS MODULE
PUBLIC LAPINL ;INITIALIZATION OF MODULE
PUBLIC LAPRIN ;RECEIVE INTERRUPT HANDLER
PUBLIC LAPTIN ;TRANSMIT INTERRUPT HANDLER

\$EJECT

; *****
; *****
; * * * * *
; * * E X T E R N A L S * *
; * * * * *
; *****
; *****

EXTRN CODE (PRTCOM) ;SEND TO PRINTER
EXTRN NUMBER (LAPSTS,LAPDAT) ;USART STATUS AND DATA
EXTRN NUMBER (RECVOF,RECVON) ;COMMAND INSTRUCTIONS
EXTRN NUMBER (TRANON,TRANOF) ;FOR RECEIVE AND TRANSMIT
EXTRN NUMBER (RXRDY) ;RECEIVE READY
EXTRN NUMBER (TXRDY,TXEMPT) ;TRANSMIT READY/EMPTY
EXTRN XDATA (LAPRCT) ;RECEIVE INTERRUPT COUNT
EXTRN XDATA (LAPTCT) ;TRANSMIT INTERRUPT COUNTER
EXTRN XDATA (RERUNA) ;RESET AND RERUN
EXTRN NUMBER (RERUNV) ; RESET VALUE
EXTRN XDATA (NWDRST) ;IGNORE WATCH DOG RESET
EXTRN XDATA (TLPART) ;RECEIVE TIMEOUT TIMER
EXTRN XDATA (TLPRTO) ;UART REFRESH TIMER


```

EXTRN XDATA (VTRCNT) ;VALUE FOR MESSAGE WAIT TIMER (RADIO)
EXTRN XDATA (TTRCNT) ;MESSAGE COUNTER BUFFER (RADIO)

EXTRN XDATA (VLRcnt) ;VALUE FOR MESSAGE WAIT TIMER (LAP TOP)
EXTRN XDATA (TLRCNT) ;MESSAGE COUNTER BUFFER (LAP TOP)

EXTRN NUMBER (LAPMIF) ; MODE INSTRUCTION FORMAT
EXTRN NUMBER (TRSRES) ;SOFTWARE RESET

; LORAN RECEIVER INTERFACE

EXTRN XDATA (LRTLTF) ;DATA READY FOR LAP TOP COMPUTER
EXTRN XDATA (PLTBUF,PLNBUF) ;LATTITUDE AND LONGITUDE BUFFERS
EXTRN NUMBER (PLTLEN,PLNLEN) ;LATTITUDE AND LONGITUDE LENGTH
EXTRN XDATA (STSBUF) ;LORAN STATUS BUFFER
EXTRN NUMBER (STSLEN) ;LORAN STATUS BUFFER LENGTH

; ALTIMETER INTERFACE

EXTRN XDATA (ALTBUF) ;ALTIMETER DATA
EXTRN NUMBER (ALTLEN) ;ALTIMETER DATA LENGTH

; SPACES FOR RADIO AND LORAN MESSAGE

EXTRN NUMBER (LATSPC)

; RADIO INTERFACE

EXTRN XDATA (RDTLTF,RDTLTB,RDTLTE)

; MESSAGE DEFINITION

EXTRN XDATA (AIRCID) ;AIRCRAFT ID IN NONVOLATILE RAM
EXTRN NUMBER (AIRIDN) ;LENGTH OF AIRCRAFT ID
EXTRN NUMBER (RSTSPC) ;NUMBER OF SPACES FOR RESET MESSAGE
EXTRN NUMBER (MSGTOT) ;MESSAGE LENGTH

; SELFTEST INTERFACE

EXTRN XDATA (SLFFLG) ;SELFTEST FLAG

EXTRN NUMBER (NOLRLT) ;NO LORAN UPDATE FOR LAPTOP
EXTRN NUMBER (TRRDST) ;NOT TRANSMITTED TO RADIO
EXTRN NUMBER (TRLTST) ;NOT TRANSMITTED TO LAPTOP
EXTRN NUMBER (OUTFLG) ;NOT ACCEPTING OTHER AIRCRAFT MSG'S

EXTRN XDATA (TRCLED) ;RECEIVE TIME ON

; MISC.
EXTRN NUMBER (CR)
EXTRN NUMBER (ENDBRK) ;END BRACKET
EXTRN NUMBER (STBRK) ;START BRACKET
;
; *****
; *****
; * * * * *
; * * * * *
; * * * * *
; * * * * *
; *****
; *****

;THIS IS DATA USED INTERNAL TO THIS FILE ONLY

DATA_AREA segment XDATA
PROG_S segment CODE
RSEG DATA_AREA

XTODPO: DS 2 ;OUTPUT POINTER FOR BUFFER
XTODPI: DS 2 ;INPUT POINTER FOR BUFFER

CHKDPI: DS 2 ;CHECK SUM POINTER

XTODBF: DS 500 ;DATA BUFFER FOR BUFFER
XTDEND: DS 1 ;END OF BUFFER

```

```
RFMDPT:      DS  2      ;RECEIVE POINTER FROM LAP TOP
LTTDVB:      DS  4      ;MESSAGE BUFFER FROM LAP TOP COMPUTER
LTTDVE:      DS  1      ;ENT OF LAPTOP BUFFER
RCVINP:      DS  1      ;PROPER RECEIVE MESSAGE BEGUN
TRNINP:      DS  1      ;TRANSMIT IN PROCESS
RSTMSG:      DS  1      ;RESET OR LP MESSAGE SENT
RDDCMN:      DS  1      ;LAST USART COMMAND
```

```
; *****
; *****
; * * * * *
; * *   E Q U A T E S   * *
; * * * * *
; *****
; *****
;
```

```
VRFDTO      EQU      12      ;REINITIALIZE AFTER 3 SECONDS
VDUART      EQU      8       ;USART REFRESH TIMER
VRCLED      EQU      04      ;RECEIVE LED ON TIME
```

```
$EJECT
; *****
; *****
; * * * * *
; * *   M A C R O S   * *
; * * * * *
; *****
; *****
;
```

;JUMP (LONG JUMP) IF A> CDATA

```
*DEFINE (JAG (CDATA,DEST)) LOCAL LABEL(
    PUSH ACC
    CLR C
    SUBB A,#CDATA
    JZ %LABEL
    JC %LABEL
    POP ACC
    JMP %DEST
```

```
%LABEL:
    POP ACC
)
```

```
$EJECT
; *****
; *****
; * * * * *
; * *   P R O G R A M   * *
; * * * * *
; *****
; *****
;
```

RSEG PROG_S

LAPHAN:

;THE FOLLOWING IS TO REFRESH BAUD RATE IF NESCESSARY

```
MOV    DPTR,#TRNINP      ;TRANSMIT IN PROCESS
MOVX   A,@DPTR           ;
JZ     REFREC            ; NO GO REFRESH RECEIVE

MOV    DPTR,#LAPSTS      ;GET STATUS
```

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```

MOVX   A,@DPTR           ;TX EMPTY
ANL    A,#TXEMPTY
JNZ    REFREC           ; NO CONTINUE

MOV    DPTR,#RDDCMN     ;GET PAST STATUS
MOVX   A,@DPTR
ORL    A,#TRANON
MOVX   @DPTR,A
MOV    DPTR,#LAPSTS     ;REFRESH USART
MOVX   @DPTR,A

```

REFREC:

```

;SEE IF NEED TO REFRESH USART

MOV    DPTR,#TLPRTO     ;HAS MESSAGE TIMED OUT
MOVX   A,@DPTR
JNZ    LAPLBM           ; NO CONTINUE

MOV    DPTR,#RCVINP     ;RECEIVE IN PROCESS
MOVX   @DPTR,A
JZ     LAPLBM           ; NO CONTINUE

MOV    DPTR,#RDDCMN     ;GET PAST STATUS
MOVX   A,@DPTR
ORL    A,#RECVON
MOVX   @DPTR,A
MOV    DPTR,#LAPSTS     ;REFRESH RECEIVE
MOVX   @DPTR,A

CALL   LPRINL           ;REINITIALIZE

```

LAPLBM:

```

;SEE IF RESET OR # MESSAGE NEEDS TO GO OUT

MOV    DPTR,#RERUNA     ;GET RERUN ADDRESS
MOVX   A,@DPTR
CJNE   A,#RERUNV,LAPLOR ; NOT READY

MOV    DPTR,#RSTMSG     ;FLAG SET MESSAGE HAS GONE OUT
MOVX   A,@DPTR
JNZ    LAPLOR
CPL    A
MOVX   @DPTR,A
CALL   RSTOUT           ; SEND OUT RESET MESSAGE

```

```

;THE FOLLOWING CHECKS TO SEE IF MESSAGES ARE READY TO
; GO TO THE LAP TOP COMPUTER.

```

LAPLOR:

```

CLR    EA               ;DISABLE INTERRUPTS

MOV    DPTR,#LRTLTF     ;MESSAGE FROM LORAN RECEIVER
MOVX   A,@DPTR
JZ     LAPRAD

CLR    A
MOVX   @DPTR,A         ; YES CLEAR FLAG

MOV    DPTR,#XTODPI     ;SAVE POINTER FOR CHECKSUM
MOVX   A,@DPTR
MOV    R0,A
INC    DPTR
MOVX   A,@DPTR
MOV    R3,A

MOV    A,#STBRK
CALL   LAPCOM

MOV    DPTR,#AIRCID     ;GET AIRCRAFT ID LOCATION
MOV    R7,#AIRIDN
CALL   R7DOUT           ;GET NUMBER OF DIGITS IN AIRCRAFT ID
                        ;OUTPUT THE AIRCRAFT ID.

```

```

MOV      DPTR,#PLTBUF      ;GET THE LATITUDE
MOV      R7,#PLTLEN      ;GET LENGTH FOR LATITUDE
CALL     R7DOUT           ;OUTPUT THE LATITUDE

MOV      DPTR,#PLNBUF      ;GET THE LONGITUDE
MOV      R7,#PLNLEN      ;GET LENGTH FOR LONGITUDE
CALL     R7DOUT           ;OUTPUT THE LONGITUDE

MOV      DPTR,#ALTBUF      ;GET ALTITUDE BUFFER
MOV      R7,#ALTLEN      ;GET ALTITUDE LENGTH
CALL     R7DOUT           ;OUTPUT THE ALTITUDE

MOV      DPTR,#SLFFLG      ;GET SELFTEST FLAG
MOVX     A,@DPTR
CALL     HEXOUT           ; OUTPUT

MOVX     A,@DPTR           ;GET SELFTEST FLAGS
ORL      A,#NOLRLT        ;SET NO LORAN UPDATE
ORL      A,#TRRDST        ;SET NO MESSAGE TO RADIO
MOV      R7,A             ;SAVE

MOV      A,#TRLTST        ;CLEAR MSG. SENT LAR TOP
CPL      A
ANL      A,1FH
MOVX     @DPTR,A

MOV      DPTR,#STSBUF      ;GET STATUS BUFFER
MOV      R7,#STSLLEN      ;GET LENGTH OF BUFFER
CALL     LORCON           ;OUTPUT LORAN STATUS

MOV      A,#' '           ;GET A SPACE
MOV      R7,#LATSPC       ; GET HOW MANY TO OUTPUT

LPLRLP:
CALL     LAPCOM           ; OUTPUT
DJNZ     R7,LPLRLP        ; ALL OF THEM

MOV      R5,#MSGTOT-3     ;GET TOTAL LENGTH OF MESSAGE

MOV      DPTR,#XTDEND      ;GET END OF BUFFER
MOV      R4,DPL           ; AND SAVE
MOV      R6,DPH

MOV      DPH,R0           ;GET START OF MESSAGE
MOV      DPL,R3
CALL     CHKADD           ;ADD CHECKSUM TO ACC
CALL     HEXOUT           ;PUT IN BUFFER

MOV      A,#ENDBRK        ; OUTPUT CARRIAGE RETURN
CALL     LAPCOM

LAPRAD:

SETB     EA               ;ENABLE INTERRUPTS

;MESSAGE FROM RADIO RECEIVER

MOV      DPTR,#RDTLTF      ;MESSAGE FROM LORAN RECEIVER
MOVX     A,@DPTR
JZ       LAPEXT
CLR      A
MOVX     @DPTR,A         ; YES CLEAR FLAG

MOV      DPTR,#TRCLED      ; SET LED TIMER
MOV      A,#VRCLED
MOVX     @DPTR,A

MOV      DPTR,#VTRCNT      ;GET WAIT COUNT
MOVX     A,@DPTR
RR       A               ;DIVIDE BY 2
MOV      DPTR,#TTRCNT      ; AND RESET OUT
MOVX     @DPTR,A

MOV      DPTR,#SLFFLG      ;MESSAGES ON
MOVX     A,@DPTR

```

```

ANL      A, #OUTFLG
JNZ      LAPEXT          ; NO CONTINUE

CLR      EA              ;DISABLE INTERRUPTS

MOV      DPTR, #XTODPI   ;SAVE POINTER FOR CHECKSUM
MOVX     A, @DPTR
MOV      R0, A           ;
INC      DPTR
MOVX     A, @DPTR
MOV      R1, A

MOV      A, #STTBK      ;SEND START BRACKET
CALL     LAPCOM

MOV      DPTR, #RDTLTE  ;GET END OF MESSAGE
MOV      R7, DPL        ; AND SAVE

DEC      R7              ;DON'T GET CHECKSUM

MOV      DPTR, #RDTLTB  ;GET START OF MESSAGE
RADLOP:  MOVX     A, @DPTR

CALL     LAPCOM          ;OUTPUT TO BUFFER

INC      DPTR
MOV      A, DPL          ;SEE IF AT END OF MESSAGE
CJNE    A, 1FH, RADLOP  ; NO CONTINUE

MOV      A, #' '         ;GET A SPACE
MOV      R7, #LATSPC    ; GET HOW MANY TO OUTPUT
LPRDLP:  CALL     LAPCOM          ; OUTPUT

DJNZ    R7, LPRDLP      ; ALL OF THEM

MOV      R5, #MSGTOT-3  ;GET TOTAL LENGTH OF MESSAGE

MOV      DPTR, #XTDEND  ;GET END OF BUFFER
MOV      R4, DPL        ; AND SAVE
MOV      R6, DPH

MOV      DPH, R0        ;GET START OF MESSAGE
MOV      DPL, R1

CALL     CHKADD          ;ADD CHECKSUM TO ACC
CALL     HEXOUT          ;PUT IN BUFFER

MOV      A, #ENDBRK    ; OUTPUT CARRIAGE RETURN
CALL     LAPCOM

LAPEXT:  SETB     EA              ;ENABLE INTERRUPTS

MOV      DPTR, #TRCLED  ;RECEIVE LED ON?
MOVX     A, @DPTR
SETB    P1.7            ;TURN OFF
JZ       LAPETO         ; AND LEAVE OFF
CLR      P1.7           ; NO TURN ON
LAPETO:  RET

$EJECT

```

```

; *****
; *****
; * * * * *
; * * * * *   I N I T I A I Z A T I O N   * * * * *
; * * * * *
; *****
; *****

```

```

;
; The following is called once by the INIT module.
;
;The following is for the initialization of the serial port used
; for transmitting to the Computer. It is initialized, and enabled.
;*****
;Baud Rate - 1200
;Start Bits - 1
;Data Bits - 8
;Stop Bits - 2
;Parity - None
;*****

```

LAPINL:

```

MOV     DPTR,#SLFFLG      ;SET OUTFLG
MOVX    A,@DPTR
ORL     A,#OUTFLG
MOVX    @DPTR,A          ; WILL NOT TRANSMIT TO OTHER AIRCRAFT

MOV     DPTR,#LAPSTS     ;
MOV     A,#TRSRES        ;RESET USART
MOVX    @DPTR,A          ;

MOV     R7,#3FH          ;WAIT A BIT
DJNZ    R7,$

MOV     A,#LAPMIF        ;SET MODE INSTRUCTION FORMAT
MOVX    @DPTR,A          ;

MOV     A,#RECVON        ;GET COMMAND INSTRUCTION FORMAT

MOVX    @DPTR,A          ;SET

MOV     DPTR,#RDDCMN     ;SET IN MEMORY
MOVX    @DPTR,A

MOV     DPTR,#LAPDAT     ;DO A DUMMY READ
MOVX    A,@DPTR

```

;SET ROTATING BUFFER POINTERS (OUTPUT)

```

MOV     DPTR,#XTODBF     ;GET START OF BUFFER
MOV     R4,DPL           ;SAVE LSB
MOV     R5,DPH           ;SAVE MSB

MOV     DPTR,#XTODPO     ;SET COM. OUT POINTER
MOV     A,R5             ;
MOVX    @DPTR,A         ;SAVE MSB
INC     DPTR             ;
MOV     A,R4             ;SAVE LSB
MOVX    @DPTR,A         ;

MOV     DPTR,#XTODPI     ;SET COM. IN POINTER
MOV     A,R5             ;
MOVX    @DPTR,A         ;SAVE MSB
INC     DPTR             ;
MOV     A,R4             ;SAVE LSB
MOVX    @DPTR,A         ;

```

;The following sets the input pointer

LPRINL:

```

MOV     DPTR,#LTTDVB     ; RESET INPUT POINTER
MOV     R4,DPL           ;
MOV     A,DPH           ; TO START OF BUFFER
MOV     R5,A             ;
MOV     DPTR,#RFMDPT     ;
MOVX    @DPTR,A         ;
INC     DPTR             ;
MOV     A,R4             ;
MOVX    @DPTR,A         ;

```

```

MOV   DPTR, #RCVINP   ;CLEAR RECEIVE INPROCESS
CLR   A
MOVX  @DPTR, A
RET   ;EXIT INITIALIZATION

```

```

$EJECT

```

```

; *****
; *****
; * * * * *
; * *   S U B R O U T I N E S   * * *
; * * * * *
; *****
; *****

```

```

;The following subroutine converts the data in the buffer STSBUF of
; length STSLEN and puts the hex characters in the Transmit buffer.

```

```

LORCON:  MOV   R1, #4           ;GET NUMBER OF BITS TO GO OUT
         MOV   R2, #0           ; CLEAR MEMORY
LORCN1:  MOVX  A, @DPTR         ;GET CHARACTER
         INC   DPTR
         CJNE  A, #2DH, LORCN2  ;IS FLAG SET
         CLR   C               ; NO CLEAR
         JMP   LORC2A
LORCN2:  SETB  C
LORC2A:  MOV   A, R2
         RLC
         MOV   R2, A
         DJNZ  R7, LORCN3      ; WHEN DONE EXIT
         DEC   R1
         JMP   LORCN4
LORCN3:  DJNZ  R1, LORCN1      ; CONTINUE TIL BYTE IS SET
         CALL  BINOUT          ;OUTPUT WHEN FULL
         JMP   LORCON          ; AND CONTINUE
LORCN4:  CLR   C
         RLC
         DJNZ  R1, LORCN4      ; CONTINUE TIL BYTE IS SET
         CALL  BINOUT          ;OUTPUT FULL
         RET                   ;EXIT SUBROUTINE

```

```

;THE FOLLOWING SUBROUTINE ADDS UP THE DATA STARTING AT DPTR FOR
; THE CHECKSUM CALCULATION.

```

```

CHKADD:
CHKLOP:  MOV   R7, #0           ;CLEAR R7
         MOVX  A, @DPTR         ;GET DATA
         ADD   A, R7            ;ADD DATA TO CHECKSUM
         MOV   R7, A           ;SAVE
         INC   DPTR            ;INC POINTER
         MOV   A, DPL
         CJNE  A, 1CH, CHK1     ; NOT AT END CONTINUE
         MOV   A, DPH
         CJNE  A, 1EH, CHK1

```

```

CHKL1:  MOV     DPTR,#XTODBF    ;AT END RESET

        DJNZ   R5,CHKLOP

        MOV    A,R7

        RET                      ;EXIT SUBROUTINE

```

```

;THE FOLLOWING SUBROUTINE CONVERTS THE DATA IN THE ACC TO ASCII AND
; OUTPUTS IT

```

```

HEXOUT:  PUSH   ACC

        SWAP  A                ;GET MSB
        CALL  BINOUT           ;OUTPUT IT

        POP   ACC
        CALL  BINOUT           ;GET LSB
                                ; AND OUTPUT IT

        RET

```

```

;This subroutine take the ACC and converts to ASCII and outputs it
; to the console

```

```

BINOUT:  ANL    A,#0FH          ;STRIP OFF MSB

        SJL   (9,BINOT1)      ;GREATER THAT 9

        ORL   A,#30H          ;MAKE ASCII
        JMP   BINOT2          ;CONTINUE

BINOT1:  ADD    A,#37H          ;MAKE ASCII

BINOT2:  CALL  LAPCOM          ;OUTPUT TO LAP TOP
        RET                    ;EXIT SUBROUTINE

```

```

; This subroutine when called saves all microprocessor pointers
; acc etc. and takes the value that is in the acc. and outputs it
; serialy.

```

```

LAPCOM:

        PUSH  PSW              ;SAVE PSW
        PUSH  DPH              ;SAVE DPTR
        PUSH  DPL              ;
        PUSH  ACC              ;SAVE ACC
        PUSH  1CH              ;R4
        PUSH  1DH              ;R5
        PUSH  1FH              ;R7
        PUSH  1EH              ;R6

        MOV   R5,A             ;SAVE DATA

;      PUT IN BUFFER AT POINT POINTED TO BY XTODPI

        MOV   DPTR,#XTDEND     ;GET END OF BUFFER
        MOV   R4,DPL           ; AND SAVE
        MOV   R6,DPH

        MOV   DPTR,#XTODPI     ;PUT IN BUFFER AT XTODPI
        MOVX  A,@DPTR          ;GET MSB
        MOV   R7,A             ;SAVE
        INC   DPTR              ;
        MOVX  A,@DPTR          ;GET LSB
        MOV   DPL,A            ;AND SAVE
        MOV   DPH,R7           ;SAVE MSB

```



```

NOCONB:
    MOV     A,R5           ;GET DATA
    MOVX   @DPTR,A       ;SAVE DATA IN BUFFER

;   RESET POINTER

    INC     DPTR          ;INC POINTER
    MOV     A,DPL         ;AT END OF BUFFER
    CJNE   A,1CH,COM1    ; NOT AT END CONTINUE

    MOV     A,DPH         ;NOW CHECK MSB
    CJNE   A,1EH,COM1

    MOV     DPTR,#XTODBF ;AT END RESET
COM1:
    MOV     R5,DPL        ;GET LSB
    MOV     A,DPH        ;GET MSB
    MOV     DPTR,#XTODPI ;IN COM IN POINTER
    MOVX   @DPTR,A       ;SAVE MSB
    INC     DPTR         ;
    MOV     A,R5         ;SAVE LSB
    MOVX   @DPTR,A       ;

    MOV     DPTR,#TRNINP ;TRANSMIT IN PROCESS?
    MOVX   A,@DPTR      ;
    JNZ    COM2          ; YES CONTINUE
    MOV     A,#0FFH     ; NO SET FLAG
    MOVX   @DPTR,A      ;

    MOV     DPTR,#RDDCMN ;ENABLE TRANSMIT
    MOVX   A,@DPTR      ; MEMORY
    ORL    A,#TRANON
    MOVX   @DPTR,A

    MOV     DPTR,#LAPSTS ;ENABLE TRANSMIT
    MOVX   @DPTR,A      ;

COM2:
    POP     1EH          ;R6
    POP     1FH          ;R7
    POP     1DH          ;R5
    POP     1CH          ;R4
    POP     ACC
    POP     DPL
    POP     DPH
    POP     PSW

    RET

```

;The following subroutine sends out the # message or reset message
; after the system has been initialized and is up and running in the
; main loop.

```

RSTOUT:
    MOV     DPTR,#XTODPI ;SAVE POINTER FOR CHECKSUM
    MOVX   A,@DPTR

    MOV     R0,A
    INC     DPTR
    MOVX   A,@DPTR
    MOV     R1,A

    MOV     A,#STTBK     ;OUTPUT A START BRACKET
    CALL   LAPCOM

    MOV     A,###       ;GET LB SYMBOL
    CALL   LAPCOM       ;OUTPUT

    MOV     DPTR,#AIRCID ;GET AIRCRAFT ID LOCATION
    MOV     R7,#AIRIDN  ;GET NUMBER OF DIGITS IN AIRCRAFT ID

    CALL   R7DOUT       ;OUTPUT AIRCRAFT ID.

```

```

MOV     DPTR,#SLFFLG   ;GET SELFTEST FLAG
MOVX   A,@DPTR
CALL   HEXOUT         ; OUTPUT

MOV     R7,#RSTSPC    ;GET NUMBER OF SPACES FOR
MOV     A,#' '        ; THIS MESSAGE
MOV     A,#' '        ;GET SPACE

RSTSPS:
CALL   LAPCOM         ; OUTPUT SPACES
DJNZ   R7,RSTSPS     ; UNTIL DONE

MOV     R5,#MSGTOT-3  ;GET TOTAL LENGTH OF MESSAGE

MOV     DPTR,#XTDEND  ;GET END OF BUFFER
MOV     R4,DPL        ; AND SAVE
MOV     R6,DPH

MOV     DPH,R0        ;GET START OF MESSAGE
MOV     DPL,R1
CALL   CHKADD        ;ADD CHECKSUM TO ACC
CALL   HEXOUT        ;PUT IN BUFFER

MOV     A,#ENDBRK    ;OUTPUT A END BRACKET
CALL   LAPCOM

RET

```

; The following subroutine outputs data at dptr for the number
; of times found in R7.

R7DOUT:

```

MOVX   A,@DPTR       ;GET ID NUMBER
CALL   LAPCOM        ; OUTPUT
INC    DPTR

DJNZ   R7,R7DOUT    ; UNTIL DONE

RET                ;EXIT SUBROUTINE

```

\$EJECT

```

; *****
; *****
; * *          INTERRUPT   HANDLER          * *
; * *          * *
; *****
; *****
;

```

; THIS INTERRUPT HANDLER HANDLES THE OUTPUTS TO THE DATA CONTROL HEAD
; AS WELL AS A JUMP FOR THE FIRST OUTPUT.
; A ROTATING BUFFER IS USED WITH THE INPUT TO THE BUFFER POINTED TO BY
; XTODPI AND THE OUTPUT POINTED TO BY XTODPO. WHEN THE LSB OF THE OUTPUT
; POINTER IS POINTING TO THE SAME POINT AS THE INPUT POINTER THE NOTHING
; IS TRANSMITTED AND THE INTERRUPT IS DISABLED.

LAPTIN:

```

PUSH   1BH           ;SAVE R3
PUSH   1CH           ;SAVE R4
PUSH   1DH           ;SAVE R5
PUSH   1EH           ;SAVE R6
PUSH   1FH           ;SAVE R7

; INC INTERRUPT COUNT

MOV     DPTR,#LAPTCT ;INC COUNT
MOVX   A,@DPTR      ;FOR INTERRUPTS
INC    A
MOVX   @DPTR,A

```

; CHECK TO SEE IF INPUT EQUALS OUTPUT POINTER

```

MOV     DPTR,#XTDEND           ;GET END
MOV     R7,DPL                 ; OF BUFFER
MOV     R6,DPH

MOV     DPTR,#XTODPI          ;GET INPUT POINTER
MOVX    A,@DPTR                ;MSB
MOV     R3,A
INC     DPTR                    ;LSB
MOVX    A,@DPTR                ;
MOV     R4,A                    ;SAVE
MOV     DPTR,#XTODPO          ;GET OUTPUT POINTER
MOVX    A,@DPTR                ;MSB
MOV     R5,A                    ;SAVE

INC     DPTR                    ;GET LSB
MOVX    A,@DPTR                ;

CJNE    A,1CH,INTE41          ;NOT THE SAME CONTINUE

MOV     A,R5                    ;GET MSB
CJNE    A,1BH,INTE41

MOV     DPTR,#RDDCMN          ;GET MEMORY
MOVX    A,@DPTR
ANL     A,#TRANOF              ;DISABLE TRANSMIT
MOVX    @DPTR,A

MOV     DPTR,#LAPSTS          ;DISABEL TRANSMIT FOR USART
MOVX    @DPTR,A

MOV     DPTR,#TRNINP          ;CLEAR TRANSMIT IN PROCESS FLAG
CLR     A
MOVX    @DPTR,A

JMP     IN4END                 ;GO TO END OF INTERRUPT

INTE41:
MOV     DPL,A                  ;SET POINTER TO OUTPUT
MOV     DPH,R5                 ;

MOVX    A,@DPTR                ;GET DATA TO OUTPUT
MOV     R5,A                    ; AND SAVE

INC     DPTR                    ;INCREMENT AND SAVE POINTER

MOV     A,DPL                  ;SEE IF AT END OF BUFFER
CJNE    A,1FH,INTE42          ; NO CONTINUE
MOV     A,DPH                  ; CHECK MSB
CJNE    A,1EH,INTE42          ; SEE IF DONE

INTE42:
MOV     DPTR,#XTODBF          ;YES SAVE START OF BUFFER
MOV     A,DPH                  ;GET MSB
MOV     R7,DPL                 ;GET LSB

MOV     DPTR,#XTODPO          ;GET OUT POINTER
MOVX    @DPTR,A                ;SAVE MSB
INC     DPTR                    ;
MOV     A,R7                    ;
MOVX    @DPTR,A                ;SAVE LSB

COUTDT:
MOV     DPTR,#LAPDAT          ; OUTPUT DATA
MOV     A,R5                    ;GET DATA OUTPUT ADDRESS
MOVX    @DPTR,A                ;
CALL    PRTCOM                 ;OUTPUT TO LAP TOP
                                ; OUTPUT TO PRINTER

IN4END:
POP     1FH                     ;POP R7
POP     1EH                     ;POP R6

```

```

POP      1DH                ;POP R5
POP      1CH                ;POP R4
POP      1BH                ;POP R3

RET

```

```

; This interrupt handler handles the data coming into
; the USART.

```

```

LAPRIN:
PUSH     1EH                ;SAVE R6
PUSH     1DH                ;SAVE R5
PUSH     1CH                ;SAVE R4
PUSH     1BH                ;SAVE R3

MOV      DPTR,#LAPRCT      ;INC COUNT
MOVX     A,@DPTR          ;FOR INTERRUPTS
INC      A
MOVX     @DPTR,A

MOV      DPTR,#TLPRTO     ;SET RECEIVE TIME OUT
MOV      A,#VRFDT0       ;TIMER
MOVX     @DPTR,A

MOV      DPTR,#LAPDAT     ;READ DATA
MOVX     A,@DPTR
MOV      R5,A            ; AND SAVE

CALL     PRTCOM           ;OUTPUT TO PRINTER

MOV      DPTR,#RCVINP     ;RECEIVE IN PROCESS?
MOVX     A,@DPTR
JNZ      LPRSAV          ; YES CONTINUE

MOV      A,R5             ;IS IT A VALID START OF MESSAGE?
CJNE     A,#'T',LAPRN1   ; YES SET FLAG
JMP      LAPRNE

LAPRN1:
CJNE     A,#'R',LAPRN2   ; NOT A VALID START OF MESSAGE
JMP      LAPRNE          ; YES SET FLAG

LAPRN2:
CJNE     A,#'L',LAPRN4   ; YES SET FLAG
JMP      LAPRNE

LAPRN3:
JMP      LPRRST          ; GO RESET

LAPRN4:
CJNE     A,#'O',LAPRN3   ; NO GO RESET

LAPRNE:
MOV      DPTR,#RCVINP     ;VALID START, SET RECEIVE FLAG
MOV      A,#OFFH
MOVX     @DPTR,A

LPRSAV:
MOV      DPTR,#RFMDPT     ;GET MESSAGE POINTER
MOVX     A,@DPTR
MOV      R4,A            ; MSB
INC      DPTR            ;
MOVX     A,@DPTR         ; LSB
MOV      DPL,A          ;AND SET IN DPTR
MOV      DPH,R4
MOV      A,R5
MOVX     @DPTR,A         ;SAVE DATA IN BUFFER
INC      DPTR            ;INCREMENT POINTER
MOV      A,DPL           ; SAVE POINTER
MOV      R4,DPL
MOV      R6,DPH

MOV      DPTR,#LITDVE+1   ;AT END OF BUFFER
MOV      R3,DPL

CJNE     A,1BH,LPRSV1    ; NO CONTINUE
JMP      LPRHAN          ; YES GO HANDLE RECEIVED MESSAGE

```

```

LPCMPE:      JMP      LPREND          ;EXIT
LPRSV1:      MOV      A,R5           ;CARRIAGE RETURN
              CJNE    A,#CR,LPCMPE   ; NO CONTINUE
LPRHAN:      MOV      DPTR,#LTDDVB   ;SEE WHAT KIND OF A MESSAGE IT IS
              MOVX    A,@DPTR
              CJNE    A,#'R',LPRHN1  ; RESET MESSAGE
              ; YES HANDLE RESET MESSAGE
              MOV      DPTR,#VTRCNT   ; CLEAR RADTIO
              CLR     A
              MOVX    @DPTR,A
              MOV      DPTR,#VLRCNT   ;CLEAR LORAN
              MOVX    @DPTR,A
              MOV      DPTR,#SLFFLG   ;TURN OFF OTHERS
              MOVX    A,@DPTR
              ORL     A,#OUTFLG
              MOVX    @DPTR,A
              MOV      DPTR,#NWRDST   ;SET IGNORE WATCHDOG FLAG
              MOV     A,#OFFH
              MOVX    @DPTR,A
LPRHN1:      JMP      LPRRST          ; CONTINUE
              CJNE    A,#'O',LPRHN2   ;NO AN O MESSAGE CONTINUE
              MOV      DPTR,#LTDDVB+1 ;GET LEAST SIG TIME DIGIT
              MOVX    A,@DPTR
              CJNE    A,#'0',LPOCLR   ; SET RECEIVE CLEAR
              INC     DPTR
              MOVX    A,@DPTR         ;GET MIDDLE DIGIT
              CJNE    A,#'0',LPOCLR   ;GET LEAST SIG. DIGIT
              INC     DPTR
              MOVX    A,@DPTR         ;GET LEAST SIG. DIGIT
              CJNE    A,#'0',LPOCLR   ;SET NO INPUT FROM OTHER AIRCRAFT
              MOV      DPTR,#SLFFLG   ;SET NO INPUT FROM OTHER AIRCRAFT
              MOVX    A,@DPTR
              ORL     A,#OUTFLG
              MOVX    @DPTR,A
LPOCLR:      JMP      LPRRST          ; CONTINUE
              MOV      DPTR,#SLFFLG   ;CLEAR NO INPUT FROM OTHER AIRCRAFT
              MOVX    A,@DPTR
              MOV     R7,A
              MOV     A,#OUTFLG
              CPL     A
              ANL     A,1FH
              MOVX    @DPTR,A
              JMP     LPRRST          ; CONTINUE
LPRHN2:      CJNE    A,#'T',LPRHN3   ; NOT A T CONTINUE
              ; HANDLE TIME MESSAGE
              ; SEE IF TIME MESSAGE IS COMPLETE
              MOV      DPTR,#LTDDVB+5 ; GET END
              MOV     A,DPL
              CJNE    A,1CH,LPRRST    ; NOT RIGHT NUMBER OF CHAR.
              ; MOST SIG TIME DIGIT IS NOT USED AT PRESENT
              MOV      DPTR,#LTDDVB+2 ;GET SEC. MOST SIG. TIME DIGIT
              MOVX    A,@DPTR
              ANL     A,#OFH          ;MAKE BCD
              MOV     R6,A            ;SAVE

```

```

RL      A                ;MULTIPLY BY 10
RL      A
RL      A
ADD     A,R6
ADD     A,R6

MOV     R6,A            ; SAVE
INC     DPTR            ;GET LEAST SIG DIGIT
MOVX   A,@DPTR
ANL    A,#0FH          ;MAKE BCD

ADD     A,R6            ;COMBINE MSB AND LSB

MOV     DPTR,#VTRCNT   ; SET IN TIMER
MOVX   @DPTR,A

CLR     A
MOV     DPTR,#TTRCNT   ; CLEAR RADIO COUNTER TO RESTART
MOVX   @DPTR,A

LPRHN3:
CJNE   A,#'L',LPRRST   ; NOT A L CONTINUE

; HANDLE TIME MESSAGE

; SEE IF TIME MESSAGE IS COMPLETE

MOV     DPTR,#LTTDVB+5 ; GET END
MOV     A,DPL
CJNE   A,1CH,LPRRST    ; NOT RIGHT NUMBER OF CHAR.

; MOST SIG TIME DIGIT IS NOT USED AT PRESENT

MOV     DPTR,#LTTDVB+2 ;GET SEC. MOST SIG. TIME DIGIT
MOVX   A,@DPTR
ANL    A,#0FH          ;MAKE BCD
MOV     R6,A           ;SAVE

RL      A                ;MULTIPLY BY 10
RL      A
RL      A
ADD     A,R6
ADD     A,R6

MOV     R6,A            ; SAVE
INC     DPTR            ;GET LEAST SIG DIGIT
MOVX   A,@DPTR
ANL    A,#0FH          ;MAKE BCD

ADD     A,R6            ;COMBINE MSB AND LSB

MOV     DPTR,#VLCRCNT  ; SET IN TIMER
MOVX   @DPTR,A

CLR     A
MOV     DPTR,#TLRCNT   ; CLEAR LORAN COUNTER TO RESTART
MOVX   @DPTR,A

; RESET MESSAGE BUFFER
LPRRST:
MOV     DPTR,#RCVINP   ; CLEAR RECEIVE INPROCESS FLAG
CLR     A
MOVX   @DPTR,A
MOV     DPTR,#LTTDVB   ; RESET POINTER
MOV     R4,DPL
MOV     R6,DPH         ; AT BEGINNING OF BUFFER

LPREND:
MOV     DPTR,#RFMDPT   ;SAVE POINTER
MOV     A,R6
MOVX   @DPTR,A
MOV     A,R4
INC     DPTR
MOVX   @DPTR,A

```

```

POP      1BH      ;SAVE R3
POP      1CH      ;SAVE R4
POP      1DH      ;SAVE R5
POP      1EH      ;SAVE R6

RET      ;EXIT INTERRUPT HANDLER

```

;
END

\$TITLE (PRINTER 8052 BASED)
NAME PRINTER

;ORIGIONAL RELEASE 01 FEB 90
;LAST CHANGE 06 FEB 90 15:13

;Basic Description of Module
; This module handles the message monitoring to the printer or whatever
; type of monitor is being used

;
; *****
; *****
; * * * * *
; * * PUBLICS * * * * *
; * * * * *
; *****
; *****
;

PUBLIC PRTCOM ;Input to this module
PUBLIC PRTIAL ;Initialization Routine
PUBLIC PRTTIN ;Interrupt Module

\$EJECT

;
; *****
; *****
; * * * * *
; * * EXTERNALS * * * * *
; * * * * *
; *****
; *****
;

EXTRN NUMBER (PRTSTS,PRTDAT) ;ADDRESS AND DATA FOR PRINTER USART
EXTRN NUMBER (PRTMIF) ; PRINTER MODE INSTRUCTION FORMAT
EXTRN NUMBER (PRTCMP) ; COMPARE VALUE FOR SELFTEST
EXTRN NUMBER (TRSRES,TRSDIS) ; PRINTER USART CONTROL
EXTRN NUMBER (TRANON) ; RESET TRANSMIT
EXTRN XDATA (PRTTCT) ; INTERRUPT COUNTER

EXTRN NUMBER (PRTVAL) ;VALUE FOR ENABLE OF PRINTER
EXTRN XDATA (PRTFLG)

;
; *****
; *****
; * * * * *
; * * DATA * * * * *
; * * * * *
; *****
; *****
;

DATA_AREA segment XDATA
PROG_S segment CODE
RSEG DATA_AREA

XTODPO: DS 2 ;OUTPUT POINTER FOR BUFFER
XTODPI: DS 2 ;INPUT POINTER FOR BUFFER

XTODBF: DS 500 ;DATA BUFFER FOR BUFFER
XTDEND: DS 1 ;END OF BUFFER

PRTINP: DS 1 ;PRINTER FLAG

```

; *****
; *****
; * * * * *
; * *   P R O G R A M   * *
; * * * * *
; *****
; *****
;
;

```

RSEG PROG_S

```

; This routine is called only by LAPHAN when called saves all
; microprocessor pointers acc etc. and takes the value that is in
; the acc. and outputs it serialy.

```

PRTCOM:

```

PUSH    PSW          ;SAVE PSW
PUSH    DPH          ;SAVE DPTR
PUSH    DPL          ;
PUSH    ACC          ;SAVE ACC

MOV     DPTR,#PRTFLG ;PRINTER ENABLED
MOVX   A,@DPTR
CJNE   A,#PRTVAL,PRTCME ; NO EXIT

POP     ACC          ;RESTORE ACC
PUSH    ACC

PUSH    1CH          ;R4
PUSH    1DH          ;R5
PUSH    1FH          ;R7
PUSH    1EH          ;R6

MOV     R5,A         ;SAVE DATA

; PUT IN BUFFER AT POINT POINTED TO BY XTODPI

MOV     DPTR,#XTDEND ;GET END OF BUFFER
MOV     R4,DPL       ; AND SAVE
MOV     R6,DPH

MOV     DPTR,#XTODPI ;PUT IN BUFFER AT XTODPI
MOVX   A,@DPTR      ;GET MSB
MOV     R7,A         ;SAVE
INC     DPTR         ;
MOVX   A,@DPTR      ;GET LSB
MOV     DPL,A       ;AND SAVE
MOV     DPH,R7      ;SAVE MSB

NOCONB:
MOV     A,R5         ;GET DATA
MOVX   @DPTR,A      ;SAVE DATA IN BUFFER

; RESET POINTER

INC     DPTR         ;INC POINTER
MOV     A,DPL       ;AT END OF BUFFER
CJNE   A,1CH,COM1  ; NOT AT END CONTINUE

MOV     A,DPH       ;NOW CHECK MSB
CJNE   A,1EH,COM1

MOV     DPTR,#XTODBF ;AT END RESET

COM1:
MOV     R5,DPL      ;GET LSB
MOV     A,DPH      ;GET MSB
MOV     DPTR,#XTODPI ;IN COM IN POINTER
MOVX   @DPTR,A     ;SAVE MSB
INC     DPTR       ;
MOV     A,R5       ;SAVE LSB
MOVX   @DPTR,A     ;

MOV     DPTR,#PRTINP ;TRANSMIT IN PROCESS?

```


\$EJECT

```

; *****
; *****
; * * * * *
; * *   INTERRUPT   HANDLER   * *
; * * * * *
; *****
; *****
; THIS INTERRUPT HANDLER HANDLES THE OUTPUTS TO THE DATA CONTROL HEAD
; AS WELL AS A JUMP FOR THE FIRST OUTPUT.
; A ROTATING BUFFER IS USED WITH THE INPUT TO THE BUFFER POINTED TO BY
; XTODPI AND THE OUTPUT POINTED TO BY XTODPO. WHEN THE LSB OF THE OUTPUT
; POINTER IS POINTING TO THE SAME POINT AS THE INPUT POINTER THE NOTHING
; IS TRANSMITTED AND THE INTERRUPT IS DISABLED.

```

PRTTIN:

```

        PUSH    1BH                ;SAVE R3
        PUSH    1CH                ;SAVE R4
        PUSH    1DH                ;SAVE R5
        PUSH    1EH                ;SAVE R6
        PUSH    1FH                ;SAVE R7

; INC INTERRUPT COUNT

        MOV     DPTR,#PRTTCT        ;INC COUNT
        MOVX   A,@DPTR             ;FOR INTERRUPTS
        INC    A
        MOVX   @DPTR,A

; CHECK TO SEE IF INPUT EQUALS OUTPUT POINTER

        MOV     DPTR,#XTDEND        ;GET END
        MOV     R7,DPL              ; OF BUFFER
        MOV     R6,DPH

        MOV     DPTR,#XTODPI        ;GET INPUT POINTER
        MOVX   A,@DPTR             ;MSB
        MOV     R3,A
        INC    DPTR                 ;LSB
        MOVX   A,@DPTR             ;
        MOV     R4,A                ;SAVE
        MOV     DPTR,#XTODPO        ;GET OUTPUT POINTER
        MOVX   A,@DPTR             ;MSB
        MOV     R5,A                ;SAVE
        INC    DPTR                 ;GET LSB
        MOVX   A,@DPTR             ;

        CJNE   A,1CH,INTE41         ;NOT THE SAME CONTINUE
        MOV     A,R5                ;GET MSB
        CJNE   A,1BH,INTE41

        MOV     A,#TRSDIS
        MOV     DPTR,@PRTSTS        ;DISABEL TRANSMIT FOR USART
        MOVX   @DPTR,A

        MOV     DPTR,#PRTINP        ;CLEAR TRANSMIT IN PROCESS FLAG
        CLR    A
        MOVX   @DPTR,A

        JMP     IN4END              ;GO TO END OF INTERRUPT

INTE41:
        MOV     DPL,A               ;SET POINTER TO OUTPUT
        MOV     DPH,R5              ;
        MOVX   A,@DPTR             ;GET DATA TO OUTPUT
        MOV     R5,A                ; AND SAVE

        INC    DPTR                ;INCREMENT AND SAVE POINTER

```

```

MOV     A,DPL                ;SEE IF AT END OF BUFFER
CJNE   A,1FH,INTE42        ; NO CONTINUE
MOV     A,DPH                ; CHECK MSB
CJNE   A,1EH,INTE42        ; SEE IF DONE

INTE42: MOV     DPTR,#XTODBF    ;YES SAVE START OF BUFFER
        MOV     A,DPH          ;GET MSB
        MOV     R7,DPL        ;GET LSB

        MOV     DPTR,#XTODPO   ;GET OUT POINTER
        MOVX   @DPTR,A        ;SAVE MSB
        INC    DPTR           ;
        MOV     A,R7          ;
        MOVX   @DPTR,A        ;SAVE LSB
COUTDT:                ; OUTPUT DATA
        MOV     DPTR,#PRTDAT   ;GET DATA OUTPUT ADDRESS
        MOV     A,R5          ;
        MOVX   @DPTR,A        ;OUTPUT TO PRINTER

IN4END:
        POP     1FH           ;POP R7
        POP     1EH           ;POP R6
        POP     1DH           ;POP R5
        POP     1CH           ;POP R4
        POP     1BH           ;POP R3

```

RET

END

\$TITLE (RFMDEC 8052 BASED)
NAME RFMDEC

:10 AUG 89 15 :00

:VA1.0 ORIGINAL RELEASE 14 JUN 89

;Basic Description of Module

; This module is to receive data from the DEC Computer
; and save it in the buffer RFMDBF incrementing the counter
; RFMDCT which will indicate the number of messages in que
; when the message is completely received

; This module will initialize the receive interrupt to
; receive a message. After it has completely received a
; message it will set the RFMRFG flag and the message will
; be found in the RFMRBF buffer. If a receive is inprocess
; for a long period of time it will time out and reinitialize
; the receive input. Should the message not be completely
; received before timing out the message will be ignored and
; another message will be able to be received.

; Interrupt Routine: RFMDIN
; Initialization Routine: RFMDIT

\$EJECT

* *
* * PUBLICS * *
* *


```

PUBLIC   RFMDEC           ;Input to this module
PUBLIC   RFMDIN          ;Receive Interrupt Handler
PUBLIC   RFMDIT          ;Initialization called by INIT
$EJECT

```

```

;
;
; *****
; *****
; * * * * *
; * *   E X T E R N A L S   * *
; * * * * *
; *****
; *****

```

```

EXTRN CODE (XTOTER)           ;OUTPUT TO TERMINAL
;
EXTRN XDATA (RFMDBF)         ;BEGINNING OF ROTATING BUFFER
EXTRN XDATA (RFMDPT)         ;DEC INPUT BUFFER POINTER
EXTRN XDATA (RFMDBE)         ;END OF ROTATING BUFFER
EXTRN XDATA (MFMDPT)         ;DEC OUTPUT BUFFER POINTER
EXTRN XDATA (RFMDCT)         ;MESSAGE COUNTER
EXTRN NUMBER (VRFDCT)        ;MESSAGE COUNTER MAX

EXTRN NUMBER (ICW1,OCW1,OCW2) ;ENABLE 8259
EXTRN NUMBER (MASKVL)        ;MASK MEMORY
EXTRN NUMBER (MASK)          ;MASK VALUE
EXTRN NUMBER (MASKE)         ;MASK VALUE FOR EXECLP

EXTRN NUMBER (CR)            ;CARRIAGE RETURN
EXTRN XDATA (RDCCNT)         ;INTERRUPT COUNTER

EXTRN NUMBER (SDCSTS,SDCDAT) ;USART STATUS AND DATA
EXTRN XDATA (RDDCMN)         ;MEMORY FOR USART COMMANDS
EXTRN NUMBER (RECVON,RECVOF) ;COMMAND INSTRUCTIONS
EXTRN NUMBER (TRANON,TRANOF) ;FOR RECEIVE AND TRANSMIT
EXTRN NUMBER (RXRDY)         ;RECEIVE READY
EXTRN NUMBER (TXRDY)         ;TRANSMIT READY

EXTRN XDATA (TRFDTO)         ;RECEIVE TIMEOUT TIMER
EXTRN XDATA (TDUART)         ;UART REFRESH TIMER

EXTRN XDATA (DTRINP)         ;TRANS. TO DEC IN PROCESS

EXTRN XDATA (ASCCON)         ;ASCII CONVERSION

EXTRN NUMBER (ASCFLG)        ;CONVERT TO ASCII

```

```

;
$EJECT
;
; *****
; *****
; * * * * *
; * *   D A T A   * *
; * * * * *
; *****
; *****

```

;THIS IS DATA USED INTERNAL TO THIS FILE ONLY

```

DATA_AREA   segment XDATA
RSEG        DATA_AREA

CHARCT:     DS      1      ;CHARACTOR COUNTER FOR BAD MESSGES
SCPDP:      DS      2      ;SCRATCH PAD POINTER FOR LAST GOOD
;MESSAGE
;

```

```

; *****
; *****
; * * * * *
; * * EQUATES * * * * *
; * * * * *
; *****
; *****
;
VRFDTO EQU 12 ;REINITIALIZE AFTER 3 SECONDS
VDUART EQU 8 ;USART REFRESH TIMER
VCHRCT EQU 07H ;CHARACTER COUNTER (NEVER MORE THAN 7)

```

```

; *****
; *****
; * * * * *
; * * PROGRAM * * * * *
; * * * * *
; *****
; *****

```

```

PROG_S segment CODE
RSEG PROG_S

```

RFMDEC:

;THE FOLLOWING IS TO REFRESH BAUD RATE IF NESCESSARY

```

MOV DPTR,#DTRINP ;TRANSMIT IN PROCESS
MOVX A,@DPTR ;
JZ REFREC ; NO GO REFRESH RECEIVE

MOV DPTR,#SDCSTS ;GET STATUS
MOVX A,@DPTR ;TX EMPTY
ANL A,#TXRDY
JNZ REFREC ; NO CONTINUE

MOV DPTR,#RDDCMN ;GET PAST STATUS
MOVX A,@DPTR
ORL A,#TRANON
MOVX @DPTR,A
MOV DPTR,#SDCSTS ;REFRESH RECEIVE
MOVX @DPTR,A ;

JMP REFEND ; CONTINUE

```

REFREC:

;SEE IF NEED TO REFRESH USART

```

MOV DPTR,#TDUART ;TIMER EXPIRED?
MOVX A,@DPTR ;
JNZ REFEND ; NO DO NOT REFRESH
MOV A,#VDUART ; YES REFRESH
MOVX @DPTR,A ;

MOV DPTR,#RDDCMN ;GET PAST STATUS
MOVX A,@DPTR
ORL A,#RECVON
MOVX @DPTR,A
MOV DPTR,#SDCSTS ;REFRESH RECEIVE
MOVX @DPTR,A ;

```

REFEND:

```

MOV DPTR,#TRFDTO ;HAS MESSAGE TIMED OUT
MOVX A,@DPTR ;
JNZ KYEXIT ; NO CONTINUE

MOV DPTR,#SCPDPT+1 ;MESSAGE COMMING IN
MOVX A,@DPTR ;
MOV R4,A ;SAVE LSB OF BUFFER
MOV DPTR,#RFMDPT+1 ;
MOVX A,@DPTR ;GET LSB OF POINTER
CJNE A,1CH,RFMDE1 ; YES CONTINUE
JMP KYEXIT ;NO EXIT

```

RFMDE1:

```
CALL    RFMDIT          ;REINITIALIZE
```

```
KYEXIT:
  RET
;END OF MAIN PROGRAM
```

```
$EJECT
```

```
;
; *****
; *****
; * *
; * *   I N I T I A I Z A T I O N   * *
; * *
; *****
; *****
```

```
; The following is called once by the INIT module.
```

```
RFMDIT:
```

```
;The following sets the input pointer
```

```
MOV    DPTR,#RFMDBF      ; RESET INPUT POINTER
MOV    R4,DPL            ;
MOV    A,DPH            ; TO START OF BUFFER
MOV    R5,A             ;
MOV    DPTR,#RFMDPT     ;
MOVX   @DPTR,A          ;
INC    DPTR             ;
MOV    A,R4             ;
MOVX   @DPTR,A          ;

MOV    DPTR,#MFMDPT     ; RESET OUTPUT POINTER
MOV    A,R5            ; TO START OF BUFFER
MOVX   @DPTR,A          ;
INC    DPTR             ;
MOV    A,R4             ;
MOVX   @DPTR,A          ;

MOV    DPTR,#SCPDPT     ; RESET SCRATCH PAD COUNTER
MOV    A,R5            ; TO START OF BUFFER
MOVX   @DPTR,A          ;
INC    DPTR             ;
MOV    A,R4             ;
MOVX   @DPTR,A          ;

RET                                ;EXIT INITIALIZATION
```

```
$EJECT
```

```
;
; *****
; *****
; * *
; * *   I N T E R R U P T   H A N D L E R   * *
; * *
; *****
; *****
```

```
; This interrupt handler handles the data coming into
; the USART.
```

```
RFMDIN:
```

```
PUSH  1EH                ;SAVE R6
PUSH  1DH                ;SAVE R5
PUSH  1CH                ;SAVE R4
PUSH  1BH                ;SAVE R3
```

```
; INC INTERRUPT COUNT
```

```
MOV    DPTR,#RDCCNT     ;INC COUNT
MOVX   A,@DPTR          ;FOR INTERRUPTS
INC    A
MOVX   @DPTR,A
```

```

RFDIN2:  MOV    DPTR,#TRFDTO      ;SET RECEIVE TIME OUT
          MOV    A,#VRFDDTO      ;TIMER
          MOVX   @DPTR,A

          MOV    DPTR,#SDCDAT     ;READ DATA
          MOVX   A,@DPTR         ;
          MOV    R5,A            ; AND SAVE

          CLR    ASCFLG          ;CLEAR ASCFLG

          MOV    DPTR,#ASCCON     ;CONVERT?
          MOVX   A,@DPTR         ;
          CJNE   A,#'A',NOCONA    ; NO CONTINUE

          MOV    A,R5            ;RETRIEVE DATA

          CALL   ASCBIN           ;CONVERT TO BINARY
          JMP    NOCONB          ; AND CONTINUE
NOCONA:  MOV    A,R5            ;GET DATA

          CJNE   A,#CR,NOCONC     ;CARRIAGE RETURN EXIT
          SETB   ASCFLG          ;SET FOR ASCII

          MOV    A,#'X'          ;OUTPUT X FOR TRANSMIT
          CALL   XTOTER           ; OUTPUT TO TERMINAL
          MOV    A,#' '          ;OUTPUT SPACE
          CALL   XTOTER           ; OUTPUT TO TERMINAL
          CLR    ASCFLG          ;CLEAR ASCFLG
          JMP    NOCONB          ;
NOCONC:  CALL   XTOTER           ; OUTPUT TO TERMINAL
NOCONB:  MOV    DPTR,#RFMDCT      ;CHECK MESSAGE COUNTER
          MOVX   A,@DPTR         ; FOR OVERFLOW
          CJNE   A,#VRFDDCT,RFDIN3 ; NOT OVERFLOW CONTINUE
          JMP    KEYEXT          ; OVERFLOW EXIT

RFDIN3:  MOV    DPTR,#RFMDPT      ;GET MESSAGE POINTER
          MOVX   A,@DPTR         ; MSB
          MOV    R4,A            ;
          INC    DPTR            ; LSB
          MOVX   A,@DPTR         ;
          MOV    DPL,A           ;AND SET IN DPTR
          MOV    DPH,R4          ;
          MOV    A,R5            ;
          MOVX   @DPTR,A         ;SAVE DATA IN BUFFER
          INC    DPTR            ;INCREMENT POINTER
          MOV    A,DPL           ; SAVE POINTER
          MOV    R4,DPL          ;
          MOV    R6,DPH          ;

          MOV    DPTR,#RFMDBE     ;AT END OF BUFFER
          MOV    R3,DPL          ;

          CJNE   A,1BH,RFDIN4    ;

          MOV    DPTR,#RFMDBF     ; YES GET BEGINNING OF BUFFER
          MOV    R4,DPL          ;
          MOV    R6,DPH          ;

RFDIN4:  MOV    A,R5            ;CARRIAGE RETURN
          CJNE   A,#CR,RFDIN6    ; NO CONTINUE

          MOV    DPTR,#CHARCT     ; GET CHARACTOR COUNT
          CLR    A               ; AND CLEAR
          MOVX   @DPTR,A         ;

          MOV    DPTR,#RFMDCT     ;CHECK MESSAGE COUNTER
          MOVX   A,@DPTR         ; FOR OVERFLOW

```

```

CJNE   A,#VRFDCT,RFDN5A      ; AT MAX NO JUMP
JMP    RFDIN7                 ;AT MAX JUMP
RFDN5A:
INC    A                       ; INCREMENT AND SAVE
MOVX   @DPTR,A                ;

MOV    DPTR,#SCPDPT           ;RESET SCRATCH PAD POINTER
MOV    A,R6                    ;
MOVX   @DPTR,A                ;
INC    DPTR                    ;
MOV    A,R4                    ;
MOVX   @DPTR,A                ;

JMP    RFMEND                  ;YES RESET POINTER
RFDIN6:
MOV    DPTR,#CHARCT           ; GET CHARACTOR COUNT
MOVX   A,@DPTR                ;
INC    A                       ; INCREMENT
CJNE   A,#VCHRCT,RFMEND       ; NOT AT MAX EXIT
CLR    A                       ;
MOVX   @DPTR,A                ;

RFDIN7:
MOV    DPTR,#SCPDPT           ;RESET ROTATING BUFFER
MOVX   A,@DPTR                ;
MOV    R6,A                    ;
INC    DPTR                    ; TO LAST KNOWN GOOD MESSAGE
MOVX   A,@DPTR                ;
MOV    R4,A                    ;

RFMEND:
MOV    DPTR,#RFMDPT           ;SAVE POINTER
MOV    A,R6                    ;
MOVX   @DPTR,A                ;
MOV    A,R4                    ;
INC    DPTR                    ;
MOVX   @DPTR,A                ;

KEYEXT:
MOV    DPTR,#ICW1             ;RESET
MOV    A,#1FH                 ;ICW1
MOVX   @DPTR,A                ;

MOV    DPTR,#OCW1             ;UNMASK
MOV    A,#0H                   ;
MOVX   @DPTR,A                ;

MOV    DPTR,#OCW1             ;UNMASK
MOV    A,#80H                  ;
MOVX   @DPTR,A                ;

MOV    A,MASKVL               ;SET INTERRUPT MASK
ORL    A,MASK                  ;
ORL    A,MASKE                 ;

MOV    DPTR,#OCW1             ;RESET MASK
MOVX   @DPTR,A                ;

POP    1BH                     ;SAVE R3
POP    1CH                     ;SAVE R4
POP    1DH                     ;SAVE R5
POP    1EH                     ;SAVE R6
POP    ACC                    ;
POP    DPL                    ;
POP    DPH                    ;
POP    PSW                    ;PUSHED ON INTERRUPT

SETB   EA                     ;ENABLE INTERRUPTS

RETI

```

```

;SUBROUTINE
; THIS SUBROUTINE CONVERTS FROM ASCII TO BINARY

```


ASCTAB:

```

DB 00H
DB 01H
DB 02H
DB 03H
DB 04H
DB 05H
DB 06H
DB 07H
DB 08H
DB 09H
DB OFFH
DB OFFH
DB OFFH
DB OFFH
DB OFFH
DB OFFH
DB OFFH
DB 0AH
DB 0BH
DB 0CH
DB 0DH
DB 0EH
DB 0FH
DB 00H

```

ASCBIN:

```

SETB  ASCFLG           ;SET FOR ASCII
CJNE  A,#CR,ASCBNO    ;CARRIAGE RETURN EXIT

MOV   A,#'X'          ;OUTPUT X FOR TRANSMIT
CALL  XTOTER          ;OUTPUT TO TERMINAL
MOV   A,#' '          ;OUTPUT SPACE
CALL  XTOTER          ;OUTPUT TO TERMINAL
JMP   ASCEXT          ;

```

ASCBNO:

```

CALL  XTOTER          ;OUTPUT TO CONSOLE

MOV   DPTR,#ASCTAB

CLR   C               ;GET BINARY CHARACTOR
SUBB  A,#'0'
MOVC  A,@A+DPTR       ;

RR    A
RR    A               ;GET AS MSB
RR    A
RR    A

PUSH  ACC

```

ASCBN1:

```

MOV   DPTR,#SDCSTS    ;RECEIVE BUFFER FULL
MOVX  A,@DPTR
ANL   A,#RXRDY
JZ    ASCBN1

MOV   DPTR,#SDCDAT    ;GET DATA
MOVX  A,@DPTR
CALL  XTOTER          ;OUTPUT TO CONSOLE

MOV   DPTR,#ASCTAB
CLR   C               ;GET BINARY CHARACTOR
SUBB  A,#'0'
MOVC  A,@A+DPTR       ;

POP   1DH

ORL   A,R5             ;COMBINE
MOV   R5,A            ;SAVE

```

ASCEXT:

```

CLR   ASCFLG          ;CLEAR ASCII FLAG

```

RET

;
ENDA:\APPENDA
D:\jnh\116601.APP

United States Patent Application

of

Edward J. Fraughton

and

Philip H. Berger

for

A UNIVERSAL DYNAMIC NAVIGATION,
SURVEILLANCE, EMERGENCY LOCATION, AND
COLLISION AVOIDANCE SYSTEM AND METHODAPPENDIX B
PROGRAMMING CODE

Copyright 1990 TERRASTARR, INC.

```

/*                POSITION.C                */
/*                */
/*                Collision Avoidance System Software                */
/*                Copyright 1990 by TERRASTARR Corporation                */
/*                */
/* <900116.1646>   Basic display system                */
/* <900205.1250>   Interrupt driven serial input                */
/* <900208.2326>   Conditional compilation for input from comm port/disk */
/* <900215.1006>   Error checking and realignment                */
/* <900220.0228>   Fixed heading error on plot                */
/* <900221.0038>   Added variable scale factor                */
/* <900301.1233>   Changed scale factor, increased range                */
/* <900307.1337>   Changed float variables to double                */
/* <900312.1005>   Added check for repeat loran position                */
/* <900319.2357>   Added bitmapped airplanes                */
/* <900329.0810>   Added altitude readout                */
/* <900405.1201>   Added "operator" controlled character plotting                */
/* <900521.1040>   Fixed floating point termination error                */
/* <900610.1334>   Fixed altitude position, added averaging (lookback var) */
/* <900611.0937>   Hdg arrow, hdg rounding                */
/* <900617.1228>   Added data logging                */
/* <900617.1229>   Last revision date                */
/*-----*/
/* INCLUDES                */
#include <stdio.h>
#include <time.h>
#include <conio.h>
#include <math.h>
#include <string.h>
#include <stdlib.h>
#include <ctype.h>
#include "colors.h"
#include "gtools.h"
#include "pgdraws.h"
#include "ghead.h"
#include "xc.h"
#include "fkeys.h"
extern long int xc_test();
/*-----*/

```

```

/* DEFINES
#define true 1
#define false 0
#define NACFT 10 /* number of aircraft */
#define NDATA 10 /* number of data per aircraft */
#define PORT COM1 /* I/O port to data-V-com */
#define RATE BAUD1200 /* I/O data rate to data-V-com */
#define COMP 1 /* 1=data from file, 2=commport */
#define NOGRAPH 0 /* 1=graph off, -1 rec count */
#define LOOKBACK 5 /* number of points to "average" */
/*-----*/
/* FUNCTIONS
void initialize(); /* Initialize system */
void plotem(); /* data plotting function */
int data_in(); /* read data from file / port */
void process_data(); /* parse data */
int search(); /* Search table for aircraft ID */
int find_empty(); /* Find slot for new aircraft */
void clear_entry(); /* release ID entry */
float dist(); /* calc dist & abs bearing */
float relative(); /* function to calc rel bearing */
void distance_circles(); /* plot distance circles */
void center_airplane(); /* plot airplane in center */
void airplane(); /* plot target airplane */
int clip(); /* keep graphics within bounds */
void plot_airplane(); /* bit mapped airplane plot */
void plot_previous_large(); /* plot prev position as a dot */
void plot_previous_small(); /* plot prev position as a dot */
void update(); /* replt all acft when center moves*/
int check(); /* verify number is within 0..9 */
void check_keyboard(); /* keyboard input & dispatch */
int toggle(); /* toggle a variable true/false */
void zoom(); /* zoom screen in/out */
int validate(); /* check for brackets * checksum */
void checksum(); /* calculate checksum of a string */
void align(); /* read new data until data aligns*/
void open_port(); /* open up communications port */
int read_buffer(); /* interrupt serial port read */
void terminate(); /* end the program gracefully */
void open_file();
void open_save_file();
void read_rec();
void plot_read_string();
/*-----*/
/* DIMENSIONED GLOBAL VARIABLES
double latitude [NACFT][NDATA]; /* storage for latitude */
double longitude [NACFT][NDATA]; /* storage for longitude */
float heading [NACFT][NDATA]; /* storage for heading */
float speed [NACFT][NDATA]; /* storage for speed */
float v_speed [NACFT][NDATA]; /* storage for v speed */
float distance [NACFT][NDATA]; /* storage for distance */
float rel_bearing [NACFT][NDATA]; /* storage for relative bearing */
float rel_heading [NACFT][NDATA]; /* storage for relative heading */
long sample_time [NACFT][NDATA]; /* time data was received */
int altitude [NACFT][NDATA]; /* storage for altitude */
int status [NACFT][NDATA]; /* loran / xmitter status */
int data_pointer [NACFT]; /* data pointer for each aircraft */
int plotted [NACFT][NDATA]; /* indicates data plotted */
char nnumber [NACFT][7]; /* aircraft identifier */
char read_string [40]; /* character input variable */

/* SINGLE GLOBAL VARIABLES
double mylat, mylon, hislat, hislon; /* current variables */
float myheading, myspeed;
long t_start, mytime, histime; /* time variables */
int myalt, hisalt;
int file_record_count; /* reset counter for data save */

/* GLOBAL CONSTANTS
double pi = 3.141592654;
double c = .01745329252; /* pi/180. */
float circle_ratio = 1.0; /* make circles round */
float scale = .25; /* set scale to 4 mi / circle */

```

```

int      radius = 50;          /* radius increment of rings */
int      myplane=LIGHTBLUE;   /* aircraft color */
int      hisplane=YELLOW;     /* aircraft color */
int      bogey=LIGHTMAGENTA;  /* aircraft color */
int      ctr = 0;             /* debugging counter */
int      records_read = 0;    /* record counter */

/* GLOBAL FLAGS */

int      vector_flag = false; /* turn on/off velocity vectors */
int      autoread_flag = false; /* turn on/off auto disk read */
int      altitude_flag = false; /* turn on/off altitude display */
int      nnumber_flag = false; /* turn on/off nnumber display */
int      previous_flag = false; /* turn on/off previous position */
int      hold_rewrite = false; /* flag to hold off clearing image */
int      print_string_flag = false; /* print-plot input string */
int      data_save_flag = false; /* save data to file */

/* FILE CONTROLS */
FILE     *save_file;

/* CONDITIONAL VARIABLES */
#if COMP==1
FILE     *fp;
#endif
/*-----*/
main()
{
    initialize();
    open_save_file();
    adapt = getAdapter();
    save_mode = getMode(&ncols);
    LINEWIDTH = 1;
#if NOGRAPH < 1
    mode = 0x12;
    setMode(mode);
    cls(BLACK);
    distance_circles();
#endif
loop:
#if COMP == 2
    if (data_in())
        process_data();
#endif
#if NOGRAPH < 1
    plotem();
#endif
    check_keyboard();
    goto loop;
}
/*-----*/

void initialize()
    /* Routine to initialize arrays: puts null in nnumber */
    /* string, sets data_pointer to -1, sets plotted flag */
    /* positive for all data points, puts 0's in position */
    /* array, sets up com port, inits Data-V-Com, gets my */
    /* nnumber, puts it into the 0 position */
{
    register int    i, j, k;
    static int     try = 0;
    int            two = 2, three = 3, five = 5, six = 6;
    int            err;
    char           dv_status[2], buffer[81];

    buffer[35] = '\0';
    vector_flag = false;
    altitude_flag = false;
    time(&t_start);
    /* make sure we don't run away */
    /* turn off velocity vector plot */
    /* turn off altitude vector plot */
    /* get reference time */

    for (i=0; i<NACFT; i++)
        /* clear all arrays and set */
        /* pointers to -1 (empty condition*/
        {
            nnumber[i][0] = '\0';
            data_pointer[i] = -1;
        }
}

```

```

for (j=0;j<NDATA;j++)
{
    latitude    [i][j] = 9999.;
    longitude   [i][j] = 9999.;
    altitude    [i][j] = 0;
    sample_time [i][j] = 0;
    heading     [i][j] = 0.;
    speed       [i][j] = 0.;
    v_speed     [i][j] = 0.;
    distance    [i][j] = 0.;
    rel_bearing [i][j] = 0.;
    rel_heading [i][j] = 0.;
    status      [i][j] = 0;
    plotted     [i][j] = true;
}
}
#endif
#if COMP == 1
open_file();
read_rec();
i = strchr(read_string,"#");
if (i != 1)
    terminate;
else
{
    strncpy(nnumber[0],&read_string[2],6);
    nnumber[0][7] = '\0';
}
#endif
#if COMP == 2
open_port();
try = 0;
if (++try < 5)
{
    xc_put(PORT,"R\r\n",&three);           /* initialize Data-V-Com */
    do
    {
        i = read_buffer(buffer,35);
        } while (i == 0);
        if (!(err = validate(buffer)))
        {
            i = strchr(buffer,"#");
            if (i == 1)
                goto ok;
        }
    }
}
else
    terminate();
ok:
memmove(nnumber[0],&buffer[2],6);          /* grab my ID */
nnumber[0][6] = '\0';                      /* terminate the line */
memmove(dv_status,&buffer[8],2);           /* save Data-V-Com status */
xc_put(PORT,"L003\r\n",&six);             /* signal ready to receive loran */
xc_put(PORT,"T004\r\n",&six);             /* signal start transmit position */
xc_put(PORT,"O001\r\n",&six);             /* receive other signals */
#endif
}

```

```

/*-----*/

```

```

void plotem()
{
    register int    i, j;

    for (i=0;i<NACFT;i++)
    {
        j = data_pointer[i];                /* anybody home ? */
        if (j != -1)
        {
            if (!plotted[i][j])            /* has it already been plotted */
            {
                if (i==0)
                {

```

```

        center_airplane(j);
        plotted[i][j] = true;
        update();
    }
    else if (strcmp(nnumber[i],"") && i!=0)
    {
        color = hisplane;
        OPERATOR = XOR;
        if (!hold_rewrite)
            airplane (i,j-1);
        airplane (i,j);
    /*      plot_previous(rel_bearing[i][j-1],distance[i][j-1]); */
    }
    }
}
hold_rewrite = false;
}
/*-----*/

int data_in()
/* Routine to accept input data from file or */
/* I/O port */
{
    int j, err;
#ifdef COMP==1
    read_rec();
    if (feof(fp))
        terminate();
    records_read++;
    read_string[35]='\0';
    #if NOGRAPH == 1
        printf("%s\n",read_string);
    #else
        if (print_string_flag)
            plot_read_string();
    #endif
    if (data_save_flag)
    {
        fprintf(save_file,"%s\n",read_string);
        file_record_count++;
        if (file_record_count>20)
        {
            fclose(save_file);
            open_save_file();
        }
    }
    return 1;
#endif

#ifdef COMP == 2
    j = read_buffer(read_string,35);
    if (j)
    {
        records_read++;
        err = validate(read_string);
    /*      printf("err = %i\n",err);*/
        while (err)
        {
            align(read_string);
            check_keyboard;
        }
        read_string[35]='\0';
    #if NOGRAPH == 1
        printf("%s\n",read_string);
    #else
        if (print_string_flag)
            plot_read_string();
        if (data_save_flag)
        {

```

```

    fprintf(save_file,"%s\n",read_string);
    file_record_Count++;
    if (file_record_count>20)
    {
        fclose(save_file);
        open_save_file();
    }
}
#endif
}
return j;
#endif
}
/*-----*/

void process_data()
/* Routine to parse input data, and calculate */
/* heading,speed,verticle speed, distance and */
/* relative bearing */
{
    double temp_num, intpart, fraction;
    float direction, length;
    int entry, j, previous, err_status;
    long tnow, deltetime;
    char temp_char[10], id[7];

    memmove (temp_char,&read_string[25],2); /* get error status */
    temp_char[2] = '\0';
    err_status = atoi(temp_char);
    if (!(err_status & 2)) /* if its not new data, ignore it */
    {
        mylat = latitude[0][data_pointer[0]]; /* get */
        mylon = longitude[0][data_pointer[0]]; /* current */

        myalt = altitude[0][data_pointer[0]]; /* data */
        myheading = heading[0][data_pointer[0]]; /* for my */
        myspeed = speed[0][data_pointer[0]]; /* plane */
        memmove(id,&read_string[1],6); /* peel off aircraft id */
        id[6] = '\0';
        entry = search(id); /* insert into data base */
        if (entry<0)
        {
            entry = find_empty();
            memmove(nnumber[entry],id,6);
        }
        data_pointer[entry] = j = check(data_pointer[entry] + 1);
        status[entry][j] = err_status;
        memmove (temp_char,&read_string[8],6); /* get latitude */
        temp_char[6] = '\0'; /* */
        temp_num = atof(temp_char) * .0001; /* */
        fraction = modf(temp_num,&intpart); /* */
        latitude[entry][j] = hislat = intpart + fraction / .6; /* */
        memmove (temp_char,&read_string[15],7); /* get longitude */
        temp_char[7] = '\0'; /* */
        temp_num = atof(temp_char) * .0001; /* */
        fraction = modf(temp_num,&intpart); /* */
        longitude[entry][j] = hislon = intpart + fraction / .6; /* */
        memmove (temp_char,&read_string[22],3); /* get altitude */
        temp_char[3] = '\0'; /* */
        altitude[entry][j] = hisalt = atoi(temp_char); /* */
        sample_time[entry][j] = histime = time(&tnow) - t_start; /* time */
        distance[entry][j] = dist(mylat,mylon,hislat,hislon,
            &direction); /* distance */
        rel_bearing[entry][j] = relative(myheading,direction);

        previous = check(j - LOOKBACK); /* old data ptr */
        length = dist(latitude[entry][previous],longitude
            [entry][previous],hislat,hislon,&direction); /* how far */
        heading[entry][j] = direction; /* heading */
        rel_heading[entry][j] = relative(myheading,direction); /* relative hdg */
        deltetime = histime - sample_time[entry][previous];
        if (deltetime)

```

```

    {
        speed[entry][j] = length / deltatime;          /* speed */
        v_speed[entry][j] = (hisalt - myalt) / deltatime; /* vert speed */
    }
    else
    {
        speed[entry][j] = 0.;                          /* no time */
        v_speed[entry][j] = 0.;                        /* no speed */
    }
    plotted[entry][j] = false;
}
}
/*-----*/

```

```

int search(char *ident)
/* Search table of nnumbers for an aircraft
/* identifier which matches ident. Return the
/* entry number of matching ident, or -1 if fail */
{
    register int i;
    for(i=0;i<NACFT;i++)
    {
        if(!strnicmp(nnumber[i],ident,6)) return i;
    }
    return -1;
}
/* id not found */
/*-----*/

```

```

int find_empty()
/* Routine to find empty slot for new aircraft
/* depends on data_pointer being set to -1 for
/* empty slots
{
    register int i;
    for (i=1;i<NACFT;i++)
        if (data_pointer[i] == -1)
            return i;
    else
        return -1;
}
/* only way this can happen is if array is full*/
/*-----*/

```

```

void clear_entry(i)
register int i;
{
    register int j;

    data_pointer[i] = -1;
    nnumber[i][0] = '\0';
    for (j=0;j<NDATA;j++)
        plotted[i][j] = true;
}
/*-----*/

```

```

float dist(double mlat, double mlon, double hlat, double hlon,
float *bearing)
/* Routine to calculate the distance and bearing
/* to a target, given latitudes and longitudes of
/* my aircraft and the target aircraft
{
    double deltalat, deltalong, avelat;
    float length;

    deltalat = (hlat - mlat) * 60.;
    avelat = (hlat + mlat) / 2.;
    deltalong = (mlon - hlon) * cos (avelat * c) * 60.;
    length = sqrt (deltalat * deltalat + deltalong * deltalong);
    if (deltalong)
    {
        *bearing = 90. - atan (deltalat / deltalong) / c;
        if (deltalong < 0.) *bearing = *bearing + 180.;
    }
}

```



```

    }
    else
    {
        if (deltalat < 0.)
            *bearing = 180.;
        else
            *bearing = 360.;
    }
    return length;
}
/*-----*/
float relative(hdg, abs_heading)
/* this routine calculates relative bearing to */
/* target, given my heading and an absolute */
/* bearing to a target */
float    hdg, abs_heading;
{
    float    rel;

    rel = 360. - hdg + abs_heading;
    if (rel>360.) rel -= 360.;
    return rel;
}
/*-----*/
void distance_circles()
/* draws distance circles on display */
{
    char    disp_buffer[30];
    int     i;

    LINEWIDTH = 1;
    for (i=1;i<5;i++)
        drawOval(0,0,i*radius,DARKGRAY, circle_ratio); /* draw circles */
    gotoxy(60,29);
    sprintf(disp_buffer,"%3.3g MILES / CIRCLE\0",1.0/scale);
    writString(disp_buffer,LIGHTGREEN,0);
    drawLine(0,239,15,230,LIGHTCYAN);
    drawLine(15,230,15,220,LIGHTCYAN); /* draw heading */
    drawLine(15,220,-15,220,LIGHTCYAN); /* arrow */
    drawLine(-15,220,-15,230,LIGHTCYAN); /* box */
    drawLine(-15,230,0,239,LIGHTCYAN); /*
}
/*-----*/
void center_airplane(int j)
/* Draws an airplane at (0,0), puts heading */
/* at top of screen */
{
    float    hdg;
    static char    disp_heading[5], disp_altitude[5], disp_count[5];
    static int    y = 0;
    int    x;

    hdg = myheading;
    LINEWIDTH = 1;
    OPERATOR = REPLACE;
    plot_airplane(0,0,LIGHTBLUE,0);
    plot_horz_str(-11,218,disp_heading,BLACK);
    if (hdg < 0)
        hdg += 360.; /* erase heading */
    sprintf(disp_heading,"%3.3i", (int)(hdg+.5)); /* convert */
    disp_heading[3] = '\0'; /* heading */
    plot_horz_str(-11,218,disp_heading,LIGHTCYAN); /* to */
    plot_horz_str(10,-25,disp_altitude,BLACK); /* string */
    drawLine(0,0,0,y,BLACK); /* plot new heading value*/
    if (vector_flag) /* erase altitude */
    { /* erase vector */
        x = 0;
        y = myspeed * 60. * 50. * scale;
    }
}

```

```

clip(x,y);
drawLine(0,0,0,y,LIGHTBLUE);
}

if (altitude_flag)
{
    sprintf(dispatch_altitude,"%3.3i", (int)myalt);
    dispatch_altitude[3] = '\0';
    plot_horz_str(10,-25,dispatch_altitude,LIGHTBLUE);
}
#endif NOGRAPH < 0
plot_horz_str(290,-226,dispatch_count,BLACK);
sprintf(dispatch_count,"%4.4i",records_read);
dispatch_count[4] = '\0';
plot_horz_str(290,-226,dispatch_count,MAGENTA);
#endif
}
/*-----*/

void airplane(int j, int k)
/* Routine to plot an airplane symbol with */
/* appropriate heading */
{
    float    length, temp, angle;
    int      x, y, x3, y3, xalt, yalt, picture;
    char     disp_alt[10];

    k = check(k);
    temp = rel_heading[j][k];
    LINEWIDTH = 1;
    x = radius * scale * distance[j][k] * cos ((450. - rel_bearing[j][k]) * c);
    y = radius * scale * distance[j][k] * sin ((450. - rel_bearing[j][k]) * c);
    if (!clip(&x,&y))
    {
        temp = temp + 11.25;
        if (temp > 360.)
            temp -= 360.;
        picture = (int) (temp / 22.5);
        plot_airplane(x,y,color,picture);
        if (vector_flag)
        {
            length = speed[j][k] * 60. * 50. * scale;
            angle = (90.0 - rel_heading[j][k]) * c;
            x3 = x + length * cos (angle);
            y3 = y + length * sin (angle);
            clip(&x3,&y3);
            drawLine(x,y,x3,y3,color);
        }
        if (altitude_flag)
        {
            sprintf(dispatch_alt,"%3.3i", (int)altitude[j][k]);
            dispatch_alt[3] = '\0';
            xalt = x + 10;
            yalt = y - 25.;
            plot_horz_str(xalt,yalt,dispatch_alt,color);
        }
    }
    plotted[j][k] = true;
}
/*-----*/

int clip(x,y)
int *x, *y;
{
    int    clipped = false;

    if (*x < -320)
    {
        *x = -320;
        clipped = true;
    }
    if (*x > 320)

```

```

{
    *x = 320;
    clipped = true;
}
if (*y < -240)
{
    *y = -240;
    clipped = true;
}
if (*y > 240)
{
    *y = 240;
    clipped = true;
}
return clipped;
}
/*-----*/

void plot_airplane(int x, int y, int color, int angle)
{
    unsigned int    mask;
    int             row, column;
    static unsigned int pattern[16][16] = {
        0x0000,0x0000,0x0000,0x0180,0x0180,0x03c0,0xffff,0xffff,0x3ffc,0x03c0,
        0x0180,0x0180,0x0180,0x0180,0x07e0,0x07e0, /* pattern 0 */
        0x0000,0x0000,0x0000,0x0000,0x6060,0x78e0,0x3ee0,0x1fc0,0x0ff0,0x07fc,0x03ff,
        0x071f,0x2600,0x7e00,0x3c00,0x0f00,0x0700, /* pattern 1 */
        0x0000,0x3000,0x3800,0x1c00,0x1e60,0x0fe0,0x0fc0,0x07c0,0x07e0,0x4ff0,
        0xfcfc,0x783c,0x380c,0x1c00,0x0800,0x0000, /* pattern 2 */
        0x0400,0x0e00,0x0e00,0x0700,0x0700,0x07b8,0x03f8,0x03f0,0xc7c0,0xdfe0,
        0xfde0,0x70f0,0x3070,0x3878,0x1038,0x0010, /* pattern 3 */
        0x00c0,0x00c0,0x01c0,0x01c0,0x01c0,0xc1c0,0xc3e0,0xffff,0xffff,0xc3e0,
        0xc1c0,0x01c0,0x01c0,0x01c0,0x00c0,0x00c0, /* pattern 4 */
        0x0000,0x1018,0x3838,0x3070,0x70f0,0xfde0,0xdfe0,0xc7c0,0x03f0,0x03f8,
        0x03b8,0x0780,0x0700,0x0700,0x0600,0x0600, /* pattern 5 */
        0x0400,0x0e00,0x1c06,0x3c1e,0x7e7c,0x27f8,0x03f0,0x03e0,0x07e0,0x07f0,
        0x0f30,0x0e00,0x1c00,0x1800,0x0000,0x0000, /* pattern 6 */
        0x0700,0x0f00,0x3c00,0x7e00,0x2600,0x077f,0x03ff,0x07fc,0x0ff0,0x1fc0,
        0x3ee0,0x78e0,0x6060,0x0000,0x0000,0x0000, /* pattern 7 */
        0x07e0,0x07e0,0x0180,0x0180,0x0180,0x0180,0x03c0,0x3ffc,0xffff,0xffff,
        0x03c0,0x0180,0x0180,0x0000,0x0000,0x0000, /* pattern 8 */
        0x00e0,0x00f0,0x003c,0x007e,0x0064,0xfce0,0xffc0,0x3fe0,0x0ff0,0x03f8,
        0x077c,0x071e,0x0606,0x0000,0x0000,0x0000, /* pattern 9 */
        0x0000,0x0010,0x0038,0x301c,0x3c1e,0x1f3f,0x0ff2,0x07e0,0x03e0,0x03f0,
        0x07f0,0x0678,0x0038,0x001c,0x000c,0x0000, /* pattern 10 */
        0x0000,0x1800,0x1c1c,0x0e0c,0x0f0e,0x07bf,0x07fb,0x03e3,0x0fc0,0x1fe0,
        0x1de0,0x01e0,0x00e0,0x00e0,0x00e0,0x0060,0x0060, /* pattern 11 */
        0x0300,0x0300,0x0380,0x0380,0x0380,0x0383,0x07c3,0x1fff,0x1fff,0x07c3,
        0x0383,0x0380,0x0380,0x0380,0x0300,0x0300, /* pattern 12 */
        0x0060,0x0060,0x00e0,0x00e0,0x01e0,0x1de0,0x1fe0,0x0fc0,0x03e3,0x07fb,
        0x07bf,0x0f0e,0x0e0c,0x1c1c,0x1808,0x0000, /* pattern 13 */
        0x0000,0x0000,0x0018,0x0038,0x0070,0x0cf0,0x0fe0,0x07e0,0x07c0,0x0fc0,
        0x1fe4,0x3e7e,0x783c,0x6038,0x0070,0x0020, /* pattern 14 */
        0x0000,0x0000,0x0000,0x0606,0x071e,0x077c,0x03f8,0x0ff0,0x3fe0,0xffc0,
        0xfce0,0x064,0x007e,0x003c,0x00f0,0x00e0 /* pattern 15 */
    };

    for (row = 0; row < 16 ; row++)
    {
        for (column = 0; column < 16; column++)
        {
            mask = 0x8000 >> column;
            if (pattern[angle][row] & mask)
                plots(x-8+column,y+8-row,color);
        }
    }
}
/*-----*/

void plot_previous_large(float bearing, float dist)
/* A routine to plot a large dot in the previous */
/* position of an aircraft */

```

```

{
  int  x, y ;

  x = radius * dist * cos ((450. - bearing) * c);
  y = radius * dist * sin ((450. - bearing) * c);
  fillOval(x,y,2,color,circle_ratio) ;
}
/*-----*/

void plot_previous_small(float bearing, float dist)
/* A routine to plot a small dot in the previous */
/* position of an aircraft */
{
  int  x, y ;

  x = radius * dist * cos ((450. - bearing) * c);
  y = radius * dist * sin ((450. - bearing) * c);
  fillOval(x,y,1,color,circle_ratio) ;
}
/*-----*/
void update()
/* This is a routine to update relative positions*/
/* after the position of the primary aircraft */
/* (myplane) changes */
{
  register int  i, j, new;
  float        direction;

  mylat = latitude[0][data_pointer[0]];
  mylon = longitude[0][data_pointer[0]];
  myalt = altitude[0][data_pointer[0]];
  myheading = heading[0][data_pointer[0]];
  myspeed = speed[0][data_pointer[0]];
  for (i=1;i<NACFT;i++)
  {
    j = data_pointer[i];
    if (j != -1)
    {
      new = check(j+1);
      data_pointer[i] = new;
      latitude[i][new] = latitude[i][j];
      longitude[i][new] = longitude[i][j];
      altitude[i][new] = altitude[i][j];
      sample_time[i][new] = sample_time[i][j];
      heading[i][new] = heading[i][j];
      speed[i][new] = speed[i][j];
      v_speed[i][new] = v_speed[i][j];
      plotted[i][new] = false;

      /*----- calculate new distance ,rel_bearing, rel_heading -----*/
      distance[i][new] = dist(mylat,mylon,latitude[i][new],
        longitude[i][new],&direction);
      rel_bearing[i][new] = relative(myheading,direction);
      rel_heading[i][new] = relative(myheading,heading[i][new]);
    }
  }
}
/*-----*/

int check(register int i)
/* This routine checks the value of the data */
/* pointer, and sets it between 0 and NDATA -1 */
{
  if (i > NDATA-1)
    i = i - NDATA;
  if (i<0)
    i = i + NDATA;
  return i;
}
/*-----*/

```

```

void check_keyboard()
{
    char key;
    if (kbhit())
    {
        key = getch();
        if (key == 0)                /* if value = 0, its a function key */
            key = getch();          /* get the key value */
        if (key == F1)
            vector_flag = toggle(vector_flag);
        if (key == F2)
            altitude_flag = toggle(altitude_flag);
        if (key == F3)
            nnumber_flag = toggle(nnumber_flag);
        if (key == F4)
            zoom(1);                /* zoom in */
        if (key == F5)
            zoom(0);                /* zoom out */
        if (key == F6)
            data_save_flag = toggle(data_save_flag); /* save input data */
        if (key == F7)
            print_string_flag = toggle(print_string_flag);

        if (key == F11)
            initialize();
#ifdef COMP == 1
        if (key == F10)
        {
            data_in();
            process_data();
        }
        if (key == SF10)
            autoread_flag = toggle(autoread_flag);
#endif
        key = tolower(key);
        if (key == 'q')
            terminate();
    }
#ifdef COMP == 1
    if (autoread_flag)
    {
        data_in();
        process_data();
    }
#endif
}
/*-----*/

int toggle( int variable)
{
    if (variable > 0)
        variable = false;
    else
        variable = true;
    return variable;
}
/*-----*/

void zoom(int i)
{
    switch (i)
    {
        case 0:
        {
            if (scale > .015625)
                scale = scale / 2.0;
            break;
        }
        case 1:
        {
            if (scale < 4)
                scale = scale * 2.0;
        }
    }
}

```

```

        break;
    }
}
cls(BLACK);
distance_circles();
update();
hold_rewrite = true;
}
/*-----*/

int validate(buffer)
char    buffer[81];
{
    register int    err;
    int            cksum, value;
    char          ascii_cksum[10], temp[10], *stop;

    if (buffer[0] != '[')
        return 1;                    /* 1 = no left bracket */
    checksum(buffer, &cksum, ascii_cksum, 32);
    memmove(temp, &buffer[32], 2);
    temp[2] = '\0';
    value = (int) strtoul(temp, &stop, 16);
/*    printf("%i;%i\n", value, cksum); */
    if (cksum != value)
        return 2;                    /* 2 = bad checksum */
    else
        return 0;                    /* 0 = no error */
}
/*-----*/

void checksum(buffer, cksum, ascii_cksum, nchars)
char    buffer[81], ascii_cksum[10];
int     nchars, *cksum;
{
    register int    i;

    *cksum = 0;
    for (i=0; i<nchars; i++)
    {
/*        printf("%c,%d\n", buffer[i], buffer[i]); */
        *cksum = *cksum + buffer[i];
    }
    *cksum = *cksum & 0xFF;
    itoa(*cksum, ascii_cksum, 16);
/*    printf("checksum = %d; %s; # digits = %d\n", *cksum, ascii_cksum, nchars); */
}
/*-----*/

void align(buffer)
char    buffer[81];
{
    register int    i, j, k;
    char          temp[80];

    i = strchr(&buffer[1], "[");    /* search for "[" in string */
    i++;
    if (i < 35)                    /* found one */
        memmove(buffer, &buffer[i], 35);    /* shift to left justify */
    else
    {
        do
        {
            read_buffer(temp, 1);        /* go find a "[" */
        } while (!memcmp(temp, "[", 1));
    }
    for (j=0; j<i;)                /* fill the rest of the buffer */
    {
        read_buffer(temp, 1);
    }
}

```

```

    if (!memcmp(temp, "", 1))
    {
        buffer[0] = '[';
        read_buffer(temp, 34);
        memmove(&buffer[1], temp, 34);
        buffer[35] = '\0';
        return;
    }
    k = 35 - i + j;
    memmove(&buffer[k], &temp[0], 1);
    j++;
}
buffer[35] = '\0';
}
/*-----*/

void open_port()
{
    if (xc_entr(2) == 0)
    {
        if (PORT == COM3)
            setport(0x2E8, 1, 4, PORT, 0);
        xc_link(PORT, 0);
        xc_init(PORT, RATE, NOPAR, DATA8, STOP1);
    }
    else
    {
        printf("Unable to open PORT");
        exit(0);
    }
}
/*-----*/

int read_buffer(buffer, count)

int count;
char buffer[81];

{
    char key;

    if (xc_test(PORT) >= count)
    {
        xc_get(PORT, buffer, &count);
        buffer[35] = '\0';
        return 1;
    }
    /* if (kbhit())
    {
        key = getch();
        key = tolower(key);
        if (key == 'q')
            terminate();
    }*/
    return 0;
}
/*-----*/

void terminate()
{
    #if COMP==2
        xc_unlk(PORT);
        xc_exit();
    #endif
    setMode(save_mode);
    cls(0);
    exit(0);
}
/*-----*/
#if COMP==1
void open_file()

```

```

                                /* Routine to open a data file          */
{
    if ((fp = fopen("data","r")) == NULL)
    {
        printf("Cannot open data file\n")      ;
        terminate()                            ;
    }
}
/*-----*/

void read_rec()
                                /* read a file record (terminated with \n) */
{
    int          i
    char         ch
                                ;
                                ;
    i=0
                                ;
    ch = getc(fp)
                                ;
    while (ch!='\n' && ch!=EOF)
    {
        read_string[i] = ch
                                ;
        i++
                                ;
        ch = getc(fp)
                                ;
    }
    read_string[i] = '\0'
                                ;
    /* printf ("%s\n",read_string)
                                ; */
}
#endif
/*-----*/

void plot_read_string()

{
char out_string[40];
static char old_string[40] = "
                                ";

    OPERATOR = REPLACE;
    sprintf(out_string,"%s\n",read_string);
    out_string[35] = '\0';
    plot_horz_str(-300,-210,old_string,BLACK);
    plot_horz_str(-300,-210,out_string,LIGHTRED);
    memmove(old_string,out_string,36);
}
/*-----*/

void open_save_file()
                                /* Routine to open file to save data          */
{
    if ((save_file = fopen("datasave","a")) == NULL)
    {
        printf("Cannot open save file\n")      ;
        terminate()                            ;
    }
}
/*-----*/
                                END OF PROGRAM
/*-----*/

```


What is claimed and desired to be secured by United States Letters Patent is:

1. A method for announcing the position of a first aircraft at a first position, the first position being within airspace containing a plurality of other aircraft, to a receiver located at a second position not farther than a predetermined range from the first position, the method comprising the steps of:
 - (a) determining on board the first aircraft the position of the first aircraft relative to a fixed reference;
 - (b) encoding an identification code, the presence or absence of any current operating communication radio frequency and the position of the first aircraft so that the identification code, the presence or absence of current operating communication radio frequency and the position of the first aircraft can be carried by a first radio frequency signal;
 - (c) avoiding conflicts with any radio frequency signals present in the airspace which conflict with the first radio frequency signal;
 - (d) transmitting from the first aircraft the first radio frequency signal carrying the identification code, the presence or absence of any current operating communication radio frequency and the position of the first aircraft;
 - (e) receiving the first radio frequency signal transmitted from the first aircraft at the second position;
 - (f) decoding the identification code, the current operating communication radio frequency if any, and the position of the first aircraft from the first radio frequency signal; and
 - (g) monitoring on a visual display the identification code, the current operating communication radio frequency if any, and the position of the first aircraft within the airspace at the second position so that the position of the first aircraft within the airspace is known relative to the fixed reference, and so that communication can potentially be undertaken with the specifically identified aircraft.
2. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 1 wherein the step of determining the position of the first craft comprises the step of receiving a LORAN signal.
3. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 1 wherein the step of determining the position of the first craft comprises the step of receiving a GPS signal.
4. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 1 wherein the fixed reference exists both within the airspace and outside the airspace.
5. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 4 wherein the step of determining the position of the first aircraft comprises the step of determining the position of the first aircraft in reference to the fixed reference.
6. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 5 wherein the step of determining the position of the first aircraft further comprises the step of determining the altitude of the first aircraft.
7. A method for announcing the position of a first aircraft at a first position to a receiver located at a sec-

ond position as defined in claim 1 or 5 wherein the step of encoding the position of the first aircraft comprises the step of encoding the position of the aircraft into a digital pulse train.

8. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 7 wherein the step of encoding the position of the first aircraft further comprises the step of translating the position of the first aircraft into an audio signal.
9. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 1 wherein the step of avoiding conflicts with any conflicting radio frequency signals in the airspace comprises the step of listening for any conflicting radio frequency signals.
10. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 9 wherein the step of avoiding conflicts with any conflicting radio frequency signals in the airspace comprises the step of waiting a period of time prior to transmission.
11. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 10 wherein the step of avoiding conflicts with any conflicting radio frequency signals in the airspace comprises the step of waiting a predetermined period of time prior to transmission.
12. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 10 wherein the step of avoiding conflicts with any conflicting radio frequency signals in the airspace comprises the step of waiting a random period of time prior to transmission.
13. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 1 or 10 wherein the step of avoiding conflicts with any conflicting radio frequency signals in the airspace comprises the step of transmitting a radio frequency signal in a non-interrogation manner.
14. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 1 wherein the step of transmitting the first radio frequency signal comprises the step of transmitting the first radio frequency signal in the frequency range from about and including the VHF band to about and including the SHF band.
15. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 1 wherein the step of transmitting the first radio frequency signal comprises the step of transmitting the first radio frequency signal in a substantially omnidirectional pattern.
16. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 1 further comprising the step of repeating steps (a) through (g).
17. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 16 wherein the step of monitoring the position of the first aircraft comprises the step of repeatedly displaying the position of the first aircraft on a display as the aircraft changes position.
18. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 17 wherein the step of

monitoring the position of the first aircraft further comprises the step of displaying the heading and the speed of the first aircraft on a display.

19. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 1 wherein the step of monitoring the position of the first aircraft comprises the step of displaying the relative position of the first aircraft on a display over a selected period of time.

20. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 1 further comprising the step of determining the location of the second position relative to the fixed reference and wherein the step of monitoring the position of the first aircraft comprises the step of displaying the position of the first aircraft and the location of the second position.

21. A method of tracking the position of a first craft by monitoring receiver at a second craft, the first craft and the monitoring receiver being within a predetermined range of each other, the method comprising the steps of:

- (a) determining, internally to the first craft using an external automated navigational system, the first craft's position in at least one plane of space;
- (b) encoding for the first craft an identification code, the presence or absence of any current communication radio frequency and the first craft's position, all in a form which can be carried by a first radio frequency signal;
- (c) determining when the first radio frequency signal should be transmitted;
- (d) transmitting the first radio frequency signal;
- (e) receiving the first radio frequency signal at the monitoring receiver of the second craft;
- (f) decoding at the monitoring receiver of the second craft the first craft's identification code, any current communication radio frequency and the position from the first radio frequency signal;
- (g) displaying the identification code, any current communication radio frequency and the position of the first craft at the monitoring receiver of the second craft; and
- (h) repeating steps (a) through (g).

22. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitoring receiver being within a predetermined range of each other, as defined in claim 21 wherein the step of determining the first craft's position comprises the step of determining the first craft's position using a LORAN receiver.

23. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitoring receiver being within a predetermined range of each other, as defined in claim 21 wherein the step of determining the first craft's position comprises the step of determining the first craft's position using a GPS receiver.

24. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitoring receiver being within a predetermined range of each other, as defined in claim 21 wherein the first craft's position is defined in reference to the latitude and longitude of the first craft.

25. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitoring receiver being within a predetermined range of each other, as defined in claim 24 wherein the step of deter-

mining the first craft's position further comprises the step of determining the altitude of the first craft.

26. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitoring receiver being within a predetermined range of each other, as defined in claim 24 or 25 wherein the step of encoding the first craft's position comprises the step of encoding the position of the first craft into a digital pulse train.

27. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitoring receiver being within a predetermined range of each other, as defined in claim 26 wherein the step of encoding the position of the first craft's position further comprises the step of translating the position of the first craft into an audio signal.

28. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitoring receiver being within a predetermined range of each other, as defined in claim 21 further comprising the step of avoiding conflicts with any conflicting radio frequency signals in the airspace.

29. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitoring receiver being within a predetermined range of each other, as defined in claim 28 wherein the step of avoiding conflicts with any conflicting radio frequency signals in the airspace comprises the step of listening for any conflicting radio frequency signals.

30. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitoring receiver being within a predetermined range of each other, as defined in claim 28 wherein the step of avoiding conflicts with any conflicting radio frequency signals in the airspace comprises the step of waiting a period of time prior to transmission.

31. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitoring receiver being within a predetermined range of each other, as defined in claim 28 wherein the step of avoiding conflicts with any conflicting radio frequency signals in the airspace comprises the step of waiting a predetermined period of time prior to transmission.

32. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitoring receiver being within a predetermined range of each other, as defined in claim 21 or 28 wherein the step of avoiding conflicts with any conflicting radio frequency signals in the airspace comprises the step of transmitting a radio frequency signal in a non-interrogation manner.

33. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitoring receiver being within a predetermined range of each other, as defined in claim 21 wherein the step of transmitting the first radio frequency signal comprises the step of transmitting the first radio frequency signal in the frequency range from about and including the VHF band to about and including the SHF band.

34. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitoring receiver being within a predetermined range of each other, as defined in claim 21 wherein the step of transmitting the first radio frequency signal comprises the step of transmitting the first radio frequency signal in a substantially omnidirectional pattern.

35. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitor-

ing receiver being within a predetermined range of each other, as defined in claim 21 wherein the step of displaying the position of the first craft at the monitoring receiver comprises the step of repeatedly displaying the position of the first craft on a display of the second craft as the first craft changes position.

36. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitoring receiver being within a predetermined range of each other, as defined in claim 21 wherein the step of displaying the position of the first craft at the monitoring receiver further comprises the step of displaying the heading and the speed of the first craft on a display at the second craft.

37. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitoring receiver being within a predetermined range of each other, as defined in claim 35 comprises the step of displaying the position of the second craft on a display, the second craft having the monitoring receiver on board.

38. A method of tracking the position of a first craft by a monitoring receiver, the first craft and the monitoring receiver being within a predetermined range of each other, as defined in claim 37 further comprising the step of determining the location of the second position relative to the fixed reference and wherein the step of displaying the position of the first craft comprises the step of displaying the position of the first craft and the location of the monitoring receiver on the display.

39. A method of tracking the position of a first craft from on board a second craft, the first and second craft being within a predetermined range of each other, the method comprising the steps of:

- (a) determining on board the first craft using an existing automated navigational system the first craft's position in at least two perpendicular planes of space which are of interest to the second craft;
- (b) encoding for the first craft its identification code and the presence or absence of a current operating communication radio frequency on which voice contact can be initiated to the first craft, and encoding the first craft's position, all in a form which can be carried by a first radio frequency signal;
- (c) determining whether the first radio frequency signal will conflict with any other radio frequency signals;
- (d) transmitting the first radio frequency signal;
- (e) receiving the first radio frequency signal at the second craft;
- (f) decoding and displaying the first craft's identification code, current operating communication radio frequency if any and the first craft's position, all from the first radio frequency signal, such that the first craft can potentially be contacted by voice communication;
- (g) apprising the operator of the second craft when the position of the first craft is within an advisory range; and
- (h) repeating steps (a) through (g).

40. A method of tracking the position of a first craft from on board a second craft, the first and second craft being within a predetermined range of each other, as defined in claim 39 wherein the step (a) of determining the first craft's position comprises the step of receiving a LORAN signal and determining the first craft's position in latitude and longitude.

41. A method of tracking the position of a first craft from on board a second craft, the first and second craft

being within a predetermined range of each other, as defined in claim 39 wherein the step (a) of determining the first craft's position comprises the step of receiving a GPS signal.

42. A method of tracking the position of a first craft from on board a second craft, the first and second craft being within a predetermined range of each other, as defined in claim 39 wherein said existing automated navigational system is available both within and outside the predetermined range.

43. A method of tracking the position of a first craft from on board a second craft, the first and second craft being within a predetermined range of each other, as defined in claim 42 wherein the step (a) of determining the first craft's position comprises the step of determining the first craft's position in latitude and longitude.

44. A method of tracking the position of a first craft from on board a second craft, the first and second craft being within a predetermined range of each other, as defined in claim 42 wherein the step (a) of determining the first craft's position further comprises the step of determining the altitude of the first craft.

45. A method for announcing the position of a first aircraft at a first position to a receiver located at a second position as defined in claim 43 or 44 wherein the step of avoiding conflicts with any conflicting radio frequency signals in the airspace comprises the step of transmitting a radio frequency signal in a non-interrogation manner.

46. A method of tracking the position of a first craft from on board a second craft, the first and second craft being within a predetermined range of each other, as defined in claim 43 or 44 wherein the step of encoding the first craft's position comprises the step of encoding the position of the first craft into a digital pulse train.

47. A method of tracking the position of a first craft from on board a second craft, the first and second craft being within a predetermined range of each other, as defined in claim 46 wherein the step of encoding the position of the first aircraft further comprises the step of translating the first craft's position into an audio signal.

48. A method of tracking the position of a first craft from on board a second craft, the first and second craft being within a predetermined range of each other, as defined in claim 46 wherein the step of determining whether the first radio frequency signal will conflict with any other radio frequency signals comprises the steps of:

- listening for any conflicting radio frequency signals; and
- waiting a period of time prior to transmission.

49. A method of tracking the position of a first craft from on board a second craft, the first and second craft being within a predetermined range of each other, as defined in claim 48 wherein the step of waiting a period of time comprises the step of waiting a randomly assigned period of time.

50. A method of tracking the position of a first craft from on board a second craft, the first and second craft being within a predetermined range of each other, as defined in claim 39 wherein the step of encoding the first craft's position comprises the step of encoding an identification code for the first craft so that the identification code can be carried by the first radio frequency signal and wherein the step of transmitting the first radio frequency signal comprises the steps of:

- transmitting the first radio frequency signal in the frequency range from about and including the

VHF band to about and including the SHF band; transmitting the first radio frequency signal in a substantially omnidirectional pattern; and transmitting the identification code for the first craft.

51. A method of tracking the position of a first craft from on board a second craft, the first and second craft being within a predetermined range of each other, as defined in claim 39 wherein the step of apprising the operator comprises the step of repeatedly displaying the position of the first craft on a display as the first craft changes position.

52. A method of tracking the position of a first craft from on board a second craft, the first and second craft being within a predetermined range of each other, as defined in claim 51 wherein the step of apprising the operator of the second craft comprises the steps of:

displaying the relative position and direction of travel of the first craft on a display; and displaying the position of the second craft on a display.

53. A method of avoiding collisions between a first craft having a first identification code and at least a second craft having a second identification code, the first and second crafts coming within a predetermined range of each other, the method comprising the steps of:

- (a) determining, on board to the first craft and using an existing automated navigational system, the first craft's position in at least one plane in space which is of interest to the second craft to avoid a collision with the first craft;
- (b) encoding the first craft's position, identification code and the presence or absence of any current operating communication radio frequency on which voice contact with the first craft can be established, all to be carried by a first radio frequency signal;
- (c) listening for any radio frequency signals present on the same frequency allocation as the first radio frequency signal;
- (d) transmitting the first radio frequency signal carrying the first craft's position, identification code and the presence or absence of any current communication radio frequency on which voice contact with the first craft can be established;
- (e) receiving the first radio frequency signal at the second craft;
- (f) decoding from the first radio frequency the first craft's position, identification code and current communication radio frequency, if any, on which voice contact with the first craft can be established;
- (g) determining the second craft's position, on board the second craft by using said existing automated navigational system, and in at least one dimension of space, which is of interest to the operator of the second craft, to avoid a collision with the first craft;
- (h) displaying, on board the second craft, the position, identification code and current communication radio frequency, if any, on which voice contact to the first craft can be established, and displaying the location of the second craft so that the operator of the second craft is apprised of the location of the both the first craft and the second craft so as to avoid a collision therebetween; and
- (i) repeating steps (a) through (h).

54. A method of avoiding collisions between a first craft having a first identification code and at least a second craft having a second identification code, the

first and second craft coming within a predetermined range of each other, as defined in claim 53 wherein the step of determining the first craft's position comprises the step of receiving a LORAN signal or a GPS signal.

55. A method of avoiding collisions between a first craft having a first identification code and at least a second craft having a second identification code, the first and second craft coming within a predetermined range of each other, as defined in claim 54 wherein the first craft comprises an aircraft and wherein the step of determining first craft's position comprises the step of determining the latitude, longitude, and altitude of the first aircraft.

56. A method of avoiding collisions between a first craft having a first identification code and at least a second craft having a second identification code, the first and second craft coming within a predetermined range of each other, as defined in claim 53 wherein the step of encoding the first craft's position comprises the step of encoding the latitude and longitude of the first craft into a digital pulse train which can be carried by a frequency modulated first radio frequency signal.

57. A method of avoiding collisions between a first craft having a first identification code and at least a second craft having a second identification code, the first and second craft coming within a predetermined range of each other, as defined in claim 53 further comprises the step of waiting a period of time after a conflicting radio frequency signal is heard prior to transmission of the first radio frequency signal, the period of time being different for the first craft and the second craft, the transmission of the first radio frequency signal being initiated without interrogation from another radio frequency signal.

58. A method of avoiding collisions between a first craft having a first identification code and at least a second craft having a second identification code, the first and second craft coming within a predetermined range of each other, as defined in claim 53 wherein the step of transmitting the first radio frequency signal comprises the step of transmitting the first radio frequency signal in the frequency range from about and including the VHF band to about and including the SHF band in a substantially omnidirectional pattern.

59. A method of avoiding collisions between a first craft having a first identification code and at least a second craft having a second identification code, the first and second craft coming within a predetermined range of each other, as defined in claim 53 wherein the step of displaying the position of the first craft comprises the step of displaying the relative direction of travel of the first craft on a display.

60. An apparatus for announcing the position of a first aircraft at a first position within airspace containing a plurality of other aircraft, to a receiver located at a second position within a predetermined range of the first aircraft, the first aircraft being within operational range of an existing navigational system which can provide position information in at least two planes of interest, the apparatus comprising:

means for making an on board determination of the position of the first aircraft using the existing navigational system;

means for encoding the position of the first aircraft, and any identification code and the presence or absence of any current communication radio frequency on which voice contact to the first aircraft can be made, into information which can be carried

by a first radio frequency signal;
means for detecting the presence of any conflicting radio frequency signal which would interfere with the first radio frequency signal and determining whether transmission of the first radio frequency should proceed;

means for transmitting the first radio frequency signal from the first aircraft such that the radio frequency signal can be received at the receiver, decoded, and such that the position, identification code and current communication radio frequency, if any, on which voice contact can be made to the first aircraft is displayed.

61. An apparatus for announcing the position of a first aircraft at a first position to a receiver located at a second position within a predetermined range of the first aircraft, the first aircraft being within operational range of an existing navigational system which can provide position information in at least two planes of interest, as defined in claim 60 wherein the means for making an on board determination of the position of the first aircraft comprises a LORAN receiver.

62. An apparatus for announcing the position of a first aircraft at a first position to a receiver located at a second position within a predetermined range of the first aircraft, the first aircraft being within operational range of an existing navigational system which can provide position information in at least two planes of interest, as defined in claim 60 wherein the means for making an on board determination of the position of the first aircraft comprises a GPS receiver.

63. An apparatus for announcing the position of a first aircraft at a first position to a receiver located at a second position within a predetermined range of the first aircraft, the first aircraft being within operational range of an existing navigational system which can provide position information in at least two planes of interest, as defined in claim 60 wherein the means for making an on board determination of the position of the first aircraft comprises means for determining the latitude and longitude of the first aircraft.

64. An apparatus for announcing the position of a first aircraft at a first position to a receiver located at a second position within a predetermined range of the first aircraft, the first aircraft being within operational range of an existing navigational system which can provide position information in at least two planes of interest, as defined in claim 63 wherein the means for making an on board determination of the position of the first aircraft further comprises means for determining the altitude of the aircraft.

65. An apparatus for announcing the position of a first aircraft at a first position to a receiver located at a second position within a predetermined range of the first aircraft, the first aircraft being within operational range of an existing navigational system which can provide position information in at least two planes of interest, as defined in claim 60 wherein the means for transmitting the first radio frequency signal comprises a radio frequency transmitter operating in a frequency range from about and including the VHF band to about and including the SHF band.

66. An apparatus for announcing the position of a first aircraft at a first position to a receiver located at a second position within a predetermined range of the first aircraft, the first aircraft being within operational range of an existing navigational system which can provide position information in at least two planes of interest, as

defined in claim 65 wherein the means for transmitting the first radio frequency signal further comprises an omnidirectional antenna.

67. An apparatus for announcing the position of a first aircraft at a first position to a receiver located at a second position within a predetermined range of the first aircraft, the first aircraft being within operational range of an existing navigational system which can provide position information in at least two planes of interest, as defined in claim 60 wherein the means for detecting the presence of any conflicting radio frequency signal comprises a radio frequency receiver.

68. An apparatus for announcing the position of a first aircraft at a first position to a receiver located at a second position within a predetermined range of the first aircraft, the first aircraft being within operational range of an existing navigational system which can provide position information in at least two planes of interest, as defined in claim 60 wherein the means for encoding the position of the first aircraft is further for encoding an identification code into information which can be carried by the first radio frequency signal.

69. An apparatus for announcing the position of a first aircraft at a first position to a receiver located at a second position within a predetermined range of the first aircraft, the first aircraft being within operational range of an existing navigational system which can provide position information in at least two planes of interest, as defined in claim 60 further comprising:

means for receiving a second radio frequency signal carrying the position and identification code for a second aircraft, and the presence or absence of any current communication radio frequency on which voice contact to the second aircraft can be established;

means for decoding from the second radio frequency signal the second aircraft's position, identification code, and the presence or absence of any current communication radio frequency on which voice contact to the second aircraft can be established; and

means for displaying, on board the first aircraft, the second aircraft's position, identification code, and the presence or absence of any current communication radio frequency on which voice contact to the second aircraft can be established so that the operator of the first craft is apprised of the position of the second aircraft and is potentially able to make voice contact with the second aircraft.

70. An apparatus for tracking the position of a second craft from on board a first craft, the first and the second craft being within range of an existing navigational system and within a predetermined range of each other, the apparatus comprising:

means, on board the first craft, for receiving a determination of the position of the first craft using the existing navigational system, the position of the first craft being determined in at least two planes of interest to the operator of the second craft;

means for encoding the first craft's position, and any identification code and the presence or absence of any current communication radio frequency on which voice contact to the first craft can be made, in a form which can be carried by a first radio frequency signal;

means for transmitting the first radio frequency signal;

means for receiving a second radio frequency signal carrying the second craft's position an identification code for the second craft and the presence or absence of any current communication radio frequency on which the second craft can be contacted by voice; 5

means for decoding the second craft's position, identification code and current communication radio frequency, if any, from the second radio frequency signal; 10

means for displaying, on board the first craft, the second craft's position, identification code and current communication radio frequency, if any, decoded from the second radio frequency signal so that the operator of the first craft is apprised of the position, identification code and current communication radio frequency, if any, of the second craft. 15

71. An apparatus for tracking the position of a second craft from on board a first craft as defined in claim 70 wherein the means for receiving an on board determination of the position of the first craft comprises a data port adapted to receive data from a LORAN receiver. 20

72. An apparatus for tracking the position of a second craft from on board a first craft as defined in claim 70 wherein the means for receiving an on board determination of the position of the first craft comprises a data port adapted to receive data from a GPS receiver. 25

73. An apparatus for tracking the position of a second craft from on board a first craft as defined in claim 70 wherein the means for transmitting the first radio frequency signal comprises: 30

a radio frequency transmitter operating in a frequency range from about and including the VHF band to about and including the SHF band; and an omnidirectional antenna system. 35

74. An apparatus for tracking the position of a second craft from on board a first craft as defined in claim 73 further comprising means for detecting the presence of any conflicting radio frequency signal. 40

75. An apparatus for avoiding collisions between a first craft having a first identification code and at least a second craft having a second identification code, the first craft being within range of an existing navigational system and the first and the second craft being within a predetermined range of each other, the apparatus comprising: 45

means for determining, on board the first craft and using the existing navigational system, the first craft's position in at least one plane in space which is of interest to the second craft to avoid a collision with the first craft; 50

means for encoding the first craft's position, the first identification code and the presence or absence of any current communication radio frequency on 55

which the first craft can be contacted by voice, such that the encoded information is carried by a first radio frequency signal;

means for detecting the presence of any conflicting radio frequency signal and determining whether transmission of the first radio frequency should proceed;

means for transmitting the first radio signal;

means for receiving, on board the first craft, a second radio frequency signal transmitted from the second craft, the second radio frequency signal carrying the position of the second craft, the second identification code and the presence or absence of any current communication radio frequency on which the second craft can be contacted by voice; 10

means for decoding, on board the first craft, the second craft's position, the second identification code and the presence or absence of any current communication radio frequency on which the second craft can be contacted by voice; 15

means for simultaneously displaying the first craft's position and the second craft's position on board the first craft so that the operator of the first craft is apprised of the position of the second craft, the second identification code and the presence or absence of any current communication radio frequency on which the second craft can be contacted by voice so as to avoid a collision therebetween. 20

76. An apparatus for avoiding collisions between a first craft having a first identification code and at least a second craft having a second identification code as defined in claim 75 wherein the means for determining the first craft's position comprises a LORAN receiver. 25

77. An apparatus for avoiding collisions between a first craft having a first identification code and at least a second craft having a second identification code as defined in claim 75 wherein the means for determining the first craft's position comprises a GPS receiver. 30

78. An apparatus for avoiding collisions between a first craft having a first identification code and at least a second craft having a second identification code as defined in claim 76 or 77 wherein the means for transmitting the first radio frequency signal comprises: 35

a radio frequency transmitter operating in a frequency range from about and including the VHF band to about and including the SHF band; and an omnidirectional antenna system. 40

79. An apparatus for avoiding collisions between a first craft having a first identification code and at least a second craft having a second identification code as defined in claim 78 wherein the means for detecting the presence of any conflicting radio frequency signal comprises a radio frequency receiver. 45

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UNITED STATES PATENT AND TRADEMARK OFFICE
CERTIFICATE OF CORRECTION

PATENT NO. : 5,153,836
DATED : October 6, 1992
INVENTOR(S) : EDWARD J. FRAUGHTON et al.

It is certified that error appears in the above-identified patent and that said Letters Patent is hereby corrected as shown below:

Title page, column 1, lines 9-10, "Salt Lake City, Utah" should be
--South Jordan, Utah--
Column 1, line 10, "any one" should be --anyone--
Column 2, line 13, "dependant" should be --dependent--
Column 2, line 16, "dependant" should be --dependent--
Column 4, line 27, "interfer" should be --interfere--
Column 9, line 48, delete "a"
Column 10, lines 38-39, "speed spectrum" should be --spread
spectrum--
Column 10, line 40, before "non-interrogation" insert
--interrogation or--
Column 232, line 24, "aircraft" should be --craft--
Column 233, line 26, eliminate "to"
Column 229, line 19, after "by" insert --a--

Signed and Sealed this

Twenty-third Day of November, 1993

Attest:



BRUCE LEHMAN

Attesting Officer

Commissioner of Patents and Trademarks